

April 21, 1958

Aviation Week

Including Space Technology

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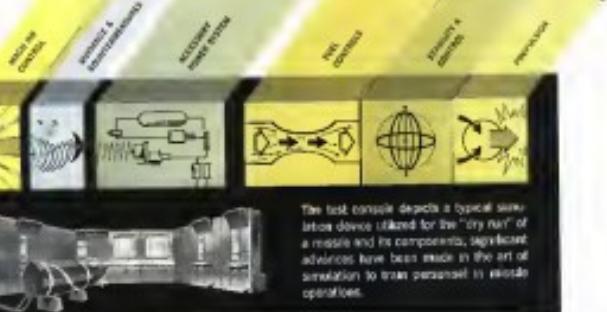
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AVIATION CALENDAR

(Continued from page 5)

program, correlated program in the Society of Aircraft Materials and Process Engi
neers and the Magazine Test Institute of Aerostatic Sciences (M.A.T.S.) in A
rlington, Va.

June 9-12—Six Annual Meeting, American
Radar Society, Inc., Ward Inn, Los
Angeles, Calif.

June 9-13—Fourth International Astronautical
Exposition and Congress, Columbus,
Ohio, U.S.A.

June 10-12—National Convention
on Metal Fatigue, Statler Park
Hotel, Washington, D.C.

June 25-July 1—Special Session Program on
Rotary Aviation, as introduced to the
aviation people, at the 1950 meeting of the
American Mechanical Institute of Tech
nology, Cambridge, Mass.

July 1-4—10th Meeting, Aviation De
partment and Manufacturing Arms, Wash
ington Hilton, Section Woods, N.H.

July 21-27—Av. Transportation Conference
International, Philadelphia, Pa., Hotel
Pennsylvani, Philadelphia, Pa.

July 26-28—American Aviation Exposition,
Columbus, Ohio, Pittsburgh, Pa., 100 dis
tinct units. An American Aviation Exposi
tion, Inc., Columbus, 2100 Superior
Ave., Cleveland, Ohio.

July 30-August 1—Motor Sport Show, Chas
ton Hotel, Grand Central, New York.

July 31-August 1—The Institute of the Aerospace
Sciences, National Seminar Meeting, Air
Institute Hotel, Los Angeles, Calif.

July 31-August 1—The International
Aerospace Congress, an Aerospace
Area, Aerostatic Laboratory, Moffet
Field, Calif.

July 24-25—10th Annual Symposium on
Corporation and Data Processing, Ahren
Hotels, Denver, Colo.

Aug. 1-2—Aeroplane Technical Conference on
New Large Magneto and Magnets. An
exhibition sponsored by the American In
stitute of Electrical Engineers, Hotel
InterContinental, New York.

Aug. 19-23—14th Annual Worldwide Show &
Conference Institute of Radio Engineers
Washington Hotel, Los Angeles, Calif.

Sept. 17-19—1950 Broadcast Flying Drills and
Exhibitions, Service of Radio and
Television, Pease Field, Bedford, Eng
land.

Sept. 15-19—1950 Chicago Engineering Con
ference, McCormick Institute of Tech
nology, Chicago, Ill.

Sept. 6-18—International Aviation Show
Columbus, New York, N.Y.

Sept. 8-13—International Congress of
the International Society, Palace Hotel
Vienna, Austria.

Sept. 8-14—Diseases of Flight. University
of Michigan receives grants for investi
gating diseases. The doctor R. M. Hilt
of Aerostatic Foundation 1413
East Farnsworth, 8000 University of
Michigan, Ann Arbor, Mich.

Sept. 13-16—1950 Annual General Meeting of
the International Air Transport Assn., New
Delhi, India.

Royal Swedish Navy Orders Vertol 44's

A special committee of the Swedish Navy made a year-long study of helicopters suitable for anti-submarine warfare as well as utility transportation. In the United States they consulted with headquarters personnel of the three armed services, visited helicopter manufacturers and toured military operating units.

At the conclusion of this intensive study, the Swedish Navy placed an initial order for four of the 22-place Vertol 44's, to be aug
mented later by additional purchases.



Official Navy announcement of the purchase said the choice was mainly based upon the Vertol 44's "good stability in hovering and landing maneuvers, good instrument flight capabilities and mission endurance... particularly unlimited maneuverability for the contemplated missions."

Sweden thus becomes the fifth western nation to purchase the sturdy Work Horse helicopter, fifth country of the empire for the most difficult assignments.

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After studying the problem, Timken Company metal stamping recommended dies made from Graph-Mo[®]—a special tool steel developed by the Timken Company. Results were outstanding. The new Graph-Mo dies cut downtime 50%. The combination of long-graphite particles

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Graph-Mo reduces 50% smaller than conventional tool steels. And its uniform response to heat treatment eliminates distortion—saves time and money in lots of rough jobs.

Graph-Mo is one of four graphites used in steels developed by the Timken Company. If you would like more information, write us for our new Timken Graphite Steel Book. The Timken Roller Bearing Company, Steel and Tube Division, Canton 6, Ohio. Cable address, "TIMKON".

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INFRARED EQUIPMENT Bulova's advanced IR products include missile cells that will automatically fire out thermal sensor targets and piloted targets—safely against any background. Bulova also makes laser and lead stability units; acoustic mirror cells; vehicles filters; and thermal or bolometers... for defense and industry.



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Fast Flights—Electra/Flight gives your passengers quick and easy take-off, fast climb...sooner in the air—saves time on the ground, too.

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3 Want delays ...in starting off ...in rescheduling or stops on the way ...in getting their luggage

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First business plane flies polar route with B. F. Goodrich De-Icers

A production model Lockheed L-1049 recently became the first two-passenger business plane to make a non-polar loop, flying through one of the world's ice belts in the world, the Lockheed made only one stop for refueling between Los Angeles and Dusseldorf, Germany. Total elapsed time—72 hours, 18 minutes.

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EDITORIAL

Financial Storm Brewing

Last summer and fall the aviation industry and its related technologies were put through a financial wringer by government fiscal policies that are still being felt in virtually every phase of the country's economy. It does not take a doctorate in economics to trace the beginning of the present economic slowdown to the traumatic shock induced in the aviation budget manufacturing industry by the government's final bungling of last summer's *Joint Budget*. Pentagon officials blithely maintained they would not pay any bills then legitimately due for at least another six months.

Basic Problem

In addition to the aviation economic damage inflicted on the aviation industry, these fiscal policies threw a cold shock of fear down through all of the economic channels of manufacturing and supplies that carry the defense dollars down to the grass roots of the American economy.

Basically, the defense fiscal problem has been the result of a commitment failure of the executive branch to face up to the full cost of the defense program they have repeatedly promised to provide for the American people. After the end of the Korean war in 1953 there were huge sums of appropriated but unexpended funds available to all three services. For several years the services were able to finance their programs adequately out of the Korean surplus despite radically reduced new appropriations. When the Korean surplus was exhausted, the civilian leaders of the Defense Department refused to face the hard financial facts of their problems. They continued the below-the-budget requests while continuing to trumpet that this would provide an adequate defense program. Each year since fiscal 1955, the gap between the programs promised and the money budgeted to finance it has widened.

Industry Forced to Borrow

This was first played over in October as defense strength labeled as "efficient economy." Last year the critics became so septic fast in the Air Force alone, according to a memo from Secretary James Douglas to the then Defense Secretary Charles E. Wilson, a \$4 billion gap existed between authorized hardware procurement programs and the funds available to pay for delivery of this hardware. To solve this problem, the Pentagon civilian leadership told the industry to go out and borrow the money they needed to take these over until the government could pay its bills. This advice came despite the fact that interest paid on these borrowed funds was denied by the same officials as a legitimate cost of doing business.

When Neil McElroy became Secretary of Defense

last fall, he made a valiant effort to straighten out the fiscal mess he had inherited from his predecessor. However, the problem was more acute than was generally realized, and Mr. McElroy's measures, instituted in an honest effort to solve the problem, are now about to be frustrated by a new flood of underfunding. Details of how this new financial crisis is hitting the Pentagon are reported on page 26 in *Credit Reactions*, AVIATION Week's Washington Bureau chief, Navy officials also have notified before Congress on the details of how their annual and involve procurement programs are being squeezed at this continuing financial level. Again the aviation industry is being asked by the government to wait for payment on work completed under legitimate contracts with the military services. "Delayed billing" is the polite Portuguese for the brutal financial practice.

Everybody from President Eisenhower to down guys that our defense effort must be strengthened and accelerated to successfully repel the Soviet challenge is in the spotlight. But how can such a goal possibly be achieved by paring fiscal policies that are viciously whittling the basic strength from our new weapons development programs?

Full Discussion Needed

We believe that this is a question that should be thoroughly explored by the Congress and the Administration together. It is a problem that so far has been pretty well kept within the confines of the Pentagon and the executive branch of the government. We seize a salute from the pin of the aviation industry, which is being most kindly ignored by these fiscal policies, to discuss them publicly. This is a grave mistake. Leaders of the aviation industry owe it to the American people to explain these fiscal facts of life on deficit. Only by a full and open discussion of these problems will any sort of intelligent and permanent solution be attainable.

The present Pentagon policy of single sweeping its fiscal debt under the rug of the next fiscal year is increasing the magnitude of the fiscal gap with each passing year. At the same time, it is eroding the military strength of our national defense program. It cannot be permitted to continue any longer without permanent and perhaps irreparable damage to the entire military posture of this country and its allies.

Whatever the hard fiscal facts of the Defense Department may prove to be, they must be faced squarely and solved effectively. Otherwise, we will peer the billions already appropriated into a vitiated effort that will fall significantly short of producing the defense strength required to maintain our position as a successful leader of the free world.

—Robert Flora



GO ... NO-GO

Automatic checkout equipment is a project of J. G. Ferguson, Senior Staff Engineer, Stavid Engineering, Inc.

Mr. Ferguson has specialized in the development of frequency synthesizers for Loran and other navigation systems, field transceivers, monitoring apparatus, earth station interface, and many other test and processing devices for electronic equipment. His current work at Stavid is to develop equipment for reliability studies and automatic testing of electronic systems. Mr. Ferguson is one of a team of Stavid scientists and engineers who are applying their knowledge... from concept through production... to projects of major importance to the defense and progress of our country.

In Stavid's objective engineering atmosphere, scientific development and manufacturing teams are producing a wide range of electronic systems for all branches of the industry. A typical project is the development of an Airborne Radar, Housing and Terrain Correction Radar System.

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• Missile Guidance Systems
• LASER Weapon System

Engineers and Scientists—Join Stavid's Advanced Electronics Engineering Team

Washington Roundup

Reorganization Lines Drawn

Despite Defense Secretary Neil H. McElroy's desire to explore, leaders said they are expected to support the President's defense reorganization plan and express opposition only if questioned by Congress. Real strides of the services toward the plan are reflected in statements by their associations. Navy League has in past strong opposition to most of the plan's features. After a study by its Airpower Policy Committee, Air Force Association last week pledged full support in principle "insofar as the plan represents a considerable advance in the direction of true cost-effectiveness called for in AFPA," but expressed concern about general aspects of the plan, "as well as about specific points." Army also is believed to be siding by going along with most of the points in the program.

AFA's greatest concern was expressed over planned management authority for the Secretary of Defense AFPA and its focus "concentration of policy-making as theory, but we consider it equally important to democratic faith due to its operations and detailed decision structure." The present interim administration of Pauline Lomax in AFPA is both sound and unobjectionable.

Proposed to establish research and development arm, AFA "cautiously gives accreditation." The association said Defense Department's role in this area "has been more fully responsive, rather than erratic," and said the transition has been toward systematic and responsible, more systematic means, rather than broad policies, existing and discontinued.

USAF Chief of Staff Gen Thomas D. White said one day later that he had studied the President's plan and "as I understand his proposal, I understand their implementation will be of great benefit to the security of the country. The Air Force is wholeheartedly in accord with the President's reorganization proposals."

Sputnik II's Return

Russia's Sputnik II earth satellite returned to earth in part so well with the main portion falling in the Atlantic Ocean near Barbados, S. W. L. A Naval Research Laboratory spokesman said that the rocket carrying was given two downings on an air base and its original point sea existence.

U.S. estimates of the size and weight of Sputnik II which have been made from tracking data indicate that the satellite weighed between 6,000 and 8,000 lb and is probably about 70 ft long, the approximate length of the complete Vostok rocket. These estimates made by U.S. agencies compared with the U.N. group agree with one made by Dr. John P. Hagan, Vice Chairman of the Joint Chiefs of Staff. The longitudinal cross-sectional area of the satellite body is about 500 sq ft. From the standpoint of aerodynamics, drag calculation, the equivalent cross-sectional area of Sputnik II in its terminal reentry is estimated at 180 sq ft.

No-Show Opposition

Airline's no-show penalty plan which, simultaneously, has been lauded as necessary by the Air Transport Association since its adoption last year, is now raising into strong widespread opposition from at least three airlines. At a recent meeting of the Air Traffic Conference, American Airlines, in early opposition of the plan, sent its vice

against a move to extend the plan until April 1, 1959. American was joined in its disavowing the no-show plan by Alaska Airlines and Midwest Airlines. The last, however, said it "is our responsibility" and the passenger's "is our own responsibility." A compromise move to extend the no-show ruling until Sept. 1 and to meet again on May 15 for further discussion on the plan was accepted unanimously by all members of the Air Traffic Conference. An Transair Asia Airlines on the plan denied from generic sources of the operation of the airline, but nevertheless indicated the plan was meeting very little passenger resistance and had naturally contributed to a decline in the number of no-shows.

Army Purchase Plans

Army plans procurement of 425 aircraft during fiscal 1959. These are 24 medium range bombers with a three-ton payload; 90 quarter-ton ground-attack; 140 fighters; 176 two-passenger reconnaissance; 160 three-passenger fixed-wing observation aircraft; 100 fixed-wing utility aircraft; 29 fixed-wing cargo assault with a 1-ton payload; special capability; new research and development aircraft.

Sen. Sam Nunn (D-Ga.) has initiated the all-sessions resolution for filing to ask Congress to authorize construction of more Polaris missile submarines in the fiscal 1959 defense appropriation request. Cohn said he would support the resolution, provided that at least six more Polaris submarines rather than the four already requested in order to provide what he termed the most effective and greatest security at the least cost.

Concentrating on missile submarine, where the U.S. already has a clear lead in production and experience, he said, can enable the nation to keep Russia development in check and return the balance of world power in the side of the free world and peace.

Progress vs. Secrecy

House Government Information Subcommittee is going on hearings on status and scientific information in space but takes no sign, except situation of "need to know," on how much information should be made available to the public. The committee's chief security concern, Rep. George E. Brown, D-Calif., said, is to "keep the doors open" to all information in the past, present or future.

Subcommittee, headed by Rep. John E. Moss (D-Calif.), contended that "the only real national security secret in scientific progress" and said this progress "has been severely hampered in exercise serious regulation administered by the Department of Defense." Report called the multiple clearance system "inefficient, unnecessary, and used to know" each "another barrier between the scientist and information he seeks."

CIA Transfers

CIA Administrator Adm. Arleigh A. Burke last week named Robert H. Harbin, director of its Office of Airports for the last five years, to serve as manager to the administrator for the development of the Washington National Airport. The new airport will be located at Chantilly, Va., and is scheduled to be ready to handle jet transports by early 1963. George R. Baier, 16-year veteran of the CIA Office of Airports, will succeed Harbin as Director of the Office of Airports. —Washington staff

New Financial Crisis Looms for Industry

Treasury Department cash stringency forces USAF, Navy to slow payments, skip schedules for delivery.

By Carl Brouhard

Washington—Aircraft industry may soon face a fiscal cut-off known as cash apportioning that will fall when government fiscal policies force rapid expenditure ceiling on a revised budget-cut plan.

After a brief pro-Spanish resolution on expenditures in the Budget Revision Act, Air Force and Navy Bureau of Aero sources are again being forced to stretch program priorities, often through cost shifting, substitutions and supplier demand bidding. The non-combatant Defense Department expenditure targets dictated by a lack of cash as laid out at the Treasury Department.

Cuts, spillovers and last week, that is far as industry is concerned, the storm can't be seen yet, but the struggle is beginning to blow off the roof.

USAFC, Navy Cuts

After effort by Air Force and BuAero of Appropriations Committee to forestall the fiscal year 1959 deficit until fiscal 1959.

Under present program, Air Force and BuAero could be faced \$1 billion short of having the money on hand to

meet their fiscal 1958 bills by the end of the fiscal year on June 30.

Now already, Air Force sharp cuts in its planned assault program (AVW April 18, p. 281) and last month's revised requires segments of industry to defer billings for hardware deliveries and allow the end of the fiscal year.

As fiscal year ended, new deliveries of aircraft were suspended in some cases, and this deferral is the first major cutback industry is asked to hold down spending since the beginning of fiscal 1959.

Payment Delays

Delay in the payment of bills on the part of the Air Force has not yet reached the peak of last October, but it is increasing—from a normal interval of 30 days to approximately 40 to Dec. 31. The delays in collection have increased still more in the last six weeks.

Under Department of Defense, Appropriations Committee and Senate, however, cuts in cash plus fixed cost contracts from 1958 to 1959 industry must finance the remaining 25% generally through local loans and debt of the final product.

If the ruling remains in force, and

there are no signs that it will be rescinded, an industry spokesman said last week that aircraft firms will have to abandon the cost-cutting normally followed if financing through bank loans.

There is no or little rationale to not how much industry has had to borrow since the rating went into effect in less than a year since it began during the spring of 1958. However, when Air Force was following a policy of paying prompt bills in late year, it was estimated that industry might be called on to borrow as much as \$1 billion, and the spokesman said, there just isn't that much money in the bank.

The alternative, industry, may be forced to turn to the market with long-term bonds such as the \$80 million issue recently offered by Douglas Aircraft Co. Smaller companies might be less capable of arranging refinance. They she might look at the risk involved since the heavy financing charges could pass to the nation if military payments failed to come through in time.

Change in Terms

The term expenditure target has remained by "expenditure ceiling" set by Defense Department during the same source date of last month and Dec. 31, to a degree, imposes less hardship on industry. The ceilings were established and on a customer-to-contract basis.

The targets are subject to at least one appeal by the service secretary to Defense and to industry to the original authority.

Rather than asking industry to take over a large portion of the financing of a particular contract as they did under the expenditure ceiling, the services are asking for delayed billings.

That isn't, a spokesman for industry said, that they are not then getting all the money fully disbursed. (Defense Secretary) McNamara promised they would not (AVW p. 281).

One area particularly hard hit is Series 206 contracts for procurement rather than aircraft. Air Force is having difficulty, for example, in finding funds to finance ground support equipment for the Convair F-106 and new models of the McDonnell F-101 series now coming off the line.

"We have several contracts in that line we want to let right now, but we just don't have any money to do it," said one industry spokesman.

The expenditure of fiscal 1959 will start no later than July and probably no later than August or September.

Concurrent appropriations for de-

fense, sponsored by the chairman of the technological panel, are generally sufficient to support the major defense programs.

The problem, and the reason behind the dragnet Defense Department expenditure targets, is a lack of available funds in the Treasury Department.

The economy in the nation's economy has not had an anticipated increase, particularly those large wave factors such as those of automation. This reduction is experienced out of its natural rate.

Caught with this is the fact that costs of federal programs are going up or up across the board basis.

An example—after this the increasing cost of defense—outlays to support the serial weapons program are not greater than the benefit payoffs when in producing them. Funds can't be supported a variety of projects are being absorbed by the federal highway program.

Unemployment benefits also are eating into the money on hand.

Debt Ceiling Increases

The Senate endorsed \$1 billion increase in the debt ceiling to help the Defense Department expenditure problem. At the same time, spokesman explained, "While Department of Defense gets the message to go back to the President and say the expenditure targets will have to be lifted, fiscal 1959 could be in bed as fiscal 1958."

"We're going to have," a senior official said, "an overall 1959 expenditure plan, drawing except it from fiscal 1958 and it's going to be a central problem."

Any substantial increase in the expenditure targets would require another boost in the debt ceiling. If the President agreed to ask for such a politically unpopular move, it still could come under terms of the forthcoming extended Congress.

Navy Annual Budget Short by \$6 Billion

Washington—Navy's proposed fiscal 1960 budget for the next 30 years—about \$6 billion more than its budget for the past four years. Vice Adm. Thomas S. Caudle, Chief of Naval Operations, reported to the House Armed Services Committee. He said:

• Aircraft procurement should be increased from \$3 billion a year to \$7.5 billion a year.

• Ship construction, comprising P-3C, improved submarine and nuclear-powered carriers, should be raised from \$1.5 billion to \$2.2 billion a year.

Adm. Combs pointed out that Navy's number of operating assault aircraft



Nike-Zeus to Resemble Nike-Hercules

Aerospace's conception of Nike-Zeus missile does like a will resemble existing version of Nike-Hercules, because a lot of that design would change over to better when missile is used against stealth. After the design an anti-missile missile vehicles with boost offboard, drawing draws weapon considerably after launch and separation of booster stage." (AVW April 18, p. 260)

with the advanced planes with which Russia has equipped both herself and her allies.

"To do so, additional funds must be devoted to aircraft procurement is coming soon."

In anti-submarine warfare, he said, "We have the added threat of Soviet submarine advances which he in the immediate future.

"At the same time increasing threat with a level of strength that provides only a marginal difference." The capabilities of our anti-submarine fleet, is a matter of urgency, he insisted.

Employment Decline Slows

Washington—More than 140,000 jobs have been eliminated in a full year of declining employment in the aircraft industry, but some observers believe the decline may level off by the end of next month.

Even with the sharp increase in defense contracting as compared with the last half of 1957, the industry probably would take a full year or more to regain what it lost in 1957, peak employment of 198,000 because of the long lag between initial contracting and volume production. Contracting figures for January and February indicate that the 1957 peak of 201,000 will not be matched again although the volume of work taken reflected to planned by Defense Department seems to be losing a considerable measure over the present number of jobs.

Downward trend that began in May probably will continue at approximately 540,000 jobs. At the beginning of last year, employment in February to February in 1957 was 15,300, according to Bureau of Labor Statistics figures. Then the income slowed considerably to 13,800 between February and March, and to only 500 workers between March and April.

From the April peak, the curve dropped steadily each month through July. At the new level first begins, and by eight subsequent periods, the job decline would be about 17,000 full between July and August, roughly about at that level—17,300—between August and September, and would again to 15,300 between September and October.

Bureau Detroit and Novosibirsk employment dropped by 41,000, taking an abrupt dip in the year. Since November, the division has been moving steadily. Novosibirsk's laborers drop to 27,000 jobs. Detroit's employment was 11,000.

The Bureau of Labor Statistics forecast, released, is expected to show a decline of 500 more than 3,000 from the previous month. An even steeper decline is expected in April, and May could see leveling off of the curve. Employment may begin to expand toward the end of June to early July.

Service Officials Back NASA Space Plan

By Fred Eastman

Washington—Landing Air Force, Army and Defense Department technologists strongly endorsed legislation last week that would transform an expanded National Advisory Committee for Aeronautics into a national space agency.

Supporting the proposal at hearings of the just-formed House Committee on Aviation and Space, Lt. Gen. Donald L. Petty, USAF deputy chief of staff for development, Herbert York, chief scientist of the Advanced Research Projects Agency, and Werner von Braun, Army Ballistic Missile Agency director of technical development, USAF support. Strongest endorsement was given by Gen. Petty who told the committee that Air Force fully supports the proposal. He said:

"NASA has much to offer in a national space agency outside the Defense Department."

The same thinks should be done in the space field as were done in the defense field to solve the problems of flight with a view toward their practical solution. And as discussing space technology it should be recognized that it is division, per se, between air and space

For all practical purposes we are space Nazis, lacking a continuum and inviolable field of operations. Thus, NASA should bear great responsibility in the aerospace business.

Gen. Petty reiterates space

as "the most important" he added. "Air Force has established a fine working relationship with the military services in the aerospace industry. It is time we expand it to the scientific data that would result from its research activities."

Two civilian space vehicle projects which also would have broadest peace value were Gen. Petty said are:

*A communications satellite for use

of saving and mapping the surface of the earth. By using space craft keeping an inventory of existing documents on the earth's surface. Mapping areas will be increased greatly. Military targets throughout the world could be plotted for greater security. We would have short-notice warning of hostile seismic activities on the surface of the earth, which in turn would prevent further damage and loss.

On the other hand, the communications satellite could prove a boon to weathermen everywhere. Thousands of hours and many dollars could be saved by previous knowledge of hurricane, typhoon and other severe weather."

*The second vehicle ... is the non-intersection satellite ... Several宇宙宇航器 equipped and having telephones could come at radio and television stations to receive signals from various points on the earth and relay them again directly to another satellite in any other point. Solar or nuclear energy power sources could give them useful life of many years."

Gen. Petty also publicly outlined Air Force plans which have not been reported to show a planned reusable capsule-vehicle to orbit around the earth by next year, (WAV April 7, p. 28) and preparation for launch pad of the moon later this year.

Both Air Force and NASA had then seen a general agreement with the proposed legislation to form the National Aeronautics and Space Agency with NASA as its nucleus but expanded responsibilities in respect to some programs.

The present NASA charter, then, is not designed to handle the present space program. The agency thus added should have more authority and responsibility than NASA had for the administration and be forced to operate as an autonomous agency. They also objected to no language that would limit or restrict its programs that have well-defined requirements.

Both von Braun and York said that such language would eliminate research on projects such as the nuclear thrust rocket and that they should have the authority and responsibility for the nuclear rocket engine, for which there is no considerable requirement but for which there will be in the future.

Von Braun and that if the U. S. wanted both there is a well-defined question before beginning research, the defense or civilian use. Werner von Braun also told the committee:

*Any could have placed a satellite in orbit to earth in 1951 or 1952 if the program had been approved in 1946. *The Secretary of Defense, in the future, will have presented to him what

in September, 1946, that traveled 3,360 mi. and reached an altitude of 500 mi., carrying a payload of 54 lb.

*Human basic research programs have

already created tremendous momentum and numbers similar to automobile culture.

the U. S. is in danger of falling behind unless more money is made available.

*Production methods and materials

in serviceable condition to the Secretary of Defense. Nixon promised that it will be kept separate; the Joint Chiefs of Staff from their day-to-day role in each of their services.

He challenged the President's plan to expand personnel of the Joint Chiefs of Staff—now limited by law to 216—and said it is operational staff must be increased the members of the chairman of the Joint Chiefs of Staff over 216. "The staff will become a problem," he said. "What would you do about it?" Nixon replied: "I am not going to implement it with an organization."

Hearing on the President's plan (WAV April 16, p. 29) and congressional opposition to it, Nixon, Rep. Leslie Alford (D-Ca.) and Rep. Paul Robins (D-Tex.) will visit the White House before the Armed Services Committee which Nixon heads.

Alford is visiting Washington on the committee and Robins in charge of the subcommittee on transportation. The Virginia Democrat Robins will denounce Republicans and fight the bill because the plan dividing Joint Chiefs of Staff and operating cost would affect his state.

"It is a good idea," Nixon said, "but it will mean more red tape and the cost will be too high." Nixon added: "The shape of the plan will enable us to maintain a strong staff ... to do our war and peace and to do our defense duties on behalf of our country."

The President said there would be "clear-cut civilian control of the military" by the Secretary of Defense, the Committees on Armed Forces and the Congress, all functioning without bounds set by the Constitution.

The Senate Armed Services Committee, the top ranking senator for his party, "I know we can't win by compromise," he said. "That is to say, overwhelming instrumental attachment to establish only civilian control and command." He in pointed his call for "one single basic strategic plan" scheme under single direction.

Although the President's plan gives the Secretary of Defense supervisory authority, Nixon said that on problems it would be dominated by the general staff.

No Secretary of Defense has the ability, the knowledge, the experience, the time, the strength and the wisdom to assume the operational control of the entire nation's establishment," he insisted and added.

'Unilateral Thinking'

The concept of this staff makes a unilateral thinking at all times. There can be no room for decisions, no room for debate, no room for advice and advocacy, no room for split opinions. From this comes the most complete control and, undoubtedly, an ability to fight a war based upon a single concept.

"The Secretary of Defense, in the future, will have presented to him what

the President's plan to have the Joint Chiefs of Staff work under him. Nixon said this was due to opposition to the Secretary of Defense. Nixon promised that it will be kept separate; the Joint Chiefs of Staff from their day-to-day role in each of their services.

He challenged the President's plan to expand personnel of the Joint Chiefs of Staff—now limited by law to 216—and said it is operational staff must be increased the members of the chairman of the Joint Chiefs of Staff over 216. "The staff will become a problem," he said. "What would you do about it?" Nixon replied: "I am not going to implement it with an organization."

Supreme High Command

The President's proposal to change supreme administration of the three services, Nixon said, "for all practical purposes eliminates them as distinct entities ... and enhances the power of the Secretary of Defense. The net result is greater concentration and less decentralization, both to the general and combat capabilities of one individual Secretary." Nixon added that the Secretary's authority will be delegated to "as far as possible" powerful agencies concerned. The Secretary will have overall power."

Nixon called for "responsible decision making" at the level of efficiency. Inter-service competition, he said, is a terrible hindrance—that each service demands the funds necessary to fulfill their important mission in our system.

Dyna-Soar Proposals

Washington—Phase I contract for the initial development of the Dyna-Soar manned orbital heavier payload will be awarded to the Air Force within a week. Two contractor teams may be selected to pursue parallel designs and the final choice will be made after Phase II or the hardware stage is reached.

The following companies or combinations submitted Dyna-Soar proposals to Air Force before April 16 in response to an Air Force request made about a month earlier:

- North American Inc.
- Combination of North American Colossus and Rockwell International
- Douglas Aircraft Co.
- Convair Division of General Dynamics Corp.
- Combination of Martin, Bell, McDonnell-Robins
- Combination of Republic, General Electric, Military and Guidance Systems Division, Thiokol and Systems Corp. of America

General Electric made no independent proposal and Westinghouse was not awarded in any way although it is believed that this was invited to participate. Thiokol is associated with several of the proposals other than the one submitted in conjunction with C. E. Gribble and Systems Corp. of America.

North America's Colossus Division was active in the competition other than the long-lost Division because of the latter's last-stage propellants on the 195-2300 chosen budget project.

Proposed entries for the Dyna-Soar would be a three-stage rocket model to the 100-2300 stage weight. Several of these entries would be checked and the glide vehicle mounted on top. It would separate from the Dyna-Soar third stage at 100-2300 sec.

Space Organization

Washington—Hugh L. Dryden, director of the National Advisory Committee for Aeronautics outlined how work steps should take, as scheduled to be taken, that would transform NASA into a national civilian space agency.

*We are making plans with regard to an internal reorganization and somewhat different to insure that the organization will be more effective in the sense of being able to grow and effectively accomplish its mission. Although these plans are not final, they include new elements as well as the effects of operating experience gained in NASA over the past 40 years.

*We are also planning additional offices and programs that will be necessary to assure rapid implementation of the proposed legislation. This includes determination of the requirements for additional staff, facilities and funds that will be needed.

*We have undertaken with Defense Department officials the review of posturing space programs currently underway or planned by the department, including their priorities.

*Flight test space requires that great effort also be devoted to such problems as electronics, guidance, and photometry or human factors. We plan to make whatever facilities are available, of significance since component needs demonstrate in these related fields of technology.

*We will work closely with National Science Foundation and the National Academy of Sciences as well as other governmental and non-governmental bodies, particularly the scientific community on a continuing basis in the planning and coordination of sensible programs for the use of space vehicles.

*We have made start of one posturing function from the beginning of the program, which are now reaching its mid-point stage. These can be used early to propose XCOR and satellite, but to find time resources and then assign tasks out to the most cost efficient.

*The work we are doing on structures and structural materials is now being defined as problems peculiar to missiles and vehicles that travel beyond the earth's atmosphere."

How Aviation Industry Fared Financially in 1957

These are the same principles as building blocks from common general concepts and their individual theoretical differences are relatively small.

COMPANY	SALES		NET EARNINGS		NET INCOME PER SHARE		NET EARNINGS		BACKUP	
	1987	1986	1987	1986	1987	1986	1987	1986	1987	1986
Acme Standard	\$10.4 billion	\$12.7 billion	\$0.4 billion	\$0.5 billion	\$0.38	\$0.40	\$0.38	\$0.40	3.3%	2.6%
Aerospace	7.1 billion	7.1 billion	1,000 million	1,000 million	9.14	9.14	9.14	9.14	113.000 million	240.000 million
Alcoa	12.0 billion	12.0 billion	1,144 million	1,144 million	5.64	5.59	5.64	5.59	114.250 million	172.245 million
Aluminum	1,200 million	1,200 million	36.1 million	32.1 million	2.24	2.44	2.24	2.44	1,200 million	1,200 million
Armco	70.5 billion	55.0 billion	44.3 billion	38.0 billion	6.40	5.60	6.40	5.60	110.000 million	100.000 million
Avon Products	1,200 million	1,200 million	1,000 million	1,000 million	4.50	5.10	4.50	5.10	1.200 million	1.200 million
Bell & Howell	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Bethlehem Steel	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Brown & Root	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Caterpillar	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Chemical Specialties	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Continental Can	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Conoco	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Dow Chemical	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Eastman Kodak	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Exxon	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Ford Motor	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Gulf Oil	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Hercules	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Imperial Chemical Industries	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
International Harvester	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Kodak	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Lever Brothers	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Marathon Oil	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Merck	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Mobil	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
National Gypsum	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Occidental Petroleum	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Petroleum Refining	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Phillips Petroleum	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Plumbco	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Procter & Gamble	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Rhône-Poulenc	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Schlumberger	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Standard & Poor's	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Standard Oil	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Texaco	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Union Carbide	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
U.S. Steel	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
W.R. Grace	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Westinghouse	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Whiting Petroleum	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million
Yankee Gasoline	1,200 million	1,200 million	1,000 million	1,000 million	2.20	2.20	2.20	2.20	1.200 million	1.200 million

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Final exam Study Guide 10

SEC Reports Industry Salaries

Washington—Following is a list of executive salaries, impressive awards and stock holdings as filed by aircraft companies with the Securities and Exchanges Commission:

BOEING AIRFLYNG CO.—8. B. After
a short and difficult flight over water, the
two passengers and their pilot were shot down
near Rock Hill, S. C., about 11:30 A. M. The
plane had been flying at 10,000 feet when
it was hit by a bullet from a machine gun.
J. P. Ladd. (See preceding)

French Back Taon

Penn-Foster's 10 months has decided to have the Budget Team as the possible second generation." NAFD lightweight editor Robert A. Connelly says that a point the author Penn Foster makes about the decision to keep the Budget Team intact the decision was based largely on the fact that the Dutchess had a self-contained, well-termed budget that may have obstructed much of the Budget's agenda. Also, the Budget Team reportedly will be easier and cheaper to build than the Empire State, which required a massive assembly of 12,000 people. The 12th Budget Team genesis presented by the Board of Governors will be presented by the Dutchess 12 on next spring.

announcements and direct to the FBI offices in Washington and to the various state police agencies. It is believed that present about 100,000 stations in the United States are equipped with telephone facilities which will enable them to receive messages from the FBI and other radio stations.

40 000 individuals under 10 years old were interviewed in 1970. The survey was conducted by telephone and the interviewers were instructed to ask questions about the child's health and growth. The survey was conducted in 1970.

Spaeth (Continued from page 100)

selection of managers and no selection after 1910, it seems a safe assumption.

adult Indians of the tribe considered their skins and furs as valuable commodities. The Indians were very fond of skins and furs, and the skins of the most valuable animals were highly esteemed. The Indians of the tribe were very fond of skins and furs, and the skins of the most valuable animals were highly esteemed. The Indians of the tribe were very fond of skins and furs, and the skins of the most valuable animals were highly esteemed.

Sparrow II Missile Production Ordered

Royal Canadian Air Force and
partner of Defense Production
will extend the Sparrow III into an
anti-gradation for the RCMP Air
Force Component.

Canadian Zinc will produce the zinc under license from Dongguan Zinc Co. Sales distribution will be by Hydro-Quebec, Hydro-Quebec Inc., Géca de Canada Ltd. and Sideris Corp.

Aircraft Sparrow III was a solid propellant experimental missile and was intended for operational use by United States forces.



Two-Place F-106B Makes First Flight

First flight of Convair F-106A, Standish aircraft version of F-106A Delta Dart, has been completed at Edwards MFB. Calif. Gun port is partially enclosed in the front plate, taken in a dark flight for 40 min. and 8 sec. Preliminary details were not provided, other than that flight was uneventful. F-106B is identical to F-106A except the second seat and longer cockpit canopy (about 1 ft). Forward section of the tail section pt lights are fabricated in the same manner as the fuselage. The two aircraft are being used by the 4th Test Wing, Tenth Air Force, at San Diego, Calif., for testing with remaining service. Both aircraft are powered by Pratt & Whitney, Calif., J57 turbojet producing 23,000 lb. thrust (AVW News p. 5, 52).



Ground Tests Start on Vertol 107

Above: drawing of aircraft proposed Vertol Model 107. Indicated under definition which is undergoing ground trials at Philadelphia International Airport. Model 107 is powered by Lycoming T53 turboshaft turbines, a about 5 ft. shorter than the H-33 (AW April 7, p. 34). Aircraft will carry about 20 persons.

Douglas Quarterly Earnings Decline

Solidly lower earnings during the rest of the year were forecast last week by Douglas Aircraft Co. which reported to stockholders that earnings for the DC-8 jet transport division in the first quarter of this year compared with last year.

Approximate assumption of DC-6 and DC-7 production plus the DC-8 unit-cell, which totaled \$1,830,000 during the first quarter, will cause the profit decline. Inventory write-downs for the fast DC-8, which cut the month's gross by 10%, contributed a \$2,768,000 item to the total.

First-quarter earnings declined more than from \$8,773,900 to \$8,996,000, but sales increased from \$271 million a year ago to \$312 million. Earnings dropped from \$2.37 a share to \$2.32.

Other companies to report earnings:

- **Vought Aircraft Corp.** Sales and earnings declined to 1975, sales from \$82,875,697 in 1976 to \$72,771,407 and earnings from \$5,912,503 in \$1,616,788. Per-share earnings dropped from \$6.34 in 1976 to \$1.75 last year. Profits for production in 1978 are modest, the company reported said, but the earnings trend has strengthened in the second quarter.

- **American Airlines Inc.** Estimated actual earnings of \$1,065,378 for the quarter ended Mar. 31, a drop of 54% from under a similar period last year. Total revenue for the quarter was \$71,847,495 against \$70,751,104 a year ago. Net earnings, including profit of \$753,000 from disposal of property, was equal to 19 cents a share in the current period.

Airline earned 1,827,037 passengers 3,145,997,200 revenue on

News Digest

Canadair Aircraft Engineering Corp.'s CH-11F IF broke an official world altitude record by reaching 76,225 ft. Super Tigre, powered by a General Electric F79 engine, was flown by Lt. Col. George Warken from Edwards AFB, Calif. National Aeronautics Assoc. of Canada set the record in Federation Aeronautique Internationale for 10 years.

ACF Industries Nuclear Products-Eco Division will build 75-ton mobile launchers for Northrop SM-62 Snark missile under a \$1 million contract. Combined carrier and launcher consists of a 16-wheel chassis on which a pair of hydraulically powered jacks are mounted to raise the missile for launching.

An Army-Navy contract totaling up to \$10 million for "a single production version" of T33A-1 turbine powerplant for Bell HU-1 and Kaman H-45D helicopters has been awarded to Boeing. Flight deliveries are scheduled to begin January 1979.

Lockheed Aircraft Employees voted to strike, beginning in August, over wage demands. Strike notice has been served at Lockheed Long Beach plant and North American, Douglas El Segundo, Torrance and Santa Monica workers were ordered to take a strike vote. Approval is expected at Lockheed, Douglas and North American contracts all have expired, but work continues during negotiations in a contract of management's own devising.

Brent Bernson, marching transport pilot has been nominated by the Vice Office of the Civil Aviation Administration. Violations reported by the CAA were present in the cockpit and landing gear area. British Air Registration Board commented with CAA's representations for engine modifications although requirements on landing gear changes reflected in several agreements between the two agencies.

Lockheed F104A and F104Bs, including those operational at Hamilton AFB, have been released for flight after being grounded for investigation of afterburner sluggishness in F76-1A engines.

Ferry Kotakovic made first trans-tasman flight in helicopter to antigen light on th 7th test flight. Aircraft climbed to 4,000 ft. and transferred power from rotor to propellers, after proceeding straight and level with auto-rotating, power was transferred to the rotor and a vertical landing was made.

American Strike Date Postponed by ALPA

Washington—Threatened strike against American Airlines by Air Line Pilots Assn. was postponed last week, while both sides agreed to further mediation efforts requested by the National Mediation Board.

Airlines had set strike date at midnight April 16, because of charges by the union that American had failed to meet standards of its competitors in pay, rules and working conditions.

In accepting the Board's request during the final hours before the strike deadline, the Pilot union emphasized it was avoiding "unjustified" strike.

Both American and the union left

with the National Mediation Board in New York Thursday to continue discussions.

arctic performance

Men of the RCAF—aircrew and groundcrew alike—have learned to work with the demanding elements of the Arctic. Their resourcefulness and courage are our greatest assets in maintaining our wide-ranging defence systems.



AVRO AIRCRAFT LIMITED

MONTREAL - CANADA

MEMBER A.V.R.O. CANADA LIMITED & THE HAWKER SIDDELEY GROUP

How U. S. Steel Supply's Any Steel, Anywhere, Any Time Service

"cut our inventories"



Reported by
Mr. G. R. Campbell,
Material Manager,
Pete Aircraft Corporation,
Chula Vista, California

Maintaining large inventories plagued our production area," says Mr. Campbell. "However, as a result of U. S. Steel Supply's **Any Steel, Anywhere, Any Time Service**, these inventories have been cut considerably.

"Our orders are placed with U. S. Steel Supply's Los Angeles plant and, when necessary, can be filled and delivered on an overnight basis. No longer are we forced to negotiate our needs for more than a 40-day period, as compared to the 90-day minimum advance period for mill orders."

"U. S. Steel Supply's **Any Steel, Anywhere, Any Time Service** gives us instant delivery on any type or grade of material, eliminates mill 'lead time' and helps solve our 'time overstock' problems."

You, too, can benefit by this service!

Let one of our representatives show you how other steel users are saving money and increasing profits as a result of **Any Steel, Anywhere, Any Time Service**. There's a good chance he can help you eliminate idle equipment, increase production and cut inventory costs. Write to U. S. Steel Supply at the address below.

Remember ... you get **Any Steel, Anywhere, Any Time Service** from ...

U. S. Steel Supply
Division of  **United States Steel**

Shipping Address: P. O. Box 1566, Dept. K-18, Chicago 90, Ill. General Offices: 300 South LaSalle Street, Chicago 4, Ill.

AIR TRANSPORT



Fairchild F-27 Flies for First Time

First flight of Fairchild F-27 turboprop transport listed 25 min. 46 sec. Four passenger, pressurized aircraft is powered by Rolls-Royce Dart engines and has cruise speed of about 300 mph. Range is more than 1,000 mi. Number of 1254 on order a 50 minutes on option is 25. Frontier Airlines, which has not announced any routes for the F-27, was of a view involving the F-27 and two models of the Vickers Viscount. However, as a result of capital gains legislation which permits airlines to sell wholly or partially owned capital assets from the date of ascent at the profit rate used to purchase additional assets (AW Sept. 16, p. 45). Frontier is currently operating Douglas DC-9s.

New Routes Proposed for Three Lines

Washington—The Civil Aeronautics Board's Bureau of Air Operations has recommended new air routes for American, National and Eastern Airlines and Southern Airways in answer to a Defense Department request for letter of intent to four of the nation's main airports.

Defense Department filed a motion to expedite portions of the complained South East Local Service Center in the interests of national defense. Specifically, the military said it proposed route 100 between Atlanta and Miami, Miami and Atlanta, Atlanta and Guided Missile Agency's Eglin AFB, Fla. (Air Proving Ground Center), Patrick AFB, Fla., USAF Missile Test Center and Ft. McRaeles, Arkansas, Arkansas.

Council Recommends

Bureau Counsel V. Rock Goodman recommended that National serve Melbourne, Fla., via Tampa. The Defense Department and Patrick could each gain route 12,000 passengers per year for 10 years. National would serve Melbourne on the intermediate points of Tampa and Eglin to provide improved service to MacDill.

Eastern would serve Melbourne as an intermediate point between Vero Beach and Orlando.

Southern Airways was recommended to serve Houston as an intermediate point between Atlanta and Miami. Atlanta would serve Southern via Dallas.

Local Service Opposes

If approved by CAB, these recommendations could provide the military with their greatest need in the areas mentioned—better air-sea rescue contacts between the two airports and communications, located both on the West Coast and the Detroit area.

However, some local service carriers can be expected to oppose the recent recommendations. Trans Texas Airways has filed a brief with commission Paul N. Pfeiffer requesting Houston on the grounds that the airline could operate a site sufficient to reduce traffic and, using increased DC-9s, offer fares lower than those could be provided by Southern.

To substantiate its "no intent" position, Trans Texas cited the operating costs per passenger of \$19.50 each, as compared to Southern's \$17.29 each.

Southern admits it would need extra subsidies for the route and operates that

the annual cost of operating the route would be \$126,114. A load factor of 55% over the route would put all three operators in the red, Southern said.

The cities affected by the recommendations are each those considered as vital to national defense and requiring immediate action. Defense Department bid earlier included such points as Ft. Butler, Ala., Turner AFB, Albu, Ga., and the Arnold Engineering Development Center, Tullahoma, Tenn., in the overall South East Local Service Case.

Routes 87 Extended

Bureau Counsel Crowley also has notified the Terminal Bureau of Air Services that Frontier Airlines will be recommended to the commission to serve that state by extension of its Route 87 to serve a Ta-Gang-Mandalay route through the intermediate points of Kawthaung, Chittagong, Tiddilapu and Shillong, Nagaing and Jidong.

Southern Airlines has applied for a Bhamo-Mandalay route on the basis that it already serves the main flights from Mandalay and could then offer the best service to all Transavia passengers and Ta-Gang.

The local service airline said it can operate all of the route without any subsidies.



BOAC's First Comet 4 Rolls Out

Each of 18 Comet 4 jet transports under evaluation for British Overseas Airways Corp is fitted out at de Havilland Aircraft Co's notable plant at Hatfield, England. Aircraft which next will get full fire check and engine runs has been Kells-Kayes Avro, English, at 30,000 ft. In flight each engine can withstand full Rolls-Royce and some other manufacturers' OAW runs. (See, p. 86)

Executive Pay, Bonuses for '57 Reported to CAB by Airlines

Washington—Following is a list of airline officers and their salaries becomes valid against compensation, expenses and steel holdings paid airline officials for the year ending Dec. 31, 1957 as reported to the Civil Aeronautics Board. The list also includes compensation paid by the airlines for services rendered good by the airlines for services rendered.

Tech Airlines

Pen Argyl - The Pen Argyl division is the largest and oldest of the 2000+ sales units. It is the oldest division in the U.S. It has been in existence since 1890. It is the largest division in terms of sales. It is also the largest in terms of employees. It is the largest in terms of sales per employee. It is the largest in terms of sales per employee. It is the largest in terms of sales per employee.

B. E. Hausemer, received the postgraduate degree in electrical engineering from the University of Illinois in 1964. He is currently working toward his Ph.D. degree at the University of Illinois. His present research interests include the development of methods for the analysis of large-scale power systems.

uses for '57
by Airlines

lower and reduced compensation expenses paid by companies and 200 more hours of insurance available in Canada. The new rates will take effect January 1, 1957.

Reduced compensation expenses paid by companies and 200 more hours of insurance available in Canada. The new rates will take effect January 1, 1957.

and 1970, expressed as the percentage of the total number of cases which had been reported by the time of the survey. The results are shown in Table I. The percentage of cases reported by the end of the year in which they occurred was highest in 1968 (75·0%), followed by 1969 (71·0%) and 1970 (68·0%).

the number of insurance cases. It is important to note that the number of insurance cases has been increasing steadily since 1950. The number of insurance cases in 1950 was 1,000,000, and in 1960 it was 1,500,000. In 1970 it was 2,000,000, and in 1980 it was 2,500,000. The number of insurance cases in 1990 was 3,000,000, and in 2000 it was 3,500,000.

is a point for temperature and time which is good for the growth of *S. B. B.* *Bacillus* and the production of *lactic acid*. At this time the *lactic acid* produced by *lactobacilli* and *lactococcus* is about 300 mg per liter and 200 mg per liter respectively.

total held to end 2000
and 2001, and the
annual growth rate was 4.6%.
The growth in the number of
clients is consistent with
the growth in total assets
over the same period.
In 2000, the firm had
1,000 clients with assets
of \$17.5 billion, and
in 2001, the firm had
1,050 clients with assets
of \$20.5 billion.

H. P. Petty, vice president—High-speed division, \$1,000 salary, will be responsible for all aspects of the high-speed product line. **C. Becker**, vice president—Sales and marketing, \$1,000 salary, will be responsible for all aspects of sales and marketing of all products. **J. M. Anderson**, vice president—Research and development, \$1,000 salary, will be responsible for R&D efforts of the company. **R. E. Johnson**, vice president—Finance, \$1,000 salary, will be responsible for company's financial affairs.

H. W. Becken, 1010 Franklin Street, San Francisco, California, salary \$10,000; expenses paid by company and an amount of \$1,000 additional to provide him with a home. **E. E. Blanchard**, 1010 Franklin Street, San Francisco, salary \$10,000; expenses paid by company and an amount of \$1,000 additional to provide him with a home.

Local Service Airlines
Allegany Airlines Inc.—E. G. Barnes, president and director, 115 Main street, Binghamton, N.Y. 13901. Capital stock \$1,000,000; surplus \$1,000,000; assets \$1,000,000; liabilities \$1,000,000. E. G. Barnes, director, 115 Main street, Binghamton, N.Y. 13901. Allegany Airlines Inc. is engaged in the operation of passenger and cargo services between Binghamton, N.Y., and the surrounding areas.

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AT SERVICE AIRLINES

American serves you better when you fly



MERCURY SERVICE NEW YORK - LOS ANGELES

most frequent DC-7 nonstops - America's fastest airliner

radar-equipped · reserved seats
spacious lounge - superb cuisine
American's famous
stewardess service



paid out by the company and four hours of time were expended. BLDG 6000 salary \$100 was paid plus \$100 as expenses and no travel. BLDG 6000 salary \$100 plus \$100 as expenses paid plus \$100 as travel. 60 day expenses paid plus \$100 as travel. BLDG 6000 salary \$100 and costs of travel \$100 each. E. G. Klemmeyer, secretary and treasurer, BLDG 6000 salary \$100 plus \$100 as expenses paid plus \$100 as travel expenses each day. BLDG 6000 salary \$100 plus \$100 as expenses paid plus \$100 as travel expenses each day.

For the third quarter, Homecrest reported a revenue of \$1.05 billion, up 10% from \$950 million in Q3 2019. Gross profit for the quarter reached \$100 million, up 10% from \$90 million in Q3 2019. Homecrest sold 1,000 units of homes in the quarter, up 10% from 900 units in Q3 2019.

—
—

10. The following table shows the number of hours worked by each employee.

W. H. G. (William H. G.)

...and the first time I ever saw a real live black bear.

...and the first time I ever saw a real live gull.

—
—

—
—
—

Soviet STOL An-14

Makes First Flight

lower onto ground (left) and prevent parrotineslosion designed by D. R. Anthony is employing shortage flight test on his parrotines. Another advantage of the new system is that it can be used in conjunction with the existing ground-based systems.

The aircraft is designed to take off and land on roads up to 300 ft. in length. Airstrip proposals for use as an agricultural or ambulance vehicle, and as a short-haul transport is equipped for instrument flying. As a transport, the aircraft can seat seven. Landing skids of vented tail surfaces are swept back on metal depicted here; lead ing edges of vertical tail surfaces are swept forward. The An-14 has a maximum speed of 294 mph.

maximum speed of 128 mph, has a flying speed of 23 mph. Peltella is expected to be well suited for operating from Romeo's more numerous airfields. (AMM News, 4-22-71)



JET AGE EXPERIENCE



To serve you now and in the future

Deflators on energy will make real demands on parasitic suppliers. With fuel requirements increased four-fold by the huge commercial jets, will your supplier be able to provide proper service in the difficult years ahead?

Twelve years ago, East pioneered and developed high-speed hydronic heating. This spacious and solar method is now available at many major international airports—a practical and efficient solution for the increased fuel capacities of today's large piston-engined airliners and

A good sign to fly with...



average, \$2,875 salary, 31,832 expenses by company and 116 shares of common stock. E. B. Moore, engineer, no salary, 10,000 bonus, additional compensation derived from his 1/2000 share in company stock. G. R. Farnham, architect, no paid-in capital or dividends, 10,233 salary, 10,000 bonus, paid by company and 116 shares of common stock. E. L. Green, president treasurer, \$2,075 salary, 1000 shares paid-in company stock and 116 shares common stock. G. F. Tolson, accountant

shares of common stock. Frankfort, Ky., is the home office of The Franklin-Fulton Automobile Insurance Company, F. T. Fulton, Pres.; Max J. Rau, Vice-Pres.; John W. H. Hauseman, Secy. The company has \$100,000,000 in assets and \$100,000,000 in policyholders' surplus. It is a member of the American Association of Life Underwriters and its agents are: A. C. Bryan, 24,000 shares; Wm. B. Johnson, 10,000 shares; and W. W. Jones, 10,000 shares. The company also has 100,000 shares and 100,000 shares of preferred stock outstanding. It is a member of the Kentucky State Automobile Association.

and, until 1960, compensation was paid in shares of the company's stock. In 1960, however, the company introduced a cash-based compensation plan to its stock. It gave greater discretion and autonomy to HR managers and enhanced compensation alignment. The result was significant performance improvement, with a 10% increase in revenue over the next five years.

AIRLINE OBSERVER

Benevento station, coupled with unfavorable weather and a poor Florida winter season, lost the stations land in March. Lead losses for the month ranged 58.6% for descriptive translators, 2.16 points below that for March, 1917. It was the with the consecutive month that lead losses have held since 1917, the longest point was March, 1918, that lead losses were maintained at such a low figure. Percentage gains for the 12 months ended in March, 1918, amounted 10.41%, the lowest percentage increase recorded by the press since the previous year's record of 10.40% in January, 1916. Such entries confirm its decline. In March, 39.21% of all radio stations in the U.S. had merged with P.E.C.I. in March last past and 55.54% in March, 1918.

Watch for a appeal of an Australian federal government decision designed to prevent Ansett Airways from taking delivery of four Lockheed electra on order (AW April 14, p. 43). Australian government presumably had made an attempt to reject applications for the importation of the electra and force the airline to purchase Vickers Viscounts. Strong pressure by Ansett against the action, coupled with threats to take the case

Post Office Department has issued four schedule airfares in protecting transport requests for subsidy by Riddif, Andros and St. Lucia Airways (AW April 13). Post Office is claiming that subsidy for the two routes will not be in the public interest. In protecting AW's position for subsidies, Riddif and the Civil Aviation Board said that it has "no much reason" to apply for such aid; it would be ridiculous to impose new fares. Riddif, S. said yesterday.¹⁴

United Air Lines has taken exception to Civil Aeronautics Board Engineers' initial decision recommending that TWA's nonstop flight between United and the summer had not been issued airmail letter rates on the basis that it established a sensible line. United charged that because of the extra space needed for a configuration, amounting to 30% more space per cabin passenger, TWA could charge an extra \$47 on each one-way transcontinental ticket.

An F-105 will return to Boeing and Prater's Whittier test performance lab competing during the KC-135 at Mach .85, 45 miles S San Diego, Calif., to the Anzares (AW April 14, p 17). Brig. Gen. William B. Smith, flight commander on the flight, and the data will be of particular interest to aircraft placing jet operations. He used tankless fuel cell of the aircraft was 10,300 lb with 50,000 gal of fuel. Temperature at takeoff was 40° F. Average air density at 44,400 ft was at an average speed of 598 mph. Original plan to Mach 1 was canceled at the Anzares because of the original jet

Civil Aviation Board is considering two measures designed to accommodate the special needs of foreign tourists in the U.S. In order to share demands by U.S. airlines for more accurate traffic data in order to share benefits tourism offices are putting forward bilateral agreements with the U.S. The Board has suggested that passenger tickets be photostatic to track travel origins and destinations of each passenger. Board also has asked Bureau of Budget to change an Immigration and Nationalization Classification so that passengers would be required to report routing, route and destination.

International Civil Aviation Organization has established basic principles to be applied by governments deciding to adopt user charges against airlines. Essentially, principles call for equal charge or costs of all countries holding the country imposing the charge. Charges should not be imposed if facilities not required nor should they be established in such a way as to encourage use of facilities essential to safety.

Department of Defense and Civil Authorities Board staff members have included a free-flow meeting as an attempt to settle controversial issues of misuse use of civil air assets by the Department of Defense.

**REPORT ON
TRAFFIC PATTERN AND
GROUND HANDLING**



THE PRODUCTION SCIENCE 707, as demonstrated by 30 years of test-flying of the prototype model, operates with ease in existing traffic patterns of commercial airports. Though it cruises at 600 mph, the Boeing jetliner's maneuvering speeds and strip requirements for holding and instrument approaches are the same as those of today's conventional aircraft.



GROUND-HANDLING poses no problems. For ground-handling and taxi characteristics of the 707—and its shorter-range sister ship, the 720—are equal to, and, in some cases, better than those of present-day commercial transports.



MANUALLY OPERATED flight control systems, proved out in hundreds of Boeing jet aircrafts, give outstanding control and stability "feel" through all flight conditions. In rate of descent, the 707 and 720 offer greater flexibility than present airliners.

SPINDLES ACT AS AIRBRAKES, give added control over rate of descent. In all areas of operation, the Boeing 707 and 720 benefit from the unequalled experience Boeing gained building more than 4600 large, multi-engine jet aircraft.

BOEING
Family of jet airliners

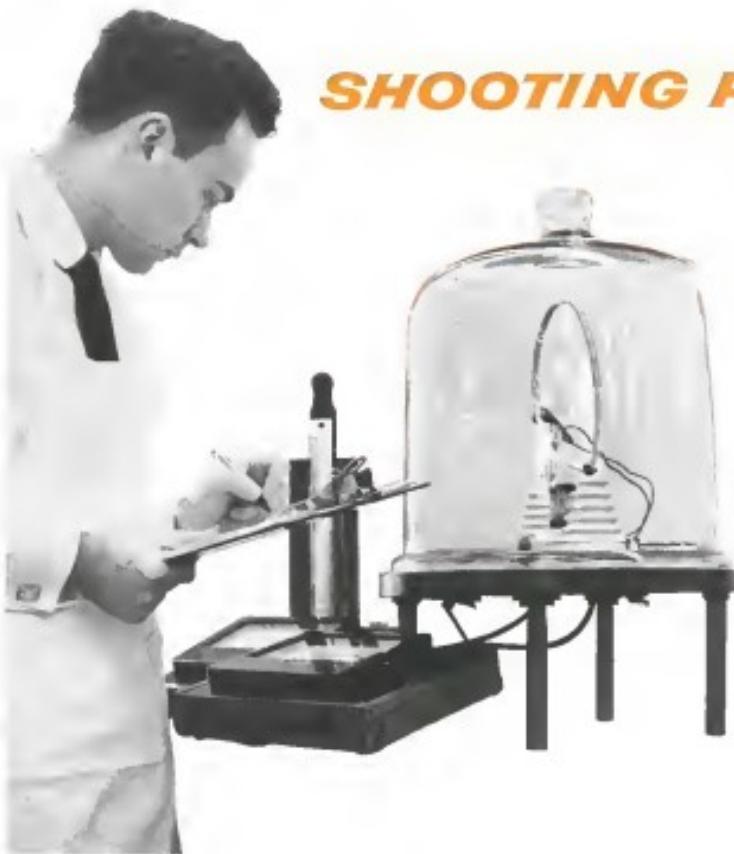
These aircraft already have ordered models of the Boeing family of jetliners:

AIR FRANCE • AMERICA'S AIRLINES • C.G.A.L. • BRAZILIA • CONTINENTAL • CZECHOSLOVAKIA • PAN AMERICAN • QANTAS • SABENA • TWA • UNITED • VARIG

Airline Income and Expenses—January 1958

	Passenger Revenue	Mail Revenue	Baggage Revenue	Flight Revenue	Other Revenue	Total Operating Revenue	Total Operating Expenses	Net Income (before tax)
DOMESTIC TRUNK								
American	\$61,270,000	Neg.	Neg.	\$1,410,000	Neg.	\$62,680,000	\$62,160,000	\$4,520,000
Braniff	4,161,377	100,311	21	131,781	842,402	5,093,450	4,797,812	295,638
Capital	7,409,717	178,211	20	281,420	12,426	7,689,543	6,162,740	-1,526,819
Continental	1,748,294	42,842	17	74,741	27,488	1,828,524	1,749,487	79,038
Delta	1,521,141	14,111	14	61,111	1,000	1,582,252	1,521,141	61,111
Eastern	20,295,471	204,783	845,857	235,714	Neg.	21,840,146	20,840,416	1,004,448
National	2,171,348	42,267	18	100,827	83,078	2,340,445	2,140,337	190,108
Northwest	1,404,161	14,143	15	61,443	6,448	1,470,652	1,372,342	97,310
Pan Am	1,743,712	185,554	200,000	200,000	2,730	1,936,454	1,872,347	64,107
Twa World	14,707,000	322,000	790,285	627,000	Neg.	16,199,000	15,799,000	400,000
United	18,750,421	207,102	275,412	840,153	39,197,456	21,052,726	20,420,485	-420,241
Western	1,566,348	95,184	54,171	74,107	6,238	1,726,707	1,640,385	82,322
INTERNATIONAL								
American	402,212	5,406	119	61,469	Neg.	408,681	311,799	93,882
Brussels	382,500	100	10	33,077	Neg.	382,500	372,500	10,000
Caribbean Airlines	177,455	2,276	4,787	4,787	150	186,299	147,854	38,445
Delta	227,821	4,176	9	9,137	Neg.	236,471	195,074	71,398
Domestic	1,446,161	34,500	10,000	10,000	Neg.	1,476,671	1,407,361	72,310
Eastern	1,404,161	14,143	15	61,443	6,448	1,470,652	1,372,342	97,310
Northwest	1,342,712	185,554	200,000	627,000	11,474	1,513,204	1,420,385	91,819
Pan American	4,116,580	11,000	34,000	326,800	Neg.	4,250,380	3,991,161	-159,800
United	3,410,000	607,200	751,000	344,000	2,741,144	3,211,000	2,741,144	-10,000
Lat. Am. Airlines*	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
Pan Am	6,116,580	487,000	540,000	381,000	3,038,261	3,038,261	3,038,261	-162,000
Pan Am	2,142,000	4,000	12,000	12,000	2,011,569	2,011,569	2,011,569	-1,000
Pan Am	2,495,000	621,000	549,000	549,000	1,416,151	1,416,151	1,416,151	-1,000
United	841,668	33,448	11,415	11,415	939,209	914,208	914,208	12,000
Western	100,581	451	1,343	1,343	106,029	104,118	104,118	-2,100
LOCAL SERVICES								
Airway	348,880	8,075	4,144	8,337	Neg.	360,223	249,492	-140,730
Alaska	365,400	3,389	1,480	8,320	160	366,760	363,199	-4,400
Colombia	200,273	2,148	1,144	1,144	Neg.	200,273	197,427	2,846
Trans. Center	227,412	212,412	3,311	22,855	9,414	232,872	262,417	-24,544
Latin Central	160,299	3,449	5,187	5,187	Neg.	163,554	163,554	0
South Central	160,299	3,449	5,187	5,187	Neg.	163,554	163,554	0
North Central	133,116	18,213	12,444	12,444	2,455	136,559	136,559	0
South	268,140	256,349	7,704	7,704	4,449	268,740	268,740	0
Pacific	1,000	1,000	1,000	1,000	Neg.	1,000	1,000	0
Hawaiian	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
Southern	170,102	7,243	8,054	2,496	227	197,771	125,834	-71,937
Trans Texas	243,350	9,261	8,161	2,496	32,356	247,449	247,449	-10,000
West Coast	149,791	8,063	1,100	2,702	3,074	166,740	166,740	0
CARGO AIRLINES								
American Fed. Airlines	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
City Transport	Neg.	Neg.	Neg.	Neg.	441,334	3,138,273	3,138,273	-347,936
Edsel	Neg.	Neg.	Neg.	Neg.	361,321	346,426	346,426	-15,002
Seaboard & Western*	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
Miss.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
REGIONS UNDER								
Chicago-Milwaukee	81,829	104,907	8,791	22,534	21,227	158,795	151,211	-26,086
Los Angeles Airways	12,442	9,026	1,137	1,137	1,137	105,823	96,020	-11,803
New York Airways	4,303	12,638	2,393	4,259	4,259	116,389	112,389	-4,000
U.S.A.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
Pacific Northwest	103,309	36,742	27,087	1,704	1,704	190,011	185,222	-4,789
ALARM 1958								
Airline available	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
† Contracted passenger figure.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.
Excludes to AFRICAN AIRLINES from airline reports to the Civil Aviation Board.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.	Neg.

SHOOTING FOR THE MOON



The engineer shown here has just completed simulated high altitude tests on a new shielded insulation that comes a step closer to meeting the needs of space travel.

Down-to-earth laboratory research like this may seem like a rather indirect way of shooting for the moon, but it happens every day at the Auto-Lite Wire Laboratory at Port Huron, Michigan. Here every engineer knows that any trip to the moon must start in the laboratory. The Auto-Lite engineer is always thinking about ways and means of creating insulations that will make possible still greater advancements in aviation and electronics.

Hundreds of different wire insulation materials are being tested here by the finest trained personnel using the most modern research techniques and equipment. Out of these tests comes a wealth of knowledge important to all customers served by Auto-Lite.

No matter what you need—whether it be high temperature aircraft wire or cable, magnet wire or wire for electrolytic and electrical applications—why not tell us your insulation problems and let us help you.

AUTO-LITE[®]
Wire and Cable DIVISION

THE ELECTRIC AUTO-LITE COMPANY
Port Huron, Michigan • Hazelton, Pennsylvania



THREE possible plans of orbital flight to a hypersonic glider are shown in artist's conception. Orbital test flight, which would probably be orbital for first research vehicles, could be controlled

in most events by glide attitude (maximum drag position, left, retarding balloon first); as electrical drag device (not shown) More positive control could be effected through atmospheric

Orbital Re-Entry Will Intensify Demands

By J. S. Bots, Jr.

Washington—Hypersonic speeds will bring changes in aircraft structures comparable to the advances from propeller design to all-metal construction.

This paragraph of a statement by Walter R. Dernberger, technical director at the president of Bell Aircraft Corp., illustrates the difficulties now encountered in designing the structure of a winged vehicle, which can orbit the earth and return safely.

No other plane of hypersonic vehicle design relies more completely on an trust technique and materials. As a result, the majority of hypersonic flight above Mach 5 generally has been limited in the program of structures and materials engineers in solving the high temperature problem.

Fabricator's Role

In the past, the role of particular usually has alternated between the manufacturer and the procurement designer, himself. Speed was controlled either by the drag and stability of the vehicle or by limited power. Rocket motors have essentially control the limits of power, and orbital velocity is now possible. Most enthusiasts agree that the fundamental uncertainties at these speeds can be resolved in thus an orbital vehicle. The development of hypersonic aircraft with which was displayed last month when major aircraft companies submitted proposals to an Air Force competition for an orbital bomber, the Dyna-Soar. A contract award is expected before May 1.



© 1962 AVIATION WEEK

AEROTHERMALIC requirements for a conventional landing are provided to control walls that can minimize heating at high speed. Adequate control and landing gear mechanisms returning part of weight, using boundary layer control or movable ground effect wings

AVIATION WEEK, April 21, 1962

although they are a serious weight penalty. The same design is the most efficient one of reentry during vehicle entry could be effected through atmospheric

heating period, which might last about five minutes. This would maximize the glider's effective lift drag ratio reduce its rate of descent into dense air and lower maximum skin temperatures.

On Structures

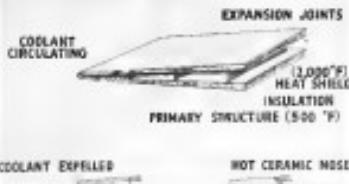
Two main structural problems faced in these programs are:

- To keep the aircraft having experienced damage to parts from reaching the aircraft's critical strength, warming it out of shape. Stressors induced by loads increase expansion reduce the strength enough available to carry an load. Warming air leads to thermal stresses in that there are thermal, results with a dangerous effect.
- To keep structural equilibrium intact so that available materials will return useful strength.

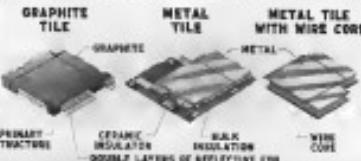
Each carries with each separate design proposal, but the problem approaches are not necessarily the same. One approach is to use a composite material, warming it out of shape. Stressors induced by loads increase expansion reduce the strength enough available to carry an load. Warming air leads to thermal stresses in that there are thermal, results with a dangerous effect.

Using these findings as a basis, the methods of reducing and accepting high temperatures at hypersonic speeds are

WING CONSTRUCTION FOR HYPERSONIC GLIDER



HIGH TEMP. INSULATED WALL STRUCTURES



Outer skin of hypersonic aircraft will probably be stripped into small sections covered by insulation pads. This would keep skin exposure due to insulation heating from causing severe heat damage. NASA suggestion above shows how expansion joints and several materials, insulation, reflective foil and cooling sections might be used to lower temperature and thermal stress of outer structure which would carry primary air loads.

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blade pitch banner at takeoff, would need a payload of at least 10% of total weight, which will probably be out of the question for many designs.

Although the ideal is not presently attainable, it is conceivable that fuel-specific impulse and wing efficiency will be improved to the point that the associated fuel consumption of hypersonic aircraft will overcome the heating problems by being twice as fast speed at high altitudes.

At present, the critical heating problem facing a vehicle exceeding Mach number less than 5 m/s. Skin temperature rises to 2,000°F and maximum temperatures around the nose and wing roots go up to three times higher.

These temperature values depend on a number of design factors and the main design effort is to achieve the combination of these factors which will give the lowest temperatures, heating rates and exposure cycles.

One of the improvements is that the aircraft can fly with a high lift coefficient so it will be efficient and stay at altitude longer. But to have the best heating, the nose and wing leading edge need to be rounded. This reduces the drag and reduces the lift/drag ratio so that the aircraft's speed and temperature are increased as it enters cruise at. This also increases the time that the maximum heating lasts.

Another component is that the vehicle needs a very light wing loading to be efficient and a lighter aircraft needs less fuel. This will be achieved by flying at a very high altitude. Hence, an addition to the wing area is achieved by increasing vertical area and skin weight too drag. Addition of cooling systems,

insulation and an afterburner of conserving the heat problem all add weight. This raises the wing loading and the aircraft will lose its loss more of its kinetic energy at lower altitudes. In fact, this will aggravate the heating problem in the cooling weight region, since going up again with another additional mass will increase the heating problems by being twice as fast speed at high altitudes.

Note for landing, the hypersonic glide is a constrained flight, also has effects on heat-shield performance and the heating problem. Wing efficients at hypersonic speeds depends primarily on two factors: span and the length of the wing measured from tail to tip, e.g., in the plan view of a swept wing, the distance between the leading edge of the root chord and the trailing edge of the tip chord measured along the centerline of the aircraft. An off-center wing with a long fore and aft dimension indicates a high sweep angle. Hypersonic skin temperatures about 10% of the way to the leading edge are efficient enough to keep the aircraft flying. However, at low speeds wave efficiency decreases a long way and a short chord. The arrow or slanted swept delta wings work as good at high speeds as poor for landing. Thus recenter jet controls, boundary layer control, or a variable geometry configuration might have to be used for safety at low speed.

It is possible that a straight or crooked aircraft weight savings might be effected and a lighter aircraft needs less fuel. This will be achieved by flying at a very high altitude. Hence, an addition to the wing area is achieved by increasing vertical area and skin weight too drag. Addition of cooling systems,

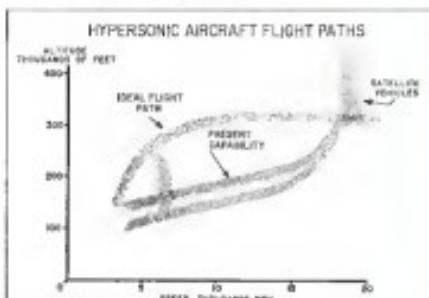


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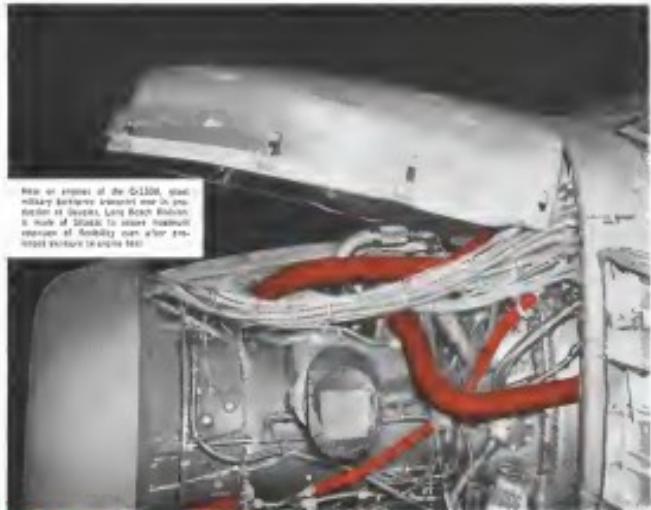
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WELDING-ELECTRODES - FUSION



FLIGHT Deviations of currently feasible hypersonic gliders are shown in the NASA schematic. The ideal flight path would only be attainable after great increases in available lift through better wing efficiency or fuel-specific impulse. Any upward shift of the curves of present capability will lower skin temperatures and relieve the thermal problem.



Interior view of the C-119, quiet military transports known now as "piggyback planes." Low temperatures caused so much static that severe limitation of flexibility over after prolonged exposure to engine heat.

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rushed ten miles, causing what is known as thermal shock.

A downward facing rocket launching from the fuselage would produce a few thousand pounds of thrust could improve the vehicle's effective lift/drag ratio from about 1.8 up to 1.9 or 1.1 and lessen the thermal shock. At the same time the aircraft's weight would be dropping rapidly as the propellants were exhausted, again reducing heat

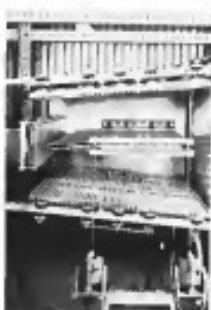
ing. It also is possible that through the use of power the aircraft could leave its angle of attack and yet maintain the same effective lift/drag ratio. This would lower the lift/drag ratio the vehicle was pulling, simultaneously lessening the effects of the heating.

If diagrams of accelerating engines continue to improve the altitude performance of these products as they have in the past, savings would be a most welcome source of power for that purpose. They would reduce the amount of fuel load or greatly increase the time that power would be available.

Designing for Heat

Most important variable in the heat problem is time. Heating rates and stagnation temperatures of many thousands of degrees can be witnessed in known situations for the few seconds of intense heat that might occur during a balloon vehicle's reentry.

However, given time it is completely conceivable to make a meaningful mark for these vehicles the heat in the wings and bodies that will



NASA aircraft flying at high altitude to simulate atmospheric heating of wings. Test hot air passes through sections specially to produce effect of flight at high Mach number.

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reflected skin temperatures to 200°F and leading edge temperatures above 400°F for several minutes. Even temperatures, heating rates and times of exposure would vary with vehicle design.

One of the most widely discussed methods of protection is to have a highly liquid metal shell which gives very little heat loss over front a heavy, relatively cool, mid-structure which carries most of the air load. Main problem with this is then insulation is to find a material that can carry high temperatures and then to insulate it in such a way that it can expand quickly and easily without buckling. This can be accomplished by separating the insulation sections which are joined together by sliding and expansion joints, going the structure a muffed or fiber insulation.

High temperatures on this will often be kept away from the mid-structure by radiation cooling and insulation, in some cases discussed in the literature. Several ways of reflecting hot radiators much of the heat to the skin. A layer of insulation around them greatly reduces the heat transfer rate to the mid-structure; so first it will not reach its maximum temperature until after insulation or lead time passed.

Radiation Cooling

Radiation cooling is one of the most efficient means of reducing the insulation. About 40% of the heat transferred is due to conduction at the boundary layer can be reduced to the atmosphere by this method. Designers are attempting to improve the reflectance qualities of the surface of insulation aircraft.

Other cooling methods which have been suggested include pumping coolant through a double skin, convection cooling using a porous or liquid metal skin so that coolant will seep out and cool the outer boundary layer, and fans to move air over the leading edge so that the skin that is covered with a thin film of liquid.

Cooling the aircraft with a solid material which will stick a glassy state at relatively low temperature also is considered promising. These aluminum materials are usually organic and become glassy around 500°F. At the normal skin temperatures the aircraft are underneath at this leading edge of skin.

Most aircraft positions are to have the insulation surround exposed at the same rate as the skin as it is heated to 400°F with no cold air. Material also has to be relatively flat so thermal stresses to keep the surface from cracking and spalling or peeling off. The negative effect hole in the abutting surface usually leads to its complete degradation.

Studies are being made to determine

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the feasibility of crashing an aircraft by spraying it with electrons which will repel a portion of the mass in the air above it about Mach 10. Another method would obtain useful thrust from the liquid used to cool a high speed structure. Coolant liquid would have a lower vaporization temperature than the equilibrium temperature of the outer skin. As the liquid was passed through ducts underneath the outer skin, it would vaporize and be expelled through nozzles in the aft portion of the fuselage and wings to provide thrust.

Heat Sink Method

The heat sink method of "handling" heat consists mainly of insulating the structure to take heating loads due to uneven heating and expansion in addition to the usual air loads. However, the sheetmetal material has less strength because of its high temperature than it would if it were cooled. Thus, one advantage when this construction method is used is a weight advantage over structures with elaborate cooling systems.

Most structural experts believe that practical aircraft will use combination of

these construction and cooling methods. A sixth serious problem which requires an answer no matter what type of construction is used is the proper design of high temperature joints. To a single base under tension, heating the parts help to increase the heat load and one end of the joint is at a higher temperature than the other. Heat through structural joints, however, need not even provide overheating because of varying stress and attachment patterns and the conductance of rivets, bolts and welds. One creates large thermal stresses which are being eliminated by using strips of good heat conductor lead between the overlapping sheets of metal to spread the heat wave evenly. Another method uses two or more types of materials in joints so that differences in their thermal expansion will overcome for their difference in temperature.

Concerning windmills the following problems concerning the stresses in basic joints and adhesives of structural stiffeners and connecting members are further cause by heat, unequal and distortion of loaded structures will occur at high temperatures, and the pro-

tection of fatigue life. The safety factor which the lack of knowledge demands in this structural problem has all authorities outside the fast hypersonic configuration because that ones which follow.

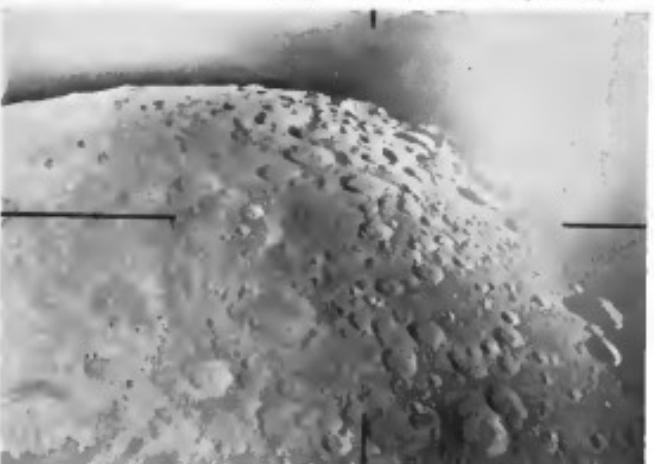
At present that is a need for test facilities which can subject large segments of aircraft aircraft to wind tunnel heating and simulate, preferably in dynamic form.

Some large facilities of this type are in various stages of construction at government laboratories.

Hot Henry

Even though the first hypersonic aircraft from which gliders are now being tested which before which later prove to be inaccurate, the vehicles probably will not be heavily constructed in comparison to current supersonic aircraft.

The gliders will fly so high in hypersonic winds that their insulation becomes present and an insulator may never be greater than three expansions at sea level at 400 mph. They also will fly in essentially straight lines with minimum sharp turns related to long air loads low.



Minor surface, photographed with Recording Optical Tracking Instrument Mark II, is magnified to appear as it would from a distance of 150 mi. ROTT II, developed by PerkinElmer Corp. to track Atlas Intermediate ballistic missile flights, is installed at Patrick AFB (AWW Oct 24, p 111).

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FROM END TO END...INSIDE AND OUT

AVIATION WEEK April 21, 1958

Reorganization Plan Responsibility Not Clarified for Defense Personnel

New York—Proposed reorganization of the Defense Department has caused some confusion among Defense personnel as to whom group responsibility for specific military and space programs will fall. Colonel M. Belakoff said at a recent Space and Astronautic Engineers National Association Meeting:

Actually, and the Defense Department doesn't do a good enough job here. The Advanced Research Projects Agency has been assigned specific projects and further assignments to that group are currently under consideration. The question is who will get responsibility for the development of a nuclear-powered aircraft engine. It may be assigned to AFSC [Air Force Systems Command] or AFM [Air Force Materiel Command].

At a recent meeting, following his budget talk, Belakoff said the program would be put under AFSC. This is because AFSC has the best facilities for propellants with higher specific impulse and smaller storage problems.

Both AFSC and AFM believe they can help meet this need. One of the older fluids, not having sonic and, it is said so far, no liquid hydrogen, is not used by Belakoff, an experienced propellant man, as an example of a trouble liquid, provided some material can be found to store it.

Some believe less than others. For one thing, it looked as though the role was going to be study before the role needed to live there were open ended.

It was not a case of getting back ground or false belief in decisions made because of lack of money. "You cannot buy thinking with money," he declared.

There is no plan to phase the Task out, he said. It will disband, go into production. At this point, however, it will be the early research and development phase. So far, nothing of the original has even been started.

Referring to the subject of nuclear propellants, Belakoff said that the current liquid combinations, particularly boron and oxygen, are not about to change. There is no reason to believe the propellants with higher specific impulse and smaller storage problems.

Both AFSC and AFM believe they can help meet this need. One of the older fluids, not having sonic and, it is said so far, no liquid hydrogen, is not used by Belakoff, an experienced propellant man, as an example of a trouble liquid, provided some material can be found to store it.



Aircraft Skin Alloy

Ron Amerson was last year at National Design Engineering Show in Chicago with Max-Way, an exotic skin alloy for high speed aircraft.

Aided when possible by being made in the substitution of liquid hydrogen, Belakoff replied that little had been accomplished concerning this. Releasing liquid hydrogen as a high energy fuel had declined to six weeks; it now stood as four to eight days development time. (Liquid hydrogen is called SF-1.)

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On the matter of the workload at Cape Canaveral, Holloman said that there was still room for more testing facilities, but was less certain that other facilities would be needed in the vicinity of Cape Canaveral development continues to grow.

Equipment and talent, facilities are being put into Camp Cooke in order to build that up as a Pacific test site. (AW March 27, p. 21)

While Cape Canaveral, the West Coast stage will not be a research and development facility to an all extent. One reason is the high cost of clearing a large area of the Pacific for range safety prior to a test.

In reply to a question as to the number of parts required to develop a new ICBM weapon, he reported to be over 2500 different items. Holloman pointed out that models could possibly look and have effects like SolarFlare will be until some basic one is developed that models can be tested. Some people in the program he said believe that a test will get out only one outstanding errors, another but 10-test of 10 or, in other words, one right after the other.

Work on new anti-submarine wing out is progressing. There is a great need for new devices, Holloman said. With regard to details, he added that the next major requirement is to penetrate a depth and an attack more effective than the old torpedoes.

Solids vs. Liquids

Between 1958 and 1970, liquid propellant model engines may be used for specialized missile applications requiring high performance at extremely high temperatures, solid-propellant abilities to handle propellant systems and for space flight systems.

The solid propellant combustion studied by George S. Sartorius of Boeing Airplane Co. also made a point-by-point comparison of solid and liquid rocket propellants.

Comparison was based on large dual-axis vernier thrust rocket engines designed as having more than three tons of thrust (excluding hardware and payload limit) and was concerned more with static adhesion and combustion than with performance.

Among the criteria weighed in his comparison listing were the following: a 5-second ignition time; Solids are much 262 lbs. heavier but are not likely to reach the 313 thrust levels with liquid because hydrocarbon storage tanks require higher and loading difficulties, however, the liquid-fuel system will be used only for special high performance applications.

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• **Thrust transmission—Closing propellent valves on hybrids is simple and effective. Shifting off exhaust nozzle in solids will not work with some propellents and may result in large longitudinal accelerations. Opening of a nozzle for long exhaust will cause a violent forward thrust, but it is still too early to see how well it works. A third option depends upon the duration of some extinguisher that might be required after the motor or solid chamber.**

• **Thrust level control—In liquids, throttling has proven flexible and effective and can be carried out on command. Gasoline design in solids is effective for pre-selected burning levels but cannot be varied after the engine is lit.**

• **Combustion instability—This has been one of a problem for liquids but is decreasing with the accumulation of experience.**

• **Thrust duration—Regeneration, adding pressurized air to extend duration is long in time is probably the simplest, that solids will ever be able to match this.**

• **Turboprop availability—Running rates or vehicle increase in which to 10% per day, so there is little regeneration. Solid fuels under development which will not have this problem.**

• **Simplicity and reliability—Liquid rocket engines with their much larger numbers of working parts, many of which are unique in nature, have power availability and regenerators were implemented for flight inspection, fueling and checklist equipment. The one shutdown problem in checking out solid engines is finding the need for an off cycle check during flight.**

• **Cost—Costly about five hours in much to develop a hybrid propellant rocket engine as it does to develop a comparable solid rocket, but no cause to consider it.**

Solid propellant rocket engines will continue to go into the small market and into auxiliary applications such as vernier engines and retro-rockets. In addition, they will find a market as a main powerplant in interorbital craft and interplanetary ballistic missiles.

For a terrestrial vehicle like the XCOR, the hybrid will, the advantages of regeneratively cooled engines make the solid propellant rocket engine a logical choice. At the same time, the applications demanding higher thrust

AERONAUTICS, April 21, 1968

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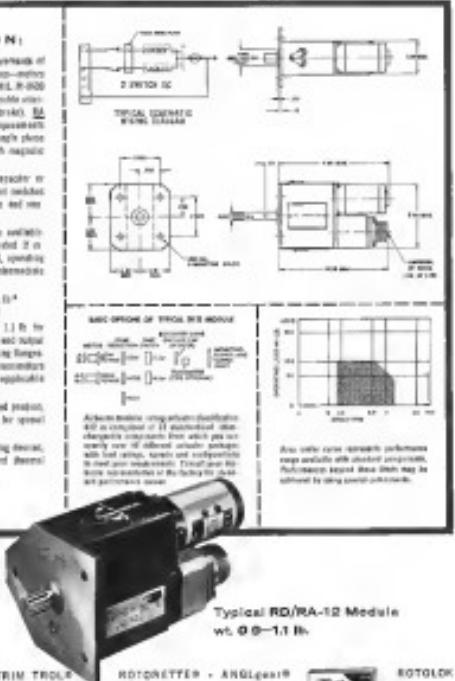
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- Operating strokes from 2° to 180° are available. An additional 30° of overtravel is provided. If extremely precise position is used, operating strokes up to 30° after side of intermediate position can be achieved.
- Maximum operating load—up to 100 in.-lb. (Ultimate static load—up to 200 in.-lb.)
- Weight—approx. 8.5 lb. for RD Series. For RA Series—includes mounting flange and output shaft. Weight for speed increasing ranges varies, depending on intermediate gear ratios and/or flywheel weight. Weight per module will be furnished on request of applicable specifications.
- Electrical connector is shown in specified position, but may be located on other surfaces by special application.
- Actuators are available with lead-freeing devices, anti-vibration mounting techniques, and thermal insulation protection.
- Actuators are designed to meet fire safety standards. Weight per unit per specification.



Typical RD/RA-12 Module
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selections, the extra performance potential of the liquid rocket becomes extremely important.

Among other things, it permits a reduction in the number of stages required as well as in the overall bulk of the vehicle.

There is little question that sustained thrust generation will be a requisite to extended space flight.

The question, after, is what kind of propellant can be developed to meet this requirement.

One possibility is the ion propulsive rocket engine. During 1957, the Directorate of Advanced Studies, Air Force Office of Scientific Research sponsored a theoretical investigation of the ion engine at Rocketdyne Division of North American Aviation, Inc. The results of this program according to Rocketdyne's Robert H. Beder indicate that an ion rocket engine in which the propelling ions are accelerated by an electrostatic field is capable of ion-beaming at low thrust levels and will be suitable to implement various other types of nuclear engines in applications to space vehicles.

There are a number of technical problems yet to be solved, of course, before the ion ion rocket engine becomes a working reality. The most formidable of these have to do with the development of the ion generator propellants and a high specific power generator.

The Directorate of Advanced Studies plans to continue sponsorship of the ion propulsion research program during 1958.

And sometime during the current year, according to Beder, Rocketdyne will begin a limited experimental program in hopes of resolving some of the data already gathered.

The prospects of ionizing boronite fuel tablets, for instance, also appear dim.

When all things such as reactor and storage tank designs are taken into account and V. C. Lee and S. F. DeGraw of Aerjet General Corp., the proponents mentioned by Beder, point out their successful use in propellants at that time.

As an example of what they mean, Lee and DeGraw took the core of a fast solid propellant that starts out with a specific impulse of 1.20. By the time the fast radicals are incorporated in a practical solid propellant via heat, the final specific impulse will have dropped to 0.60.

They said so much as 50% of the available energy can go into the initiation and generation of reaction products resulting in useful heating due to the high activation temperatures (10,000K) associated with free radicals.



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AIR TRANSPORT



FACTS AND FIGURES

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OFFICIAL PUBLICATION of the



The Standard Reference of United States Scheduled Air Transportation

WHAT THE INDUSTRY DID IN 1957

The scheduled airlines set new traffic highs in 1957. They carried more people, and more property, more places and for longer distances than ever before.

1957 was a year in which the airlines flew:

- + 49,339,000 passengers;
- + 31,513,100,000 revenue passenger miles;
- + 103,685,000 ton miles of U. S. mail; and
- + 97,656,000 ton miles of freight.

Operating revenues were also at all-time highs. But operating expenses were up at an even higher rate, cutting sharply into the overall net profits of the industry. When the year was over, the airlines found that they had:

- + Totals in \$2,114,386,000 in operating revenues;
- + Spent \$1,651,339,000 in operating expenses;
- + Kept \$42,477,000 as net profit after taxes and interest.

Air Transportation—Still a Bargain Buy

In spite of the relentless upward trend of prices over the last twenty years, the airlines have held the fares to 1958 levels. In that year, for example, the average revenue per passenger mile for the domestic airlines was 1.82 cents. For twenty years it has been about that figure until today, including the recently authorized increase of 4 per cent with a \$1.00 service charge per ticket, the average passenger revenue is only 5.42 cents based on carrier estimates for 1958, a gain of 1.8 per cent over 1957.

While air fares have gone up only slightly, service has improved markedly. Flying time between many cities has been cut in half, there is a greater selection of flights, cabin and meal service has improved, and planes are more comfortable, more dependable, and safer.

Decreasing Dependence on Government Aid

Over the last two decades, the scheduled airlines have continually progressed toward self-sufficiency.



In 1958, the airlines received 42.4 per cent of their revenue from the government in the form of mail pay, which included both subsidy and payment for carrying the mail.

Today, the payments have been separated into service mail pay and public service revenue. Although the airlines have earned increasingly larger amounts of mail, by 1957 their dependency on this revenue had been reduced

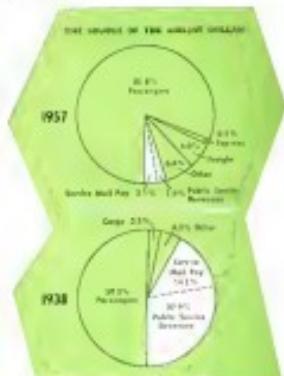
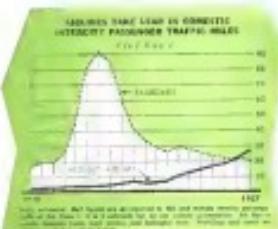
to the point where only 3.1 per cent of airline revenues came from mail service.

Safety also continues to play a leading role in airline income. In 1957, only 1.8 per cent of the airlines' total revenue came from the government through these public service subsidies.

The loss of the domestic truck lines went off subsidy during 1957. Today, the local service airlines receive the majority of public service revenues, a guarantee that smaller communities will receive the benefits of air transport service.

The other public service revenues are used for the development of helicopter service and for the development and maintenance of national interisland routes in Alaska, Hawaii and Latin America.

For the year, the airlines flew more than 25 billion passenger miles while the railroads operated approximately 21 billion. The buses, during the period, operated about 16 billion passenger miles.



Airlines Now No. 1 Common Carrier

During 1957, the air transport industry took over first place among the common carriers competing for intercity passenger traffic.

Since the mid-1930's when the railroads first assumed the leadership with the large lines and stage coaches, the Iron Horse had been the leader.

Airlines Safety Record Outstanding

In 1957, the combined U. S. scheduled domestic and international airlines achieved a safety record almost unsurpassed in airline history.

Today, R. is more than four times as safe to travel by domestic scheduled airline than by automobile. During 1956, the latest period available, there were 26,100 auto-and-truck passenger fatalities, a rate of 2.7 fatalities per 100 million passenger miles.

The airline rate was two-thirds of a passenger fatality per 100 million passenger miles in 1957, as compared with five-tenths of a fatality per 100 million passenger miles in 1956. This record was made during a year when scheduled carriers flew an all-time high of more than 31 billion passenger miles.

Cargo Business At New Peaks

Last year, the air cargo business leaped to new highs with the operation of more than 716,432,000 ton miles of U. S. mail, express and freight, an 8.6 per cent gain over 1956.

The scheduled airlines also achieved new levels of revenues from their cargo business

DOMESTIC TRUNKLINES

Public service revenues—commonly referred to as subsidy—dropped to zero for the domestic trunklines operations by the end of the year.

At the same time, it was a year when trunksline commitments to purchase jet jet and prop-jet aircraft and related ground equipment reached a figure of about \$8 billion, representing a capital investment of approximately three times the value of present flight property.

As the domestic trunklines geared for the jet age during 1967, traffic continued to rise in all categories except one.

Revenue passenger miles showed an increase of 13.4 per cent, climbing from \$1,633,141,000 in 1966 to \$1,899,610,000 in 1967. This increase is a slight improvement over the previous year's gain of 12.6 per cent. The trunklines accounted for 78 per cent of the scheduled airline industry's total passenger mileage in 1967.

Mail volume rose 6.4 per cent to \$1,194,400 ton-miles. Air express showed a drop of 14.8 per cent to 43,753,000 ton-miles—attributable primarily to the Railway Express Agency strike.

Domestically, air freight rates have become in many cases, comparable with, and in some cases, lower than, the fastest surface transportation rate.

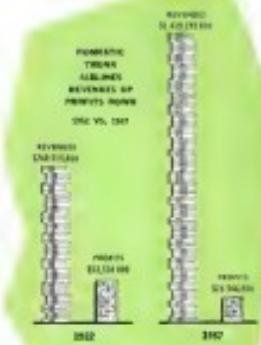
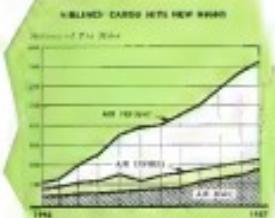
No Show Problem Reduced

With the co-operation of the public, the domestic airlines' program to reduce the number of no-shows made notable gains during the year.

The airlines' "no-show" program was completed last September when a \$5 penalty was initiated. Previously, the industry had instituted a ticket pick-up time-limit provision and a reconfirmation rule.

Surveys conducted during two one-week periods showed that only about one half of one per cent of the industry's passengers were assessed the no show penalty.

The survey also indicated that the number of no shows has been reduced from 16 per cent of passengers boarded to about six per cent.



Meanwhile, air freight volume totaled 218,132,000 ton miles for the year, representing a sharp increase of 14.6 per cent compared to the 9.5 per cent gain recorded in the previous year.

Although the trunklines handled a record volume of passenger and cargo business in 1967, earnings were seriously reduced by a mounting expense level that jumped 10.6 per cent over 1966. Net profit for the industry tumbled from \$83.7 million for 1966 to \$26.5 million in 1967—a drop of 64 per cent.

INTERNATIONAL AIRLINES

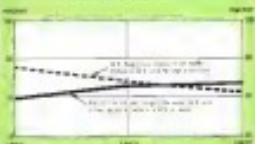
There is a critical need for improving our ties with peoples from all parts of the world. In this age of advanced technology the world has become smaller and countries find themselves next door to one another. It thus becomes essential that the people of the world come to understand one another in order to maintain or develop the status of friendly neighbors.

The best way to understand one another is to know one another. Toward this end, U. S. international airlines have made much progress over the years in helping to remove international travel barriers. Customs, immigration and public health facilities have been streamlined in the interest of quick entry into the United States of returning citizens and foreign visitors. The United States now competes favorably with respect to border-crossing facilities with most of the nations of the world.

These efforts are reflected in the increasing volume of international airline traffic. During 1967, U. S. international carriers flew 5,254,663,000 revenue passenger miles, as compared with 5,112,312,000 during the previous year. Cargo ton miles reached a new high of 321,386,000—an increase of 15.9 per cent over 1966, while mail ton miles jumped from 65,154,000 in 1966 to 57,260,000 ton miles in 1967.

The year was marked, however, by the unanticipated grant of valuable air routes to foreign competitors—KLM and QANTAS—without exacting in return rights to U. S.-flag airlines of comparable economic value. These grants were contrary to the spirit of the United States' bi-partisan air transport policy.

FOREGOING AIRLINES TAKING HIGHER SHARE OF U.S. MARKET



LOCAL SERVICE AIRLINES

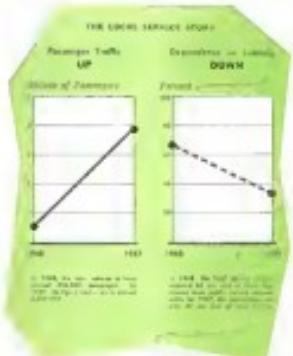
The local service airline industry rounded out its twentieth year of operations in 1967 by carrying 3,949,000 revenue passengers, for a 19-year total of 38,365,000.

Created directly after World War II to link the intermediate cities of the nation with one another and with metropolitan centers, the local airline industry earned only 35,000 passengers in 1946, first full year of service.

Pursuing a program of constantly increasing public usefulness, the 13 local service carriers today operate a fleet of 225 transports over a system that reaches 21,425 unpopulated route miles and reaches into all but four of the 48 states.

It is particularly significant that more than one half of all the cities served by the local airlines would have no direct scheduled airline passenger, mail or cargo service if it were not for the local carriers.

In the past 30 years the local airline industry has increased its passenger load factor from 27 per cent to 46 per cent and has increased passenger revenues ten-fold and total operating revenues five-fold. Accordingly, the percentage of public air service revenues has declined—from 66 per cent of the industry's total income down to 56 per cent. Since 1952 the industry has increased its traffic over 300 per cent.



Gulf-skies available, 126 per cent more than in 1956.

Air freight tonnage remained in 1957—a gain of 100 per cent to 11,000 ton miles. An express tonnage increased at a faster rate than during the preceding year—33,800 ton miles, or 155% for an increase of 16 per cent. The helicopter airlines flew 81,000 ton miles of mail, a gain of two per cent.

The three helicopter lines altogether operate 24 aircraft over 250 route miles serving 24 certified points.

THE ALASKAN CARRIERS

In 1957 permanent notification for operations operating between points in the United States and Alaska was authorized, permitting these airlines the same permanent operating rights that were authorized in 1956 for carriers operating within Alaska.

During 1957 the Alaskan carriers improved their scheduled traffic figures in the movement of passengers and mail, but showed a drop in cargo.

Revenue passenger miles went up 109 per cent to a 1957 figure of 151,884,000. U. S. mail ton miles went up from 2,386,000 to 2,702,000, a gain of 15.4 per cent. Cargo ton miles dropped 8.8 per cent with a loss of 7,556,000 compared to 1956 when 10,506,000 ton miles were flown.

Revenue ton miles showed a drop of 26.6 per cent with a 1957 total of 32,500,000 compared to 44,800,000 the year before. The tail-off was due largely to a reduction in the number of military charter, more than offsetting the gains in scheduled service.

TERRITORIAL CARRIERS

One-hundred and forty-five per cent more passengers flew scheduled helicopter in 1957 than in 1956, testifying to the growing popularity and usefulness of scheduled helicopter service.

Serving greater Los Angeles, Chicago and New York, the helicopter airways carried 135,000 revenue passengers in 1957, compared to only 62,400 in 1956. Revenue passenger-miles totaled 3,233,000 in 1957, an increase of 106 per cent.

Overall, the revenue ton miles of the territorial carriers went up 5.8 per cent, from 8,363,000 to 8,648,000.

THE ALL-CARGO LINES

The all-cargo airlines registered impressive gains in all categories last year. Their overall ton-mile haul was up 33.4 per cent over 1956, rising from 248,781,000 to 326,962,000.

Most spectacular increase was in air express which increased 39.4 per cent—from 1,296,000 ton miles in 1956 to 1,638,000 in 1957.

U. S. mail ton miles increased from 1,069,000 to 1,849,000—up 73.8 per cent while freight ton miles went up 35.8 per cent—from 148,420,000 to 185,120,000.

IMPROVING SERVICE TO POSTAL USERS

The year 1958 will see three anniversaries marked—all significant milestones in the history of air mail service and the story of its growth and usefulness to the postal service and the public.

They are: The 90th Anniversary of Air Mail, the 20th Anniversary of the Civil Aeronautics Act, and the 10th Anniversary of Air Parcel Post.

Air mail came into being on May 15, 1918 when World War I Army pilots flew mail between Washington, Philadelphia and New York. During that first year of mail operations about 94 tons of mail were carried to produce \$100,700 in "aeromail" postage revenue. Today, air mail produces some \$89 million annually in revenue for the Post Office Department.

On June 22, 1938 the Civil Aeronautics Act was passed which placed an obligation on the scheduled airlines to serve the postal system of the United States. Since then, public use of air mail has increased more than a thousand fold.

At least two other additions to mail movements by air have added to the usefulness and versatility of air mail. On September 1, 1948, parcel post gained wings and on September of 1953 the Post Office began what is known as the "Three-Cent Air Mail Experiment," under which domestic airlines carry first-class letters daily between certain cities on a space-available basis.

When the airlines first began to carry the mail and fulfill their obligations under the Civil

Aeronautics Act, public service revenues were an important source of income for the young industry. In 1938, 44.5 per cent of airmail revenues went to the airlines while the Post Office kept 55.5 per cent. Today, airmail receive only 63.5 per cent of the remainder goes to the Post Office. "We save money every time we use air," the Postmaster General said recently.

Increasing usefulness of air mail is reflected in air transport industry figures for 1957. A total of 143,744,000 ton miles of letters, cards and parcel post were flown—an increase of 4.9 per cent over 1956. Paid post—lusty young



members of the air mail family, is growing steadily. Air parcel post shipments increased by 8.6 per cent over a six year period.

Three-cent mail carried by air increased, too. During 1957 this class total totaled 16,836,000 ton miles, up over 23.6 per cent. This service benefits the public by making mail deliveries possible 12 to 48 hours sooner than if it had raced by surface, and a Post Office spokesman has said that since of the experiment showed "that costs were not greater, in fact, they were running a bit less than when we were using surface transportation."

NATIONAL DEFENSE

While the airlines are dedicated to the public service, to the nation's commerce and postal service, they are dedicated also to the national defense.

As part of defense planning, the airlines—in cooperation with the Department of Defense and other government agencies—have established the Civil Reserve Air Fleet (CRAF) and

In 1957 Congress passed two bills designed to encourage airlines to re-enter the market. One guarantees loans that the local airlines and certain other carriers may receive for purchase of improved flight equipment and the other makes certain equipment trust provisions applicable to aircraft and aircraft equipment. Also, a bill was proposed which would permit local air carriers to apply the capital gains from the sale of obsolete aircraft to the purchase of many more modern planes instead of having the proceeds disbursed from their public service revenues.

HELICOPTER CARRIERS

One-hundred and forty-five per cent more passengers flew scheduled helicopter in 1957 than in 1956, testifying to the growing popularity and usefulness of scheduled helicopter service.

Serving greater Los Angeles, Chicago and New York, the helicopter airways carried 135,000 revenue passengers in 1957, compared to only 62,400 in 1956. Revenue passenger-miles totaled 3,233,000 in 1957, an increase of 106 per cent.

Overall, the revenue ton miles of the territorial carriers went up 5.8 per cent, from 8,363,000 to 8,648,000.

the War Air Service Pioneers (WASP) for maximum domestic and global military airlift in time of national emergency.

The CRAF—composed of 368 long-range four-engine airliners, including 244 from the scheduled airline fleets—would operate on a



global basis, carrying troops, supplies and equipment in support of the military effort. All four-engine fleet, and twin-engine aircraft in the civil fleets would make up the WARP in provide an insurance system of priority airlift in support of defense protection at home.

The machinery has been set up whereby the services can bring all or part of their own aviation into military service immediately should an emergency arise and be fully operational within 48 hours.

The advent of the jet age promises even greater benefits to the national security. Most of the turbine-powered aircraft on order are the types most desirable for wartime military support operations. When delivered, the civil capability under wartime conditions will be more than 8 billion ton miles annually, compared to the 2.8 billion of the present piston CRAF. The airlift capacity will be nearly 15 times more than the average annual airlift provided by the airlines during World War II in military and civilian service.

During 1957, the Air Forces emphasized that it will depend upon the scheduled airlines for normal jet air transport lift in time of emergency. Secretary of the Air Force James H. Douglas said: "We have not ordered any low-budget aircraft for the Military Air

Transport Service such as the airlines have on order in large numbers and which we could not afford in time of national emergency."

In an unusual move, the Defense Department intervened in the General Passenger Fare Investigation hearings last year. The Department stated that "it recognizes as a matter of the greatest urgency the necessity for maintaining a strong, modern and economically sound air carrier industry to meet the requirements of national defense during peace time and national emergencies."

The airline defense role represents substantial saving to the American taxpayers. It would cost the government an estimated \$550 million initially just to acquire the present piston CRAF, plus \$200 million annually to maintain the resources. Since most of the aircraft on order will be suitable for military airlift, most of the \$2.5 billion investment being made by the airlines constitutes further substantial saving to the taxpayers.

Airlines as Military Partners

This unique capability of the airlines stands out as one of America's finest examples of military-industry partnership—a close, working relationship that serves the nation in peacetime as well as in time of emergency.

Every hour of every day the scheduled airlines serve the Department of Defense through the Military Bureau of the Air Transport Association and its field offices. To better serve the various branches of the armed forces, the scheduled airlines have established Joint Airline Military Traffic Offices (JAMTO's) at 68 military installations throughout the country. Under the jurisdiction of local military commanders of the industry, these JAMTO's assist in making arrangements for movement of both cargo and personnel.

It is estimated that some 22 million man hours were saved during fiscal 1957 by the use of scheduled air transportation.

AIR NAVIGATION AND TRAFFIC CONTROL

One of the pressing needs for the orderly growth of civil aviation is a modern navigation and traffic control system. Such a system does

not now exist but the government is encouraging its development.

A positive step toward this end was taken in May when the Special Assistant to the President for Aviation Facilities Planning, General Edward P. Curtis, made his report.

The report called for the setting up of the Airways Modernization Board, an independent agency "responsible for developing and considering the requirements for future systems which are needed to provide the necessary communications, navigational aids, and control needed to accommodate the future air traffic in the United States. It would be responsible as well for the systems engineering, the evaluation, and the selection of such aids as will best serve the needs of aviation."

It is clear that a revised system will require considerable expenditure. The scheduled airline industry believes that the principle of payments for the use of such federally furnished facilities within the U. S. is a sound one. The airlines have been paying a fair share of the cost and they expect to do so in the future.

Increase in Fuel Tax Proposed

Any allocation of costs that is made, however, must take into consideration the degree to which the airways are used by the three components of aviation: military, air transport, and private and business flying. At present, the air transport industry was operating 1,829 planes, the military 40,000 and private and business interests, 60,000 planes.

Before assuming a higher fuel tax, as the government recently proposed, there should be a thorough study which will determine the actual use of the airways. According to the Administrator of the Civil Aeronautics Administration, the military makes use of the airways system 45 per cent of the time. Since the military also has an overriding and understanding, priority call on the exclusive use of the airways, this fact should also be taken into account.

The vanishing air space remained a critical problem at year's end. As the Curtis report pointed out, "The American airspace is a natural resource that is on the verge of exhaustion in terms of the capacity of the system over a

place for managing our free and safe access to this resource."

A start on the problem was made when the Civil Aeronautics Board delegated to the Civil Aeronautics Administrator final authority in the designation of restricted areas.

Prior to this action, the air spaces had been controlled by the Administrator with the assistance of the Air Space Panel of the Air Coordinating Committee. Because unanimous consent of the panel members was needed however, some of the companies that resulted were in conflict with the public interest, the Board noted.

The airlines strongly support the new procedure and are in accord with the Civil Aeronautics Board's position that it is a "major forward step in determining how air space will be used," and that it "will resolve the conflicts that arise between the various users of air space."

Airport Rentals and Leasing Fees

For the last ten years, the airlines have been paying steadily increasing landing fees to airports. At one large city, they have gone up in that period from 4 cents to 16 cents per 1,000 pounds; at a medium size city, from 3.5 cents to 8 cents; and at a smaller city, from 5.5 cents to 8 cents.

In 1957, the domestic trunk and local service airlines paid out an estimated \$14 million in landing charges.

To this figure must be added an estimated \$10 million representing rental payments made by the airlines during 1957. This total of \$24 million paid to airports is just about equal to



the net profit of the domestic industry for the year.

JOB PROGRAM THREATENED BY FINANCIAL SQUEEZE

At the end of the year, the scheduled airlines had an order 674 new aircraft for delivery between 1958 and 1963.

Making up the airline order were 250 pure jet planes, 167 turbo-prop, 28 piston aircraft and seven helicopters.

The equipment orders represent an investment of more than \$2.3 billion. This investment became all the more important when the end of 1987 saw the beginning of a recession that forced many industries to cut back on capital expenditures.

The airlines' re-equipment program will go a long way toward the priming of the pump of American industry, providing new jobs, creating new ships, and providing the revitalizing energy throughout the whole economy.

Airline estimates indicate that 39 per cent of all traffic will be flying in turbine-powered aircraft by the end of 1959 and that the new jets will be the basic airline aircraft by 1965. The Civil Aeronautics Administration predicts that by 1965, domestic volume will reach 88 million passengers, almost twice the amount carried in 1962.

With the introduction of the jets, the public will be able to fly faster and more comfortably than ever before. The world will be shrunk to a little more than half its present size and time

Shippers will see their goods being moved across the country and even oceans in half the time. The enormous lift capacity of the new jets will open up new dimensions for cargo movement that have only been hinted at so far. This move will move faster, more often, and in greater quantity.

At the same time that the airlines were placing this bet on the future of their industry, they were beset by a financial squeeze that grew increasingly severe.

The profit margin for domestic trunk lines was 1.8 per cent in 1967, compared to 7 per

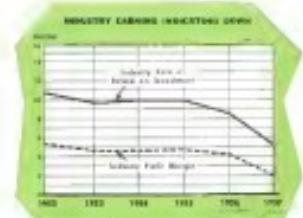
cent in 1982. The operating profit margin was 5.4 cents per available ton-mile for 1983 but it had dipped to 3 cents by 1987. Operating margin per revenue ton-mile was 8.9 cents for 1981, compared to 1.5 cents for last year.

Since 1968, the average revenue per passenger mile has gone down from 5.32 cents to 4.28. During that same time, the overall cost of living went up 28 per cent, bus tickets went up 37 per cent and rail freight rates went up 44 per cent.

Despite the fact that virtually every trunk airline filed requests for fare increases ranging from 12 to 17 per cent, the Civil Aeronautics Board granted, in early 1958, only a 4 per cent increase with a \$1 service charge per ticket.

The insufficiency of the increase can be seen in the comparison of profit margins over recent years. Last year, as noted, the profit margin was 1.9. With the recent assumption, it is estimated that the profit margin for 1968 will be only 2.6, a long way down from the 4.6 per cent of 1954, or the 7 per cent of 1952. The estimated gross revenue for 1968 will more than double those of 1952, but the profit margins will

The earnings decline has affected not only the domestic trunk lines, but the entire industry as well. The rate of return on investment for the scheduled airlines dipped from 10.8 per cent in 1982 to a low of 5.1 for 1987. Industry profit margins fell off similarly, going from 5.8 per cent in 1982 to a dangerously low level of 2.0 per cent in 1983.



AVAILABLE SERVICES AND UTILIZATION

Fig. 3. Rehospitalized Airlines Industry

(For Selected Years, in Millions)

THIS TABLE SHOWS THE EVER INCREASING GROWTH IN THE SERVICES THE SCHEDULED AIRLINES ARE OFFERING TO THE PUBLIC AND THE INCREASING USE OF THIS SERVICE BY THE PEOPLE, THE GOVERNMENT AND SHIPPERS.

Airline's Ten Miles From	Revenue Per Miles Flown	Total Load Factor (%)	Airline's Ten Miles From	Revenue Per Miles Flown	Pasenger Per Miles Factor (%)	Reven- ture Per Miles Flown
Domestic Trunk Airlines						
1918	N.A.	32.8	—	301.4	407.3	10.21
1919	112.3	43.1	46.16	7,496.2	8,930.1	78.20
1920	1,162.7	206.7	51.6	1,996.2	3,424.9	56.34
1921	1,694.1	191.9	54.12	20,394.6	23,138.6	70.20
1922	2,191.2	197.6	58.26	19,861.1	21,129.9	67.04
1923	2,479.4	142.9	51.16	22,118.8	34,250.4	46.49
1924	3,104.1	133.6	55.19	25,444.6	34,386.3	43.31
1925	3,882.7	2,160.1	53.63	20,112.3	26,217.2	48.64
1926	4,169.2	144.0	58.02	30,353.6	21,643.1	63.13
1927	5,355.4	2,728.0	52.81	37,938.3	24,499.8	41.83
Local Service Airlines						
1918	—	—	—	—	—	—
1919	—	—	41.21	9.0	8.8	27.12
1920	5.8	9.3	27.65	323.3	47.9	31.34
1921	62.6	35.6	32.32	195.2	188.8	20.11
1922	94.2	34.0	37.37	106.6	159.3	37.44
1923	119.2	41.1	37.31	1,373.6	219.7	36.34
1924	112.8	47.6	43.24	1,897.5	400.8	43.03
1925	101.9	59.3	42.26	1,184.6	82.3	49.04
1926	108.2	64.8	41.91	1,389.6	43.32	46.72
1927	179.3	36.6	49.05	1,621.6	247.1	47.19
Territorial Airlines						
1918	N.A.	—	—	4.7	4.2	43.07
1919	8.1	3.7	45.59	41.1	30.6	77.94
1920	9.2	8.1	34.24	91.0	92.9	21.39
1921	10.9	10.9	34.24	10.0	9.7	41.65
1922	14.3	3.1	55.23	24.1	16.8	97.42
1923	13.9	7.4	44.74	24.6	20.8	47.17
1924	15.2	7.6	46.66	24.7	22.7	94.04
1925	16.1	8.6	32.18	24.7	26.1	93.99
1926	14.6	8.5	47.10	44.9	43.8	44.50
1927	18.9	7.6	54.21	34.7	29.8	87.70
Helicopter Airlines (not separately)						
1918	—	—	—	—	—	—
1919	—	—	—	—	—	—
1920	100	28	28.12	—	—	200
1921	109	40	22.12	—	—	400
1922	93	78	41.04	—	—	410
1923	206	129	34.74	291	26	13.41
1924	109	162	39.19	716	182	25.54
1925	109	162	39.19	716	182	25.54
1926	434	195	44.92	1,356	628	36.77
1927	167	477	46.03	3,551	1,500	44.37
1928	1,054	480	42.61	8,649	3,270	48.18

See Footnote at Bottom of Page 11.

**Available Service
and Utilization**
(continued)

International and Overseas Airlines

	Available For Miles Flown	Revenue Per Miles Flown	Total Miles Flown	Available Passenger Miles Flown	Revenue Passenger Miles Flown	Revenue Cargo Miles Flown
1938	—	54.2 ^a	114,1	153,2	65,81	7.0
1939	211,7	48.1	1,621,7	1,102,7	78,83	19.6
1940	—	54.9	2,093.3	1,688.9	52,33	9.1
1941	409.8	54.9	—	—	—	—
1942	318.2	54.8	2,159.4	2,056.4	59,70	9.9
1943	419.7	47.39	8,688.1	8,028.8	62,20	10.6
1944	—	47.39	—	—	—	—
1945	745.3	47.44	2,462.3	1,270.1	40,95	10.6
1946	514.1	47.48	8,254.9	7,762.3	88,84	11.1
1947	—	47.48	—	—	—	—
1948	110.4	83.8	1,312.7	1,012.9	61,95	10.7
1949	1,104.0	241.2	4,618.0	3,112.3	83,38	14.8
1950	1,272.8	324.7	4,119.1	3,570.7	63,04	14.7

American Airlines

	Available For Miles Flown	Revenue Per Miles Flown	Total Miles Flown	Available Passenger Miles Flown	Revenue Passenger Miles Flown	Revenue Cargo Miles Flown
1938	—	—	—	—	—	—
1939	—	—	—	—	—	—
1940	28.1	54.32	42.4	38.4	44.12	4.6
1941	37.2	54.29	54.0	22.6	41.69	4.6
1942	26.3	16.6	66,80	10,82	74.2	42.16
1943	24.1	17.6	37.49	10,62	9.8	19.4
1944	24.4	17.6	36,91	7,70	47.15	9.6
1945	41.0	29.6	40,93	210.9	106.6	47.33
1946	94.7	69.8	48,93	28,61	137.0	47.27
1947	99.0	32.8	55,64	339.7	61.9	44.37

All Other Airlines

	Available For Miles Flown	Revenue Per Miles Flown	Total Miles Flown	Available Passenger Miles Flown	Revenue Passenger Miles Flown	Revenue Cargo Miles Flown
1938	—	—	—	—	—	—
1939	—	—	—	—	—	—
1940	—	—	—	—	—	—
1941	96.2	50.8	24,30	—	—	—
1942	118.6	10.2	81,92	—	—	—
1943	113.9	59.3	61,61	—	—	—
1944	128.8	18.8	81,48	—	—	—
1945	164.0	136.1	79,41	—	—	—
1946	221.5	249.8	39,34	—	—	—
1947	411.7	186.9	70,04	—	—	23.1

CONSOLIDATED INDUSTRY

	Available For Miles Flown	Revenue Per Miles Flown	Total Miles Flown	Available Passenger Miles Flown	Revenue Passenger Miles Flown	Revenue Cargo Miles Flown
1938	—	51.1 ^a	11,514.8	321.2	44,90	20.0
1939	1,211.7	207.8	1,618.5	5,661.7	57.07	60.1
1940	—	44.12	12,208.0	2,621.2	17.20	44.0
1941	1,211.7	5,204.1	17,779	14,800.3	16,611.6	18,04
1942	—	44.12	12,208.0	2,621.2	17.20	44.0
1943	1,211.7	1,944.9	31,21	18,504.1	14,260.8	16,03
1944	—	44.12	12,207.2	21,457.8	16,27	39.1
1945	1,211.7	1,944.9	31,61	18,504.1	14,260.8	16,03
1946	—	32,072	31,59	22,047.7	21,979	16,79
1947	3,119.7	3,081.2	41,71	34,660.1	34,314	27.1
1948	—	32,074	37,77	45,660.2	37,416.8	16,84
1949	3,119.7	4,891.5	54,64	31,403.6	31,403	33.8

^a Data not available for American Airlines in 1938 and 1944. Total revenue operations initiated in 1945.

* Revenue Per Miles data for items other than passenger too miles for International and Overseas carriers not available for 1938. These total data do not reflect those items.

^b All Other Airlines began operations in South America in 1948.

^c R.A. Not available.

Note: Available Too Miles and Revenue Per Miles includes charter operations; all other items are for scheduled service only.

REVENUE TON MILES OF TRAFFIC CARRIED

U. S. Scheduled Airline Industry

(For Selected Years, In Thousands of Revenue Ton-Miles)

THIS TABLE SHOWS BY CATEGORIES THE EVER INCREASING USE OF THE SCHEDULED AIRLINES BY PASSENGERS AND COMMERCE

Passenger ^a	U. S. Mail ^b	Passenger U. S. Mail ^b	Cargo ^c	Non Passenger ^a	
				Passenger	Mail ^b
Domestic Airline Traffic					
1938	—	43,847	—	7,444	—
1939	—	569,795	32,899	11,621	14,410
1940	—	403,227	37,818	23,971	3,104
1941	—	703,771	31,841	18,539	12,140
1942	—	1,251,693	48,298	48,555	11,129
1943	—	1,388,120	49,526	47,654	13,758
1944	—	1,540,195	49,659	51,169	40,228
1945	—	1,030,411	71,159	54,175	11,621
1946	—	2,054,089	77,705	53,971	20,012
1947	—	2,103,134	82,985	56,119	21,708
Local Service Airlines					
1938	—	447	18	24	4
1939	—	8,151	234	190	116
1940	—	11,134	610	623	118
1941	—	30,011	810	194	1116
1942	27,011	901	46	154	1,177
1943	40,323	756	330	1,263	1,188
1944	46,713	929	339	1,483	1,239
1945	64,384	1,192	344	1,487	1,230
1946	59,956	1,179	348	1,503	1,242
Territorial Airlines					
1938	—	401	3	—	3
1939	—	2,043	14	112	114
1940	—	4,329	53	134	104
1941	—	4,628	46	119	104
1942	—	8,611	83	138	93
1943	—	5,540	57	100	83
1944	—	8,111	58	147	83
1945	—	8,750	58	144	83
1946	—	8,748	45	145	76
1947	—	9,359	45	131	76
Microlight Airlines					
1938	—	—	—	—	—
1939	—	—	—	—	—
1940	—	—	—	—	—
1941	—	—	—	—	—
1942	—	—	—	—	—
1943	—	—	—	—	—
1944	—	—	—	—	—
1945	—	—	—	—	—
1946	—	—	—	—	—
1947	—	—	—	—	—

See Footnotes at Bottom of Page 10

**Revenue Ton Miles of
Traffic Carried**
(continued)

	Passenger ¹	Priority ² U.S. Mail	Non- Priority ³ U.S. Mail	Express	Freight	Charter Flights	Space Reserves	Total
International and Overseas Airlines*								
1968	—	12,791	—	—	—	—	81,191	—
1969	—	118,330	6,441	—	—	—	—	—
1970	259,843	12,222	41,480	6,412	7,090	8,336	210,303	—
1971	257,441	21,148	48,513	10,050	5,710	9,825	194,996	—
1972	219,669	22,046	—	—	2,184	7,044	110,087	404,800
1973	244,541	24,544	—	—	2,9	74,427	7,242	14,156
1974	264,504	33,353	—	167	14,446	10,790	14,156	104,720
1975	462,176	33,429	—	140	20,311	19,761	17,048	162,794
1976	524,367	33,119	—	169,326	10,020	21,957	260,171	—
1977	589,510	37,258	—	161,159	63,013	39,175	154,619	—

Alaska Airlines⁴

	Passenger ¹	Priority ² U.S. Mail	Non-Priority ³ U.S. Mail	Express	Freight	Charter Flights	Space Reserves	Total
All Others Airlines								
1968	—	—	—	—	—	—	—	—
1969	—	—	—	—	—	—	—	—
1970	2,109	297	—	1,017	9,599	46	12,554	—
1971	2,413	761	—	812	4,058	52	10,722	—
1972	7,654	1,281	—	4,952	955	58	14,933	—
1973	9,321	1,162	—	5,929	1,642	114	10,317	—
1974	9,860	2,658	—	6,118	2,034	116	9,426	—
1975	11,688	2,279	—	7,399	2,791	102	9,517	—
1976	14,771	3,382	—	8,048	3,927	261	9,411	—
1977	14,331	3,763	—	7,350	3,348	279	9,318	—

All Delta Airlines

	Passenger ¹	Priority ² U.S. Mail	Non-Priority ³ U.S. Mail	Express	Freight	Charter Flights	Space Reserves	Total
CONSOLIDATED INDUSTRY⁵								
1968	181,949	2,449	—	—	3,114 ⁶	—	476	111,172 ⁷
1969	183,818	21,119	—	16,054	14,446	—	11,200	203,464
1970	200,462	26,727	—	25,674	31,529	38,396	15,158	191,987
1971	209,176	31,510	—	21,293	39,429	23,115	20,871	134,976
1972	219,751	32,875	—	21,494	34,462	24,146	20,974	134,553
1973	274,942	72,218	1,318	45,697	33,489	37,417	19,497	7,334,179
1974	289,116	72,629	10,343	42,594	31,163	33,258	20,591	3,231,373
1975	294,713	77,168	9,494	50,861	33,664	46,710	20,014	3,321,128
1976	264,121	112,194	32,452	37,973	41,061	36,178	40,981	3,933,659
1977	301,658	94,794	34,893	46,231	37,679	35,438	40,361	4,011,763

NA Not available.

Data not available for Alaska Airlines in 1968 and 1969. Georgia Airlines began operation in fourth quarter of 1974.

Local Service operations reflected in 1968.

Delta Air Lines discontinued in 1972.

Passenger and Freight reflected in 1972.

Revenue ton miles for data other than passenger and freight for International and Overseas services not available for 1972; hence total does not reflect these items.

* Freight and general by International and Overseas Airlines reflected in Consolidated Industry Data.

** Data reflected in Consolidated Industry Data.

† Data reflected in 1972.

‡ Data reflected in 1972.

§ Data reflected in 1972.

|| Data reflected in 1972.

** Passenger ton miles were reduced to coincide with standard passenger weight as presented by the CAB effective Jan 1, 1972.

OPERATING REVENUES

U. S. Scheduled Airline Industry

(For Selected Years, In Thousands of Dollars)

THIS TABLE SHOWS THE DOLLARS OF SALES THE SCHEDULED AIRLINES EARNED FOR THE VARIOUS SERVICES THEY RENDER

**Domestic Trunk
Airlines**

	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979
Passenger	\$ 31,736	39,579	56,734	49,016	67,387	73,702	87,034	132,088	143,157	136,722
U. S. Mail	—	—	—	—	—	—	—	—	—	—
Priority	\$ 1,759	10,204	8,704	4,411	38,910	11,118	11,121	24,110	38,107	10,108
Non-Priority	—	—	—	—	—	—	—	—	—	—
Public Service Revenue ¹	—	—	—	—	—	—	—	—	—	—
Express	\$ 1	—	—	—	—	—	—	—	—	—
Freight	\$ 1,157 ²	4,981	13,035	21,939	29,935	31,160	32,021	41,626	43,179	49,801
Other	\$ 193	—	—	—	—	—	—	—	—	—
Total	\$ 41,251	51,193	63,335	64,139	149,018	131,793	129,210	133,346	134,321	142,920

**Local Service
Airlines³**

	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979
Passenger	\$ —	105	4,410	16,358	29,364	33,104	27,673	32,140	40,156	47,604
U. S. Mail	—	—	—	—	—	—	—	—	—	—
Priority	\$ —	1,689	18,911	14,841	31,177	1,203	1,138	1,084	1,084	1,084
Non-Priority	—	—	—	—	—	—	—	—	—	—
Public Service Revenue ⁴	—	—	—	—	—	—	—	—	—	—
Express	\$ —	12	32	230	417	402	416	416	279	231
Freight	\$ —	84	74	317	455	442	467	467	556	1,041
Other	\$ —	41	193	344	314	271	310	310	310	1,754
Total	\$ —	1,500	18,911	30,370	41,779	41,318	41,715	47,490	47,712	42,816

Territorial Airlines

	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979
Passenger	\$ 826	2,720	3,890	4,120	4,433	4,771	8,230	8,685	4,542	4,738
U. S. Mail	—	—	—	—	—	—	—	—	—	—
Priority	\$ 47	122	181	295	301	49	41	48	50	52
Non-Priority	—	—	—	—	—	—	—	—	—	—
Public Service Revenue ⁵	—	—	—	—	—	—	—	—	—	—
Express	\$ 5	—	—	—	—	—	—	—	—	—
Freight	\$ 129	228	301	291	942	472	732	782	731	731
Other	\$ 5	6	137	410	430	128	148	137	364	601
Total	\$ 834	3,380	4,810	4,878	4,936	4,798	6,790	7,114	7,420	7,204

Helicopter Airlines

	1970	1971	1972	1973	1974	1975	1976	1977	1978	1979
Passenger	\$ —	—	—	—	—	—	—	10	82	308
U. S. Mail	—	—	—	—	—	—	—	—	—	—
Priority	\$ —	—	—	—	—	—	—	—	—	—
Non-Priority	—	—	—	—	—	—	—	—	—	—
Public Service Revenue ⁶	—	—	—	—	—	—	—	—	—	—
Express	\$ —	—	—	—	—	—	—	—	—	—
Freight	\$ —	—	—	—	—	—	—	—	—	—
Other	\$ —	—	—	—	—	—	—	—	—	—
Total	\$ —	—	—	—	—	—	—	393	3,311	8,026

See Footnotes at Bottom of Page 34

Operating Revenues
(continued)

	1958	1948	1946	1945	1944	1943	1942	1941	1940	1939	1938	1937
International and Overseas Airlines												
Passenger	\$ 4,455	91,417	61,308	140,673	21,454	231,019	344,194	256,639	162,911	371,543	11,360	
U. S. Mail (Priority)	1,939	20,581	21,122	31,667	10,333	18,037	20,681	20,631	20,928	21,153		
Public Service Revenue*												
Revenue	\$ 11,412	19,428	15,763	17,171	26,709	31,141	1,693	8,358	6,610			
Growth	\$ 547	16,414	1,179	5,861	26,101	27,317	24,414	31,851	26,441	41,624		
Other	1,187	10,818	19,556	22,103	28,010	21,758	10,238	20,112	21,124			
Total	\$ 19,113	141,756	149,256	161,531	146,918	307,264	351,049	318,364	402,655	377,372		

American Airlines*

Passenger	\$ 2,480	2,758	1,801	8,111	4,479	8,161	16,200	11,360			
U. S. Mail	1,028	2,021	7,324	9,049	2,298	2,313	2,477	2,159			
Public Service Revenue*											
Cargo	\$ 127	807	1,414	1,851	1,107	2,044	2,718	2,189			
Other	1,158	3,803	1,364	1,516	1,662	3,247	2,640	4,230			
Total	\$ 3,667	9,438	15,361	21,320	19,104	22,324	21,382	24,259			

All-Cargo Airlines*

Passenger	\$ —											
U. S. Mail	—											
Promotional	\$ —											
Non-Airline	\$ —											
Passenger Revenue												
Passenger	\$ —											
Freight	\$ 6,930	14,495	14,120	13,758	10,640	23,344	30,121	30,121				
Other	3,111	2,545	4,131	3,881	8,231	34,486	16,320					
Total	\$ 12,041	17,041	21,216	14,155	20,935	51,830	46,441					

CONSOLIDATED INDUSTRY*

Passenger	\$ 31,755	34,025	38,731	69,731	102,927	112,371	114,202	130,377	134,176	137,770		
U. S. Mail	—											
Promotional	\$ 31,397	47,704	118,039	52,156	118,985	41,187	15,609	39,444	59,773	60,438		
Non-Promotional	—											
Passenger Revenue*												
Passenger	\$ 32,770	29,464	25,257	16,495	12,544	12,198	20,247	22,120	16,210			
Freight	\$ 1,980	6,108	18,363	37,448	49,170	26,620	19,648	30,415	45,294	126,333		
Other	\$ 1,912	10,097	10,454	45,112	48,339	16,893	39,733	64,444	159,679	110,761		
Total	\$ 40,680	40,659	107,679	93,682	124,812	142,298	140,118	145,037	121,150			

* Preliminary.

NA Not Available.

Data not available for Alaska unless in 1938 and 1940.
All-Cargo airfreight revenue in fourth quarter of
1945. Local Service revenues reflected in 1946. Multi-
carrier operations started in 1947.

* Express and freight revenues are not reflected in the
consolidated industry totals.

* Prior to October 1, 1941, public service revenues were
not segregated from express and passenger.

DISTRIBUTION OF OPERATING EXPENSES

U. S. Scheduled Airline Industry
(For Selected Years, in Thousands of Dollars)*

THIS TABLE SHOWS HOW THE AIRLINES SPEND THEIR DOLLARS TO INSURE
FAST, SAFE, ECONOMICAL FLYING OPERATIONS AND EFFICIENT PASSENGER
AND CARGO HANDLING.

Explanation of New Classification of Operating Expenses

The classification of operating expenses is different from that used in prior years. Due to a revision of the form in which the current report is cast, it is not feasible to bring forward original 1946 figures which previously appeared in Part II and Figures. For this reason the data shown here for years prior to 1946 were recast for the publication into the format of the new report. Approximate results for 1946 are included. The data for 1946 and 1947 are as reported by the carriers. As through the availability of prior year data with HSD, it is not practical to make detailed adequate for general use state-by-state data at present.

The classification of expense is identical to just below of "Facts and Figures," except grouped as follows:

NEW CLASSIFICATION	OLD CLASSIFICATION
Promotion and sales	Marketing and selling
General and administration	General and administrative
Depreciation and amortization	Depreciation—Right equipment Amortization—Right equipment

As pointed out above, the method of making accounts is not perfect. To throw off HSD dollar in the following reports from those shown for 1946 and earlier.

1) Amortization of other delayed charges,* delayed depreciation for 1946 and before, is grouped in "Depreciation and amortization" in 1947.

2) Legal fees and expenses* are grouped in several accounts prior to 1947, it is now General and administration.

3) Fuel, oil, taxes and royalties unless otherwise indicated as General and administrative, otherwise 1946 and 1947 are distributed in other appropriate accounts.

4) Airport related office expenses reflected in Promotion and Sales for earlier years under "Miscellaneous and Selling."

5) State subsidies and development expense are classified as operating expense in prior years are included in "Depreciation and Amortization," in 1947.

NEW CLASSIFICATION	OLD CLASSIFICATION	1938	1946	1948	1942	1943	1944	1945	1947	1949
Domestic Trunk Airlines										
Flying Operations										
Maintenance	\$ 1,457	9,739	10,154	12,866	15,384	20,839	20,124	30,389	30,076	40,612
General Services & Administration	—	7,495	10,166	11,169	11,405	12,208	13,116	14,405	15,209	19,503
Passenger Service	\$ —	—	24,916	21,110	10,875	47,049	31,119	31,125	31,103	36,811
Aircraft & Vehicle Servicing	\$ —	—	9,154	44,815	18,941	34,054	11,229	11,229	11,229	11,229
Promotion & Sales	\$ 4,231	—	—	—	—	—	—	—	—	—
General & Administrative	\$ 3,259	—	—	—	—	—	—	—	—	—
Total G.S. & A.	\$ 15,537	—	18,110	106,319	109,793	277,686	108,713	349,318	409,449	475,119
Depreciation and amortization	\$ 8,848	28,173	47,252	46,267	44,043	87,822	104,118	101,791	148,321	148,952
Total Operating Expenses	\$ 40,317	111,121	111,299	144,308	142,915	296,431	299,919	310,304	312,230	317,970

See Footnote at bottom of Page 21.

Distribution of Operating Expenses (continued)

	1981	1980	1979	1978	1977	1976	1975	1974	1973	1972	1971	1970	1969	1968	1967
Local Service Airlines*															
Flying Operations	\$ 497	4,412	4,228	3,394	15,746	13,246	10,968	11,416	16,487						
Maintenance	\$ 848	3,653	3,286	2,621	10,018	9,418	12,844	12,010	16,026						
General Services & Administration															
Passenger Service	\$ 12	418	1,396	1,744	3,211	3,185	2,167	3,388	4,609						
Aircraft & Traffic Servicing	\$ 441	2,102	4,795	7,264	8,406	8,762	7,651	11,187	20,947						
Promotion & Sales	\$ 115	1,458	3,261	6,046	7,419	8,148	9,387	11,391	8,676						
General & Administration	\$ 179	1,404	2,494	3,693	4,039	4,108	5,391	4,999	4,999						
Total U.S.A.	\$ 1,114	6,973	11,761	19,930	23,091	23,016	26,023	21,318	31,011						
Depreciation & Amortization	\$ 101	1,048	1,036	2,953	3,919	3,698	2,039	3,316	3,749						
Total Operating Expenses	\$ 3,040	15,559	27,258	41,477	60,925	61,103	36,764	48,311	57,711						
Territorial Airlines															
Flying Operations	\$ 187	578	746	1,231	3,673	1,875	3,008	1,532	3,110	3,354					
Maintenance	\$ 117	467	716	943	548	1,117	1,243	1,118	1,219	1,209					
General Services & Administration															
Passenger Service	\$ 18	140	190	315	319	198	248	312	224						
Aircraft & Traffic Servicing	\$ 159	704	861	1,061	1,121	1,265	1,264	1,317	1,386						
Promotion & Sales	\$ 250	403	833	1,068	911	797	1,040	1,112	1,127						
General & Administration	\$ 14	407	661	241	199	163	568	568	667						
Total U.S.A.	\$ 147	1,605	2,096	2,671	3,154	3,100	3,112	3,919	3,194						
Depreciation & Amortization	\$ 127	258	411	412	234	469	594	405	514	498					
Total Operating Expenses	\$ 946	3,053	4,411	8,216	9,972	8,767	9,000	12,310	12,707	11,903					
Helicopter Airlines*															
Flying Operations	\$ —	—	—	94	285	364	561	803	814	417	1,138				
Maintenance	\$ —	—	—	97	162	117	721	802	811	915	1,281				
General Services & Administration															
Passenger Service	\$ 1	—	—	—	—	—	—	—	—	—	—				
Aircraft & Traffic Servicing	\$ 1	—	—	21	98	118	238	314	418	564	—				
Promotion & Sales	\$ 1	—	—	2	3	—	167	119	212	—	—				
General & Administration	\$ 1	—	—	47	112	104	204	146	202	494	—				
Total U.S.A.	\$ 1	—	—	79	218	109	639	821	1,019	1,233	3,242				
Depreciation & Amortization	\$ 1	—	—	88	113	144	413	415	451	445	811				
Total Operating Expenses	\$ 1	—	—	346	712	1,006	2,347	2,819	3,795	3,656	8,146				
International Small and Overseas Airlines															
Flying Operations	\$ 33,447	67,143	30,768	37,348	91,497	51,355	30,581	125,603	163,793						
Maintenance	\$ 24,818	44,448	41,498	54,231	84,617	61,113	38,798	72,807	79,447						
General Services & Administration															
Passenger Service	\$ 5,295	16,026	16,039	19,384	20,027	22,172	26,773	31,083	32,879						
Aircraft & Traffic Servicing	\$ 27,397	31,208	21,618	19,754	42,119	42,197	46,870	47,863	47,051						
Promotion & Sales	\$ 18,398	31,211	36,114	40,494	41,419	44,846	41,889	70,003	26,788						
General & Administration	\$ 14,338	22,167	22,176	24,714	23,354	28,708	24,291	33,809	24,447						
Total U.S.A.	\$ 71,975	101,123	104,011	103,930	141,344	149,015	128,814	183,866	194,718						
Depreciation & Amortization	\$ 10,813	22,383	19,613	29,183	24,120	11,203	10,894	30,368	30,368						
Total Operating Expenses	\$ 14,302	19,843	24,783	104,333	104,247	103,959	112,464	116,164	49,541	40,238					

See Footnotes at bottom of Page 23

Distribution of Operating Expenses (continued)

	1981	1980	1979	1978	1977	1976	1975	1974	1973	1972	1971	1970	1969	1968	1967
Alaska Airlines*															
Flying Operations	\$ —	—	—	—	—	—	—	—	—	—	—	—	—	—	—
Maintenance	\$ —	—	—	—	—	—	—	—	—	—	—	—	—	—	—
General Services & Administration															
Passenger Service	\$ 268	168	194	211	244	241	250	277	264	264	264	264	264	264	264
Aircraft & Traffic Routing	\$ 168	164	2,003	2,444	2,612	2,700	2,721	3,000	3,000	3,000	3,000	3,000	3,000	3,000	3,000
Promotion & Sales	\$ 129	134	1,371	1,441	1,449	1,459	1,469	1,479	1,479	1,479	1,479	1,479	1,479	1,479	1,479
General & Administration	\$ 983	1,333	1,767	1,892	1,912	1,912	1,912	1,912	1,912	1,912	1,912	1,912	1,912	1,912	1,912
Total U.S.A.	\$ 1,213	2,390	6,037	10,007	11,707	12,707	13,707	17,707	17,707	17,707	17,707	17,707	17,707	17,707	17,707
Depreciation & Amortization	\$ 931	1,038	1,035	1,145	1,395	1,128	1,394	1,394	1,394	1,394	1,394	1,394	1,394	1,394	1,394
Total Operating Expenses	\$ 1,810	3,928	9,975	18,105	21,715	22,715	23,715	27,715	27,715	27,715	27,715	27,715	27,715	27,715	27,715
All Cargo Airlines															
Flying Operations	\$ 4,633	4,382	4,042	3,842	3,556	3,356	3,156	2,956	2,756	2,556	2,356	2,156	1,956	1,756	1,556
Maintenance	\$ —	—	1,749	1,603	1,581	1,556	1,536	1,516	1,496	1,476	1,456	1,436	1,416	1,396	1,376
General Services & Administration															
Passenger Service	\$ 21	309	342	342	342	342	342	342	342	342	342	342	342	342	342
Aircraft & Traffic Servicing	\$ 1,203	2,263	2,044	2,112	2,196	2,263	2,331	2,400	2,469	2,538	2,607	2,676	2,745	2,814	2,883
Promotion & Sales	\$ 1,091	1,213	1,012	1,012	1,012	1,012	1,012	1,012	1,012	1,012	1,012	1,012	1,012	1,012	1,012
General & Administration	\$ 1,242	1,283	1,260	1,260	1,260	1,260	1,260	1,260	1,260	1,260	1,260	1,260	1,260	1,260	1,260
Total U.S.A.	\$ 4,088	8,079	8,309	8,309	8,309	8,309	8,309	8,309	8,309	8,309	8,309	8,309	8,309	8,309	8,309
Depreciation & Amortization	\$ 201	404	991	1,128	1,257	1,386	1,516	1,645	1,774	1,903	2,032	2,161	2,289	2,418	2,547
Total Operating Expenses	\$ 4,289	15,517	18,577	19,440	20,202	22,314	24,443	26,574	28,705	30,836	32,967	35,108	37,249	39,390	41,531

P Preliminary

* Data not available for Alaska Airlines in 1970 and 1971. All cargo airlines began operations in fourth quarter of 1966. Total Service Corporation began operations in 1968. Air Charter, Inc. began operations in 1970.

† Detailed expense breakdown not available

** Total is greater than sum of individual service corporations since recognition of expense was not reported by some of the carriers.

SUMMARY OF PROFIT OR LOSS

U. S. Scheduled Airline Industry

(For Selected Years, In Thousands of Dollars)

THIS TABLE SHOWS THE AMOUNT OF DOLLARS THE AIRLINES WERE ABLE TO KEEP FROM PURCHASE OF NEW AIRCRAFT AND PAYMENT OF DIVIDENDS TO STOCKHOLDERS. IT ALSO SHOWS THESE DOLLARS AS A PER CENT OF SALES AND THE RATIO OF TOTAL RETURN TO INVESTMENT.

1958 1960 1962 1964 1966 1968 1970 1972 1974 1976 1978 1980¹

Domestic Trunk Airlines

Total Operating Revenue	\$ 42,280	31,092	412,162	129,127	788,019	479,393	531,218	131,346	1312,021	1,419,213
Total Operating Expenses	5 43,107	31,120	411,278	128,338	472,051	176,021	379,338	130,048	1,342,230	1,219,311
Net Operating Income	\$ (826)	(128)	1,085	4,791	58,967	41,372	99,889	12,279	160,061	41,421
Interest on Long-Term Debt	\$ N/A	N/A	4,284	4,331	5,362	8,039	6,926	4,549	7,764	16,261
Net Other Non-Operating Income	\$ N/A	N/A	794	1,048	12,035	14,147	9,420	16,380	20,837	29,213
Income Tax	\$ N/A	(93)	1,063	29,940	41,448	50,579	59,804	36,041	29,045	29,045
Net Profit or Loss	\$ N/A	(107)	1,063	29,940	41,448	50,579	59,804	36,041	29,045	29,045

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 14.1 11.8 11.2 11.8 8.4 4.8

— 10.0 8.0 8.2 8.0 <1.0 1.9

Local Service Airlines⁴

Total Operating Revenue	\$ —	1,921	18,921	27,810	41,179	47,158	54,718	57,466	47,213	63,016
Total Operating Expenses	\$ 2,640	15,038	21,268	41,497	52,920	33,088	36,764	46,275	42,711	44,211
Net Operating Income	\$ (719)	343	444	1,119	1,145	1,416	1,934	1,981	4,501	1,801
Interest on Long-Term Debt	\$ N/A	97	33	187	252	148	215	354	228	—
Net Other Non-Operating Income	\$ N/A	(712)	(114)	558	1,020	(382)	341	364	(105)	—
Income Tax	\$ —	17	44	270	141	147	155	161	205	7
Net Profit or Loss	\$ (301)	(301)	(816)	(1,036)	558	(382)	(361)	(1,000)	—	—

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 13.1 3.7

— 1.8 0.4 —

Territorial Airlines

Total Operating Revenue	\$ 319	3,311	4,460	5,113	6,049	6,704	8,390	7,416	7,418	8,304
Total Operating Expenses	\$ 545	3,017	4,412	4,716	5,172	5,700	6,370	7,035	7,090	7,090
Net Operating Income	\$ 44	348	317	770	206	(370)	123	123	123	123
Interest on Long-Term Debt	\$ N/A	N/A	—	1	187	179	164	97	188	—
Net Other Non-Operating Income	\$ N/A	N/A	(14)	(67)	(15)	(11)	(14)	(14)	(14)	—
Income Tax	\$ N/A	142	45	30	(41)	(41)	7	7	18	—
Net Profit or Loss	\$ N/A	191	(166)	144	41	(417)	(139)	13	242	—

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 1.2 3.9 —

— 1.0 0.2 —

Hawaiian Airlines⁵

Total Operating Revenue	\$ —	—	373	793	1,044	2,408	3,079	2,159	2,715	3,831
Total Operating Expenses	\$ 546	733	26	10	(4)	238	412	409	46	1,020
Net Operating Income	\$ —	—	—	—	—	—	—	—	—	—
Interest on Long-Term Debt	\$ —	—	—	—	—	—	—	—	—	—
Net Other Non-Operating Income	\$ (29)	(28)	(41)	(41)	(41)	(41)	165	(34)	24	—
Income Tax	\$ —	—	4	11	18	148	202	8	(11)	—
Net Profit or Loss	\$ —	—	(4)	21	(41)	193	189	(26)	(11)	—

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 6.2 4.8 10.8 —

— 1.8 2.1 15.2 —

See Footnotes at End of Page 22

Summary of Profit or Loss (continued)

U. S. Scheduled Airline Industry
(For Selected Years, In Thousands of Dollars)

International and Overseas Airlines

Total Operating Revenue	\$ 18,143	144,354	242,294	340,110	314,918	320,284	166,847	284,358	452,946	487,932
Total Operating Expenses	\$ 14,820	121,387	238,101	348,101	304,267	317,102	162,654	267,941	462,148	471,017
Net Operating Income	\$ 3,323	23,971	1,193	32,010	10,651	13,176	1,193	16,417	91,799	95,795
Interest on Long-Term Debt	\$ N/A	N/A	297	2,054	1,018	2,391	2,489	1,120	3,028	4,882
Net Other Non-Operating Income	\$ N/A	N/A	14,120	4,347	5,615	1,733	6,250	8,354	8,186	9,967
Income Tax	\$ N/A	12,016	2,615	3,429	6,651	12,016	12,016	8,322	12,770	12,770
Net Profit or Loss	\$ N/A	(4,331)	4,344	10,006	7,739	12,000	16,000	13,434	30,800	34,448

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 4.4 7.0 8.8 4.8 8.2 7.6

— 2.8 3.7 4.1 2.9 4.8 3.8

Airlines⁶

Total Operating Revenue	\$ —	8,149	9,418	10,901	10,204	22,334	21,312	24,158
Total Operating Expenses	\$ 8,029	9,190	10,209	10,102	10,165	21,316	21,146	24,155
Net Operating Income	\$ 120	229	718	799	782	1,019	1,159	1,013
Interest on Long-Term Debt	\$ —	32	38	52	108	165	74	222
Net Other Non-Operating Income	\$ —	(11)	196	209	(79)	221	228	(91)
Income Tax	\$ —	45	57	173	103	231	216	419
Net Profit or Loss	\$ —	185	476	1,003	766	913	426	1,117

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 16.9 16.2 8.3 10.1

— 2.9 4.7 3.0 3.8

All-Cargo Airlines⁷

Total Operating Revenue	\$ —	17,361	17,067	19,116	16,905	29,695	30,935	31,448
Total Operating Expenses	\$ 17,017	16,917	17,295	17,772	16,405	30,979	31,079	31,441
Net Operating Income	\$ 343	150	1,158	1,241	1,501	866	956	314
Interest on Long-Term Debt	\$ —	158	198	205	185	863	863	132
Net Other Non-Operating Income	\$ —	224	1,412	1,799	1,191	1,132	6,079	1,638
Income Tax	\$ —	911	308	1,157	1,157	860	860	1,280
Net Profit or Loss	\$ —	1,364	1,419	2,359	1,705	1,083	1,822	1,317

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 16.7 15.8 9.4 3.0 0.8

— 9.8 11.8 6.4 3.4 3.4

CONSOLIDATED INDUSTRY⁸

Total Operating Revenue	\$ —	13,319	442,701	491,207	509,587	1,146,031	1,193,310	1,085,466	1,034,925	1,039,180	3,116,393
Total Operating Expenses	\$ 8,465	441,645	474,704	519,177	508,177	1,193,021	1,202,146	1,049,174	1,024,379	1,016,320	3,101,289
Net Operating Income	\$ 4,854	5,066	23,003	72,030	1,410	13,902	13,183	36,293	25,748	23,859	15,104
Interest on Long-Term Debt	\$ —	2,979	1,008	1,007	1,007	5,008	5,015	5,015	5,015	5,015	46,849
Net Other Non-Operating Income	\$ 9,843	8,843	8,843	8,843	8,843	12,073	12,073	12,073	12,073	12,073	3,000
Income Tax	\$ —	8,843	12,073	12,073	12,073	12,073	12,073	12,073	12,073	12,073	3,000
Net Profit or Loss	\$ —	8,843	12,073	12,073	12,073	12,073	12,073	12,073	12,073	12,073	3,000

Ratio of Return on Investment² (%)

Profit Margin on Sales³ (%)

— 16.5 5.9 8.6 10.0 0.8 0.8

— 2.3 4.7 4.4 4.2 4.2 0.8

NA: Not Available

Data not available for All-Cargo Airlines. In 1958 and 1960, All-Cargo Airlines losses amounted to much greater than its revenues in 1958, and less than its revenues in 1960.

¹ Net income before interest and after taxes as a percent of average net worth and long-term debt.

² Profit as a percent of revenues.

³ Profitability.

⁴ Domestic revenue share.

ASSETS, LIABILITIES AND STOCKHOLDERS' EQUITY

U. S. Scheduled Airline Industry

(For Selected Years, In Thousands of Dollars)

THIS TABLE SHOWS THE FINANCIAL SITUATION
OF THE SCHEDULED AIRLINES

1968 1969 1970 1971 1972 1973

Assets, Liabilities and Stockholders' Equity (continued)

	1968	1969	1970	1971	1972	1973
Territorial Airlines						
Assets						
Current Assets	\$ 18,021	162,382	204,618	218,278	459,827	292,458
Investments and Special Funds	8,381	41,146	58,649	54,818	144,147	139,149
Flight Equipment	22,239	11,142	20,297	15,207	17,904	13,979
Reserve for Depreciation and Maintenance	16,071	10,086	17,142	20,292	331,170	3,518
Retired Property and Equipment	b	46,174	54,475	165,044	166,3	197,041
Reserve for Depreciation	b	16,040	40,947	48,936	97,017	91,247
Other Property	211	15,909	3,714	14,161	2,097	40,226
Balanced Changes	—	1,200	1,200	7,271	11,264	33,058
Other Assets	764	119	1,115	1,213	6,199	3,598
Total Assets	4 26,037	217,303	262,929	214,938	1,367,111	312,113
Liabilities and Equity						
Current Liabilities	\$ 4,807	106,461	170,197	241,942	310,074	301,128
Long Term Debt	3,594	91,573	100,442	118,013	214,671	204,457
Other Non-current Liabilities	—	206	—	—	—	1,188
Operating Reserves	231	1,038	3,751	8,791	18,496	41,227
Deferred Credits	1,196	8,258	11,357	12,613	27,644	41,227
Preferred Stock	2,000	40,000	40,000	40,000	40,000	40,000
Common Stock	20,176	82,911	42,876	67,140	101,084	101,101
Other Paid-in Capital	11,387	46,990	46,644	51,145	118,610	117,766
Retained Earnings	24,247	117,192	46,818	203,814	237,777	211,182
Dividends Declared	24,247	117,192	20,270	46,818	329,497	40,226
Total Liabilities and Equity	4 34,038	167,006	642,828	104,914	1,367,111	312,113

Domicile Trunk Airlines*

Assets

Current Assets	\$ 18,021	162,382	204,618	218,278	459,827	292,458
Investments and Special Funds	8,381	41,146	58,649	54,818	144,147	139,149
Flight Equipment	22,239	11,142	20,297	15,207	17,904	13,979
Reserve for Depreciation and Maintenance	16,071	10,086	17,142	20,292	331,170	3,518
Retired Property and Equipment	b	46,174	54,475	165,044	166,3	197,041
Reserve for Depreciation	b	16,040	40,947	48,936	97,017	91,247
Other Property	211	15,909	3,714	14,161	2,097	40,226
Balanced Changes	—	1,200	1,200	7,271	11,264	33,058
Other Assets	764	119	1,115	1,213	6,199	3,598
Total Assets	4 26,037	217,303	262,929	214,938	1,367,111	312,113
Liabilities and Equity						
Current Liabilities	\$ 4,807	106,461	170,197	241,942	310,074	301,128
Long Term Debt	3,594	91,573	100,442	118,013	214,671	204,457
Other Non-current Liabilities	—	206	—	—	—	1,188
Operating Reserves	231	1,038	3,751	8,791	18,496	41,227
Deferred Credits	1,196	8,258	11,357	12,613	27,644	41,227
Preferred Stock	2,000	40,000	40,000	40,000	40,000	40,000
Common Stock	20,176	82,911	42,876	67,140	101,084	101,101
Other Paid-in Capital	11,387	46,990	46,644	51,145	118,610	117,766
Retained Earnings	24,247	117,192	46,818	203,814	237,777	211,182
Dividends Declared	24,247	117,192	20,270	46,818	329,497	40,226
Total Liabilities and Equity	4 34,038	167,006	642,828	104,914	1,367,111	312,113

Local Service Airlines

Assets

Current Assets	\$ 1,028	7,494	11,027	14,811	16,722	
Investments and Special Funds	911	665	518	514	2,921	
Flight Equipment	2,211	10,865	11,610	14,395	16,298	
Reserve for Depreciation and Maintenance	129	5,021	9,972	12,428	18,112	
Retired Property and Equipment	442	2,048	4,114	4,114	4,114	
Reserve for Depreciation	129	1,650	3,620	3,694	3,694	
Other Property	472	2,16	4,04	2,071	1,211	
Balanced Changes	199	1,894	2,018	3,184	1,838	
Other Assets	179	2,001	2,118	2,118	2,118	
Total Assets	4 4,416	14,744	26,307	31,919	36,936	
Liabilities and Equity						
Current Liabilities	\$ 1,493	4,927	10,664	11,601	20,607	
Long Term Debt	369	1,465	1,721	1,959	8,498	
Other Non-current Liabilities	—	—	—	—	179	
Operating Reserves	82	397	614	1,198	—	
Deferred Credits	128	148	42	355	108	
Preferred Stock	440	1,000	1,000	422	143	
Common Stock	1,170	2,925	4,114	3,717	4,114	
Other Paid-in Capital	2,785	4,115	5,684	4,935	4,873	
Retained Earnings	4,102	10,016	10,011	14,920	10,110	
Stockholders' Equity—Net	4 3,662	9,319	18,018	11,251	35,941	
Total Liabilities and Equity	4 8,408	14,744	26,307	31,919	36,936	

- a. FAA required net property and equipment only.
- b. Recovery and depreciation not segregated from flight equipment.
- c. Other noncurrent liabilities not segregated from long term debt.
- d. Income sheet data for domestic trunk airlines reflects international as well as domestic operations.

Note: Change in the captioned components of the U.S. airline industry in 1972 data to 1973 are not directly comparable with other years. However, the data have been placed in as to make them generally comparable.

International and Overseas Airlines

Assets

Current Assets	\$ 2,138	19,391	34,013	30,989	31,560	32,933
Investment and Special Funds	3,918	21,213	34,827	21,334	42,723	34,827
Flight Equipment	1,742	10,074	22,395	22,395	22,395	22,395
Reserve for Depreciation and Maintenance	1,162	5,116	8,816	8,816	14,074	12,238
Retired Property and Equipment	b	3,415	32,611	32,611	31,913	32,933
Reserve for Depreciation	b	3,415	4,873	4,873	19,016	20,233
Other Property	364	3,145	2,000	2,000	2,000	2,000
Balanced Changes	212	1,152	21,494	4,449	4,449	4,449
Other Assets	8,821	8,821	8,821	8,821	8,821	8,821
Total Assets	4 26,016	107,934	218,331	240,938	107,934	302,933
<i>Unaudited and Domestic Airlines Included in total assets</i>						

See Footnotes at End of Page 12

**Assets, Liabilities and
Stockholders' Equity**

(continued)

1936 1946 1956 1966 1976 1986 1987

International and Overseas Airlines

Carriers and Carriers

	\$	4,005	39,461	52,643	61,624	82,150	97,722
Current Assets		1,000	11,100	11,100	11,100	11,100	11,100
Long-Term Assets		3,000	28,361	29,673	32,426	32,426	32,426
Other Non-Current Assets		0	0	294	294	571	571
Capital or Reserves		5	16,870	4,764	3,296	3,184	3,184
Deferred Credits		5	348	5,045	4,287	4,386	4,386
Deferred Charges		5	0	0	0	0	0
Cash and Bank		5	5,034	18,913	19,916	19,792	11,100
Other in Capital		5	30,001	21,516	42,076	63,128	45,680
Total Assets		5	44,006	47,950	52,808	54,970	54,970
Liabilities and Capital		5	21,511	35,919	37,412	120,944	120,944
Total Liabilities and Capital		5	24,506	37,956	51,223	242,916	242,916
					321,167	333,177	

American Airlines¹

Assets

Current Assets	\$	1,765	4,093	7,125	7,504	
Investment and Special Funds		0	100	400	500	
Right Retainage		0	4,630	8,231	11,264	11,264
Reserves for Depreciation and Maintenance		5	3,157	5,265	6,929	6,929
General Property and Equipment		5	141	1,418	4,230	4,495
Reserves for Depreciation		5	100	1,000	1,000	1,000
Other Property		5	100	100	100	100
Deferred Charges		5	204	170	422	472
Other Assets		5	192	140	124	124
Total Assets		5	8,841	10,280	14,887	14,924

Liabilities and Equity

Current Liabilities	\$	2,802	3,497	4,120	2,648	
Long-Term Debt		0	496	1,628	3,063	4,956
Other Non-Current Liabilities		5	0	0	0	19
Deferred Credits		5	314	375	259	259
Deferred Credit		5	36	107	82	82
Purchased Stock		5	84	0	0	0
Common Stock		5	2,184	2,184	2,184	2,184
Other Paid-in Capital		5	2,918	2,589	2,672	2,645
Retained Earnings		5	10,121	10,821	11,160	11,160
Stockholders' Equity—Total		5	2,680	8,862	16,798	16,852
Total Liabilities and Equity		5	8,841	10,280	14,887	14,924

All-Cargo Airlines

Assets

Current Assets	\$	5,623	3,083	11,124	19,510	
Investment and Special Funds		5	279	12,544	14,767	14,767
Right Retainage		5	2,079	20,946	31,100	41,425
Reserves for Depreciation and Maintenance		5	803	1,183	1,279	1,279
General Property and Equipment		5	1,049	2,246	3,640	5,104
Reserves for Depreciation		5	961	1,116	1,032	1,032
Other Property		5	0	0	0	0
Deferred Charges		5	453	235	2,828	3,110
Other Assets		5	0	0	0	0
Total Assets		5	10,124	25,879	49,540	44,136

Liabilities and Equity

Current Liabilities	\$	3,683	6,994	19,390	38,568	
Long-Term Debt		5	511	7,987	11,823	18,172
Other Non-Current Liabilities		5	0	0	0	0
Deferred Credits		5	121	1,078	2,211	297
Deferred Charges		5	45	51	2,057	2,192
Purchased Stock		5	993	1,660	1,637	1,637
Common Stock		5	4,083	8,117	8,150	7,986
Other Paid-in Capital		5	2,732	7,295	14,346	14,346
Retained Earnings		5	(2,611)	9,102	6,101	4,495
Stockholders' Equity—Total		5	9,043	20,073	38,568	31,118
Total Liabilities and Equity		5	9,026	20,078	49,540	44,136

See Footnotes at Bottom of Page 26

**Assets, Liabilities and
Stockholders' Equity**

(continued)

1936 1946 1956 1966 1976 1986 1987

Consolidated Industry

Assets

General Assets	\$	10,311	12,159	31,439	48,406	48,146	50,158
Investment and Special Assets		5	4,385	3,139	40,919	16,412	19,441
Right Retainage		5	49,385	142,191	221,519	152,491	140,720
Reserve for Depreciation and Maintenance		5	78,081	34,292	77,035	116,244	26,256
General Prepaid and Equipment		5	—	62,244	112,556	185,712	119,461
Reserve for Supplies		5	—	10,249	10,249	10,249	10,249
Other Prepaid Assets		5	219	19,554	8,343	10,317	9,644
Related Companies		5	1,112	24,212	46,923	1,846	22,741
Other Assets		5	4,386	31-9	1,483	4,386	3,748
Net Assets		5	44,196	101,559	197,009	147,344	144,164

Liabilities and Equity

General Liabilities	\$	14,097	142,497	241,493	481,381	481,487	
Long Term Debt		5	4,700	60,425	101,054	107,549	107,549
Other Non-Current Liabilities		5	115	56,041	56,041	56,041	56,041
Deferred Credits		5	7,250	14,249	14,249	14,249	14,249
Deferred Charges		5	1,072	28,187	27,052	14,898	41,104
Prepaid Assets		5	600	6,649	6,649	24,842	47,708
Common Stock		5	30,178	76,312	37,745	91,934	106,914
Other Stock Capital		5	41,100	76,012	54,247	54,247	101,795
Retained Earnings		5	1,199	9,443	10,234	10,234	10,234
Stockholders' Equity—Total		5	32,310	107,179	107,009	106,911	144,158
Total Liabilities and Equity		5	44,196	101,559	197,009	147,344	144,164

See Footnotes at Bottom of Page 26

DOMESTIC INTERCITY PASSENGER MILE MARKET

(For Selected Years, In Millions)

1936 1946 1956 1966 1972 1973 1974 1975 1976 1977

Passenger and Air Travel

Rail Passages (Mile ¹)	7,164	17,010	11,018	9,208	5,904	7,268	4,150	4,946	5,249	
Air Passages	487	9,112	9,035	4,179	10,010	10,975	13,329	12,823	14,083	15,740
Air-Car-Cab	5	1,067	1,236	3,715	5,321	6,756	8,274	8,516	8,810	9,110
Total Air	487	9,112	9,035	4,179	12,601	16,681	16,975	17,741	17,324	16,290
Passenger and Air Combined	7,651	27,171	20,953	22,388	22,620	28,944	38,181	38,991	40,951	42,331
% Above Combined Total	5-16	33-19	46-33	46-33	46-33	46-33	46-33	46-33	46-33	46-33
Total Airline Travel	24,851	51,119	44,814	34,509	42,938	41,381	40,152	42,023	42,608	40,767
% Above Airline Travel	1-13	4-49	9-12	14-20	19-19	20-16	20-16	20-16	20-16	20-16
Passenger Intensity Rotatable ²	236,779	263,829	267,408	422,943	493,927	329,495	484,763	465,973	477,765	488,928
Total Commerce Center	263,120	344,019	203,114	403,913	503,104	370,495	465,783	445,992	477,765	488,928
% Above Passenger Intensity	1-16	1-16	1-17	2-20	2-20	2-20	2-20	2-20	2-20	2-20
Passenger Miles per Capita ³	1,530	3,444	2,811	3,965	3,638	3,738	3,738	3,738	3,738	4,218

¹ 1936-1941 from Federal Transportation Data Book; 1946-1950 from Bureau of the Budget; 1951-1955 from Bureau of the Budget; 1956-1960 from Bureau of the Budget; 1961-1965 from Bureau of the Budget; 1966-1970 from Bureau of the Budget; 1971-1975 from Bureau of the Budget; 1976-1980 from Bureau of the Budget.

² 1936-1941 from Federal Transportation Data Book; 1946-1950 from Bureau of the Budget; 1951-1955 from Bureau of the Budget; 1956-1960 from Bureau of the Budget; 1961-1965 from Bureau of the Budget; 1966-1970 from Bureau of the Budget; 1971-1975 from Bureau of the Budget; 1976-1980 from Bureau of the Budget.

³ 1936-1941 from Federal Transportation Data Book; 1946-1950 from Bureau of the Budget; 1951-1955 from Bureau of the Budget; 1956-1960 from Bureau of the Budget; 1961-1965 from Bureau of the Budget; 1966-1970 from Bureau of the Budget; 1971-1975 from Bureau of the Budget; 1976-1980 from Bureau of the Budget.

⁴ 1936-1941 from Federal Transportation Data Book; 1946-1950 from Bureau of the Budget; 1951-1955 from Bureau of the Budget; 1956-1960 from Bureau of the Budget; 1961-1965 from Bureau of the Budget; 1966-1970 from Bureau of the Budget; 1971-1975 from Bureau of the Budget; 1976-1980 from Bureau of the Budget.

⁵ 1936-1941 from Federal Transportation Data Book; 1946-1950 from Bureau of the Budget; 1951-1955 from Bureau of the Budget; 1956-1960 from Bureau of the Budget; 1961-1965 from Bureau of the Budget; 1966-1970 from Bureau of the Budget; 1971-1975 from Bureau of the Budget; 1976-1980 from Bureau of the Budget.

⁶ Based on 1936-1941 passenger traffic in billions of miles per capita.

⁷ Based on 1936-1941 passenger traffic in billions of miles per capita.

⁸ Based on 1936-1941 passenger traffic in billions of miles per capita.

⁹ Based on 1936-1941 passenger traffic in billions of miles per capita.

¹⁰ Based on 1936-1941 passenger traffic in billions of miles per capita.

REVENUE PASSENGERS CARRIED

U. S. Scheduled Airline Industry
(For Selected Years, In Thousands of Passengers)

	1950	1951	1952	1953	1954	1955	1956	1957	1958
Domestic Trunk Airlines	5,161	5,910	12,334	18,115	21,149	26,137	31,526	36,813	37,876
Tourist Service Airlines	—	31	436	547	1,736	3,002	2,413	3,037	3,463
International Airlines	21	211	418	477	518	522	541	511	427
Helicopter Airlines	—	—	—	—	1	9	29	42	182
Intermediated and Overseas Airlines	—	189	1,041	1,273	1,675	2,342	2,086	3,276	3,888
Airline Airlines	—	—	111	168	176	220	228	261	318
TOTAL SCHEDULED AIRLINE INDUSTRY	1,204	12,504	18,867	19,041	21,438	28,620	31,423	31,810	31,201

¹ Includes data for 1948 Plus 1950 passenger charter flights. Data prior to 1948 not available.

² Passengers reported in a different basis for 1957 from prior years; hence data not entirely comparable with prior years.

AVERAGE REVENUE PER PASSENGER MILE

Industry Common Carriers
(For Selected Years, In Cents per Mile)

	1950	1951	1952	1953	1954	1955	1956	1957	1958
Domestic Scheduled Airlines									
Czech or French	—	—	—	—	—	—	—	—	—
All Services	8.32	4.62	6.25	6.94	9.94	9.40	9.37	6.27	4.18
International Scheduled Airlines									
Czech or French	—	—	—	—	—	—	—	—	—
All Services	8.34	8.31	8.01	7.18	7.06	4.98	4.78	6.08	6.36
International Motor Buses									
Red Cross ¹	2.37	2.08	2.10	2.28	2.36	1.36	2.26	3.31	3.39
Czech	1.96	1.62	2.29	2.47	2.10	2.10	2.06	2.47	2.86
Total Inter-city Motor Bus	1.87	1.66	1.74	1.89	2.02	1.86	1.84	2.12	2.38

¹ Only offices.

² Partially estimated.

³ Data include payments to Pullman Company for road, hotel, etc.

⁴ 1958-First year available.

ICA. Not available.

Note: Average passenger fare is defined by dividing passenger revenue by passenger miles.

AIRCRAFT OWNED

By U. S. Scheduled Airline Industry
(For Selected Years)

THIS TABLE SHOWS HOW THE SIZE AND TYPE OF AIRCRAFT USED BY THE SCHEDULED AIRLINES HAS IMPROVED OVER THE YEARS

Aircraft	Type	1950	1951	1952	1953	1954	1955	1956
Beech	2400-3000	21	—	—	—	44	64	74
Boeing	707	—	—	—	—	—	102	108
Globe	120	—	—	—	—	3	121	134
Grumman	446	—	—	—	—	—	19	31
Cessna	C-185	—	—	—	—	79	94	95
Douglas	DH-DC-1	21	—	—	—	—	—	—
	DC-3	76	116	419	308	156	310	310
	DC-4	202	182	188	202	202	218	218
	DC-5	—	704	—	—	—	—	—
	DC-6	—	—	—	—	81	152	203
Learjet	Pratt	40	2	—	—	—	—	—
	Unidad	—	11	—	—	10	10	10
	Other early jetliners	14	6	15	125	103	112	121
	Cessna 310	—	—	35	—	—	—	—
Mitsubishi	MRJ	—	—	—	—	—	—	—
	202	—	—	—	21	28	25	31
	404	—	—	—	96	180	177	86
McDonnell	(series)	—	—	—	—	—	—	—
Vickers	Viceroy	—	—	—	—	—	56	99
Other	—	—	—	—	17	20	26	37
Total Fixed Wing	—	317	799	3129	1463	1766	3605	3605
Total Helicopters	—	—	—	18	26	19	36	36

AIRCRAFT ON ORDER

U. S. Scheduled Airlines
(As of January 1, 1958)

THIS TABLE SHOWS HOW THE SCHEDULED AIRLINES WILL CONTINUE TO ADD NEW AND FASTER AIRCRAFT TO INSURE IMPROVED SERVICE FOR THEIR CUSTOMERS

Aircraft Type	Total in Order	Year of Delivery
Boeing	—	—
Boeing 727	35	1961
Boeing 737	55	1962
Cessna 414	4	—
Douglas DC-8	35	1963
Prop jets	—	—
McDonnell Douglas	8	—
Pan American	21	1964
United	—	—
Varig	14	1965
Varig, Vickers Viscount	18	—
Pilots	—	—
Douglas DC-10-3	29	1966
Shaw DC-10	33	1967
Lockheed L-1011	4	—
Motoliner	—	—
Vickers 808	6	—
Vickers 888	2	—

COMPARATIVE TRANSPORT SAFETY RECORD

*Passenger Fatality Rate per 100,000,000 Passenger Miles
(For Selected Years)*

	1939	1944	1949	1950	1951	1952	1953	1954	1955	1956	1957	1958
Domestic Scheduled Airlines												
Passenger	28	25	83	94	46	36	16	15	15	10	10	10
Rate	\$ 4.1	\$ 2.7	\$ 1.40	\$ 1.21	\$ 1.7	\$ 0.9	\$ 0.8	\$ 0.7	\$ 0.6	\$ 0.4	\$ 0.4	\$ 0.4
International and Overseas Domestic Airlines												
Passenger	3	40	30	48	54	2	0	0	2	9	10	40
Rate	\$ 1.2	\$ 3.03	\$ 1.03	\$ 1.15	\$ 1.05	\$ 0.6	\$ 0.6	\$ 0.6	\$ 0.7	\$ 0.7	\$ 0.7	\$ 0.7
Motor Vehicles												
Passenger	4	175	128	102	100	70	60	100	80	80	80	80
Rate	\$ 0.18	\$ 0.18	\$ 0.17	\$ 0.16	\$ 0.15	\$ 0.11	\$ 0.11	\$ 0.11	\$ 0.11	\$ 0.11	\$ 0.11	\$ 0.11
Railroad Passenger Trains												
Passenger	78	115	22	188	14	10	23	18	13	13	14	14
Rate	\$ 1.36	\$ 1.18	\$ 1.17	\$ 0.9	\$ 0.6	\$ 0.4	\$ 0.3	\$ 0.3	\$ 0.3	\$ 0.3	\$ 0.3	\$ 0.3
Passenger Auto and Trail												
Passenger	11,000	18,400	16,500	17,000	20,000	19,000	22,000	24,500	26,100	26,100	26,100	26,100
Rate	\$ 2.0	\$ 2.5	\$ 2.1	\$ 2.2	\$ 2.6	\$ 2.8	\$ 2.8	\$ 2.6	\$ 2.7	\$ 2.7	\$ 2.7	\$ 2.7

¹ Above data not available prior to 1944.

² Motor fatality rates as of Passager Auto and Trail.

³ Preliminary.

⁴ Not Available.

COMPARISON OF RAIL AND AIR FARES WITH TRAVEL TIMES

CITY—CITY	FARE				TIME	
	CAR	SEAT	POST CLASS ¹	SEAT	AIR	SEAT
Baltimore—N.Y.	\$ 10.00	\$ 4.00	11.00	\$ 6.00	50	4½ hr
Seattle—Portland	\$ 12.16	29.80	42.00	48.20 ²	2½ hr	18½ hr
Chicago—New York	\$ 14.18 ³	36.50	47.96	61.72	2½ hr	15½ hr
New York—Miami	\$ 65.82 ⁴	92.52	91.00	129.64	3½ hr	24½ hr
New York—Washington	\$ 12.15	1.70	14.00	18.60	1½ hr	3½ hr
Los Angeles—New York	\$ 204.20	39.67	162.65	161.60 ⁵	2½ hr	61½ hr
Philadelphia—Detroit	\$ 14.32 ⁶	26.01	48.18	43.78	2½ hr	10½ hr
Chicago—Washington	\$ 18.60 ⁷	36.80	46.18	52.94	2½ hr	18½ hr
Washington—New Orleans	\$ 43.02 ⁸	34.49	47.95	61.59	3½ hr	24½ hr
Atlanta—Chicago	\$ 28.02 ⁹	22.21	47.76	42.61	2½ hr	11½ hr
Atlanta—Dallas	\$ 30.11 ¹⁰	39.95	32.45	44.84	2½ hr	14½ hr
Gainesville—Miami	\$ 42.95 ¹¹	26.44	48.10	33.70	3½ hr	19½ hr
Connells—Pittsburgh	\$ 16.00	12.00	20.40	23.10	1½ hr	7½ hr
Chicago—Honolulu	\$ 46.65 ¹²	32.52	11.35	58.63	4½ hr	23½ hr
Chicago—St. Louis	\$ 14.25 ¹³	5.00	11.25	17.50	1½ hr	10½ hr
Los Angeles—Chicago	\$ 30.00	30.64	120.31	107.31	3½ hr	19½ hr
St. Louis—New Orleans	\$ 36.05 ¹⁴	19.43	47.03	38.41	2½ hr	14½ hr
Chicago—Kansas City	\$ 20.00	12.64	26.20	24.05	1½ hr	8½ hr
St. Louis—Dallas	\$ 18.00	23.03	30.00	42.44	4 hr	17½ hr
San Francisco—Honolulu	\$ 48.00	10.67	36.75	76.31	3½ hr	19½ hr
Seattle—San Francisco	\$ 72.72	52.01	102.31	61.04	4 hr	41½ hr
D. Louis—Los Angeles	\$ 16.00	56.00	106.05	101.71	6½ hr	41½ hr
D. Louis—Honolulu	\$ 44.00	47.00	87.40	89.47	0.17	12½ hr
Los Angeles—Honolulu	\$ 33.00	14.35	20.75	25.90	1½ hr	9½ hr
Portland—Seattle	\$ 3.18	6.00	12.70	12.94	—	8½ hr
Calif. City—Chicago—Portland	\$ 36.00	24.29	45.30	46.61	6 hr	25½ hr

¹ Round-trip price of lower berth.

² Economy when no first class is shown or available.

CLASSES OF UNITED STATES COMMERCIAL AIR CARRIERS

At the present time there are seven recognized classes of air carriers in the air transport industry of the United States. The classification is used by the Civil Aeronautics Board in connection with the economic regulation of the industry and under the Civil Aeronautics Act is based largely on the scope of operations authorized or allowed by the Act. Classes Due to War have conditions of limitation and diversity and similar regularly scheduled services.

1. The Domestic Trunk Lines include those carriers which primarily have passenger operating rights within the continental United States. These lines derive largely from operations by persons or predecessor companies marketing the Civil Aeronautics Act of 1938 which granted plane "franchise rights." There are currently twelve trunk lines, most of which operate high density feeder routes between the major urban centers of the United States.

Operating Basis	Continental Trunk Lines	National Trunk Lines	Trans World Airlines
Domestic operations	Alaska Air American Airlines Delta Eastern Midwest North Central Pan American TWA	Delta Midwest National Northeast Southwest United Western	Trans World Airlines

2. The Domestic Local Service Lines consist, with one exception, being completed early 1949. These carriers operate routes of lesser traffic density between the smaller urban centers and between these centers and some non-airline points of domestic road links like Miami and the Caribbean.

Operating Basis	Domestic Local Service Lines	Pan American Caribbean ¹
Domestic operations	Alaska Air American Airlines Delta Eastern Midwest National Northeast Southwest United Western	Pan American Caribbean ¹

3. The International and Overseas Lines include all U.S. based air carriers operating between the United States and foreign countries other than Canada. Some of these carriers conduct operations between foreign countries and some non-airline points of domestic road links like Miami and the Caribbean.

Operating Basis	International and Overseas Lines	Pan American Caribbean ²
Domestic operations	Alaska Air American Airlines Delta Eastern Midwest National Northeast Southwest United Western	Pan American Caribbean ²

4. The Territorial Lines include two groups of carriers. The broader lines operate in the U.S. island possessions in the Pacific and the Caribbean and the Alaska Lines operate between the U.S. and Alaska and within Alaska.

Territorial Lines	Alaska Lines	Operations within Alaska
Pacific Trans-Pacific Trans-Pacific Trans-Pacific	Alaska ³ Alaska ⁴ Alaska ⁵ Alaska ⁶ Alaska ⁷ Alaska ⁸ Alaska ⁹ Alaska ¹⁰ Alaska ¹¹ Alaska ¹² Alaska ¹³ Alaska ¹⁴ Alaska ¹⁵ Alaska ¹⁶ Alaska ¹⁷ Alaska ¹⁸ Alaska ¹⁹ Alaska ²⁰ Alaska ²¹ Alaska ²² Alaska ²³ Alaska ²⁴ Alaska ²⁵ Alaska ²⁶ Alaska ²⁷ Alaska ²⁸ Alaska ²⁹ Alaska ³⁰ Alaska ³¹ Alaska ³² Alaska ³³ Alaska ³⁴ Alaska ³⁵ Alaska ³⁶ Alaska ³⁷ Alaska ³⁸ Alaska ³⁹ Alaska ⁴⁰ Alaska ⁴¹ Alaska ⁴² Alaska ⁴³ Alaska ⁴⁴ Alaska ⁴⁵ Alaska ⁴⁶ Alaska ⁴⁷ Alaska ⁴⁸ Alaska ⁴⁹ Alaska ⁵⁰ Alaska ⁵¹ Alaska ⁵² Alaska ⁵³ Alaska ⁵⁴ Alaska ⁵⁵ Alaska ⁵⁶ Alaska ⁵⁷ Alaska ⁵⁸ Alaska ⁵⁹ Alaska ⁶⁰ Alaska ⁶¹ Alaska ⁶² Alaska ⁶³ Alaska ⁶⁴ Alaska ⁶⁵ Alaska ⁶⁶ Alaska ⁶⁷ Alaska ⁶⁸ Alaska ⁶⁹ Alaska ⁷⁰ Alaska ⁷¹ Alaska ⁷² Alaska ⁷³ Alaska ⁷⁴ Alaska ⁷⁵ Alaska ⁷⁶ Alaska ⁷⁷ Alaska ⁷⁸ Alaska ⁷⁹ Alaska ⁸⁰ Alaska ⁸¹ Alaska ⁸² Alaska ⁸³ Alaska ⁸⁴ Alaska ⁸⁵ Alaska ⁸⁶ Alaska ⁸⁷ Alaska ⁸⁸ Alaska ⁸⁹ Alaska ⁹⁰ Alaska ⁹¹ Alaska ⁹² Alaska ⁹³ Alaska ⁹⁴ Alaska ⁹⁵ Alaska ⁹⁶ Alaska ⁹⁷ Alaska 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19TH EDITION
Facts and Figures, 1958

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...even when his feet are on the ground!

His name is Lionel Olivier Case. His job: Supervisor of Flight Personnel of the Air France fleet. He spends most of his time in the air, checking planes and flight personnel. And even when he's back on the ground, his head-and heart-are still 4 miles up.

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of flying. He's at home behind the controls of a Constellation, Caravelle, Super Starliner—and the new Boeing 727 International jet.

Like all Air France personnel, Lionel Case was hand-picked for his intelligence and skill. He's one more reason you can fly Air France with confidence!

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DOUGLAS DC-8 jetliners in final production phase. Speed backs behind wing that will be installed on flat tail mold only.



NIGHT VIEW cockpit speed nose section, reptile-like wing. Aperture at nose is one of two air intakes for cabin air resealing.

Douglas Uses Extra Time to Refine DC-8 for Jet Race

By Russell Hawkes

Seattle, Wash.—Based with emergency funds which is fast on the scene, Douglas Aircraft Co.'s DC-8 is the product of carefully planned exploitation of all resources in being seized and a present program which Douglas engineers believe to be unprecedented in scale and thoroughness.

Douglas' chief advantage is being seized in the opportunity to build a newer airplane with later designs in the basic design. It is recognized in every manufacturer competing in the transoceanic transport field, as well

denied by limitations in freeze designs until the last minute and salvaging to design modified versions of the first opportunity. Opposite Douglas feels unique point set that the DC-8, the definitive example of success, was not the first of its class to see service.

In value, lack of a living prototype has been a disadvantage but the market for the DC-8 has been assured by Douglas' reputation for success and its fast-flying customer service network. Unavoidable complexity of modern aircraft and the high cost of down time have made the reliable service system a key factor in the sales competition and

have made worldwide logistic support for the customer a clear growth target for transport builders.

Until now, delivery dates have been the most important single factor in determining which airline an airline will buy. This can be demonstrated by a month-by-month comparison of the cumulative backlog of Douglas and Boeing Aircraft Co., the two main cost could competition in the past year. Boeing shows that it becomes increasingly difficult for one to get an order when gains in customer morale attributable to better delivery are only in the opinion of the customer. When one airline buys an aircraft which comes along the same route we are compelled to bid and wait until delivery at the earliest possible date.

Besides of the attraction and cost reduction of the jets, a line which fails to get early delivery can expect to sacrifice its burdens in passenger revenue lost.

First Deliveries

In this war, backlog and delivery dates in San Diego and Seattle have advanced rate by rate through the first round of orders and there are reported to be an additional number in the world who have not yet bought long range jets. Those are National Air Lines and Canadian Pacific Airlines. Douglas now has 118 orders. First deliveries will be made in the summer of 1959 before modification is complete to give customers opportunity for crew



JETLINER fuselage and wing are tested. DC-8 has lower cabin ceiling than other jets, requires less additional window size does not justify full side weight.

for Jet Race

training before plane goes into service and commercial service will be delayed to October, 1961.

Second round of orders for the big jetliner is not expected to be as large as in the first, but may determine whether or not Douglas will score a financial success for air racing. Opening date for the second round will be decided possibly by the availability of financing. First round has nearly fixed speed long flight intervals and time must be allowed for recuperation—even with the effects of the recent 5% fare increase allowed by the Civil Aviation Board.

Douglas experts, and presumably those of other manufacturers, predict that there will be a fair race designer after three deliveries in the second round since it will cost an airline less to wait once it has its first batch of jets operating.

Short Margin

Plane enough for the airline dollar less left surface load to trim profit margin per unit at the hope of raising it up to volume sold. Training costs is a more difficult proposition. Douglas commercial transports cause property to be disclosed in mass produced because of the number of causes variations of lowest. That should not be a big selling point because it allows a lower to seek a speed advantage in the competition but, according to Douglas chief engineer Ed Barnes, it also is the major cost problem yet faced in the DC-8. He reports that the first 10 airplanes



DC-8 RECENT passenger seating arrangement shown here has been expanded by most European airline operators. What this configuration features passenger lounge (shown).

will include domestic and oversea models, three different powerplants, trim ratios and different materials on aircrafts. The many one-shot experiments, plus trim, take-off and multiple administrative chores. How ever, Douglas experts believe that the sale appeal of certain equipment make them counterbalance cost factors.

Curtiss approach is especially valuable and expensive inroads in the sale of cabin interior designs. Cabin is the showcase of the airline and the main place where the operator can demon-

strate to the passenger that his DC-8 is better or at least different than his competitor's DC-8.

Besides of this, airline operators have been willing to pay large sums to get certain interiors.

Douglas first began to consider jet transport design in about 1945 but concluded that cutting engines lacked the power and low speeds did not compete needed to be economically practical. In 1951, when the Pratt & Whitney JT-3 (distantly 1951) appeared on the horizon, Douglas and Boeing



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4. Greater efficiency and improved system performance without the use of additional electronic components.
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began work on their respective jet transports.

Without military support for development there was little hope of halting competition until the air and Douglas began to figure ways of finding enough air traffic to keep the planes busy. To meet current traffic and dimension standards, while still giving early delivery.

One of the first and most important steps was the decision to dispose with a prototype and design a production airplane immediately. Using the first wire off the production line in an accelerated flight test program. This procedure is not uncommon in the early stages of civil transports and allows engineers to study alternatives before the final design frozen. Douglas engineers decided that fundamental differences between a lead safety wire and a production airplane were not to be considerable and that one initial operation of a prototype aircraft could lead off to future racing-money checkups, would prove little about its operation in the hands of actual users in the field.

ATA Formula Used

Estimated factors of DC-8 spectrum are compared on the basis of the 1955 Air Transport Area formula for fatigue testing. The new test set up in strict accordance with each standard for aircraft production is known but Formula is volatile in a variable for cost purposes. Production of DC-8's has gone up 150 million yards since yet goes over three times that of the DC-7C. Average load factor in the U.S. has been a constant 0.75 for many years.

DC-8 is being offered in discrete and intermediate versions, but decreases only in the two identical. Intermediate version has been given intermediate

DC-8 Specifications

Airplane weight empty (manufacture's)	122,573 lb.
Operator's static weight	7,941 lb.
Airplane weight empty (operating)	130,564 lb.
Per load	35,688 lb.
Zero fuel weight	146,444 lb.
Fuel	321,076 lb.
Minimum landing weight	194,083 lb.
Maximum takeoff gross weight	267,560 lb.
Maximum cruise speed (IAS)	313 ft.
Maximum range	5,856 miles
Flaps down speeds, up to	
-at 20 deg.	250 ft.
-at 50 deg.	149 ft.
Gear down speed-lessor of	
250 ft. or Mach .946	

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JETLINER is prepared for rollout to find assembly problems. Target date for first flight is May 15, but Douglas officials say it may be delayed (AW April 24, p. 40).

about 35,000 lb of fuel and by increasing strength of wing and landing gear. Maximum gross weight increases from 265,000 lb for the domestic airplane to 297,500 lb for the overseas version. Some of the overseas aircraft will be able to take off eastward Southeast-Tokyo soon in the event adverse jet stream winds

Missionary loading weight for overwater sections is 194,000 lb and that of the domestic airplane is 198,000 lb. Because of this, the crew difficult to meet the operational requirements of the two or six-seat flight length.

Passenger Preferences

Douglas intends the DC-8 to operate in point-to-point patterns and in pairs without preferential treatment, though market between cities may prove to be too small at jet transition cruise speeds. In mixed propeller and jet traffic, the DC-8 will be able to hold in a standard staircase pattern 9.5 miles on long at 5,000 ft with an eight-minute cruise rate using a half-rate of 3.5 miles over. Speed would be 530 or 540 mph and total speed would be 1,550 mph. Fuel economy would be substantially improved by holding at higher altitudes. An intercontinental community pattern would be 15.2 mi long at 5500 ft with an eight-minute cruise time and turn time from one end to the other. Turn radius would be 12.5 mi. Standard altitude of three deg/sec cannot be used because the high angle of bank needed would cause passengers discomfort.

Bearing down traffic speeds for the DC-8 would also be slow in flow, as DC-9. Full approach speed would be 80 sec of dip at maximum landing height of 152 ft at 12.5 deg. At maximum below minimum landing weight, speeds are normally lower. At 160,000 lb, the DC-8 will touch down at 152 ft.

DC-8 will make a 150 deg turn 600 feet on a runway 85 ft wide. A clear distance 188 ft, no distance is needed for wing tip clearance in the turn. Wingspan is 149 ft. Short turning radius depends

four-wheeled bogie. Landing gear is made possible by removing off legs because of overhang ground clearance of 10 ft. At part of what is called a "bogeon" where the two wheels are at the center of the turn and when the nose wheel steering angle is greater than 45 deg. Arc traveled by the outside wheels is usually great enough to scrub the free side-wings. Maximum nosewheel steering angle is 71 deg.

Metric considerations were given to ease of maintaining transoceanic time. Studies indicate that jet service can be completed in 10 days. Fueling requires only 10 min with pressurized fueling.

The extra dozen enable passengers to depart in an emergency.

To prevent runway delays from being pulled onto the engines during takeoff, Douglas engineers have invented an "overrotation" series called a "blowover." Studies showed that compressor suction creates a suction vortex between intake lip and ground. Low pressure center of vortex is enough to influence air ducts along the ramp surface and debris travels through this to the compressor. Low pressure blowover generates an air vortex around the intake. Natural airflows. Computerized flight management and Douglas website software will be effective against debris and blown up by engines of other aircraft.

Passenger Seating

DC-8 will carry 115 passengers with a 60-second, fast-eject seating arrangement, 152 with raised dualaisles and forward access, 140 seats, and 176 in the new economy class configuration for some European operators. It is a non-identical arrangement with closer than 30-in. pitch.

A total of 15,000 lb of cargo and baggage can be carried in pressurized compartments forward and aft. An aerial compartment in the forward cargo space is pressurized with valves said to be "adequate for the interpretation of fair leg room."

Margin of time gained by late free-

LIN BRASCOPE
ball/disc integrator
For all needs involving:
- direct ratio integration
- differential analysis



In the photograph parts of the ball/disc integrator system is assembled to measure the position of the aircraft.

In the photograph, aircrafts are shown. The first aircraft is a Douglas DC-8, the second is a Convair CV-1000 and the third is a Douglas DC-10. The aircrafts are shown in flight, illustrating the use of the device for aircraft monitoring.

The LIN BRASCOPE ball-disc integrator can be used as a mechanical integrating element in various aircraft systems. It is useful in aircraft navigation, attitude monitoring, wing computing, flow rate monitoring and wind sensing systems, aircraft control and aircraft load monitoring.

The device is manufactured under the license of the company

DAVATRONIC INC. (USA) and the

DAVATRONIC LTD. (UK).

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ing of the design was used by Douglas aerodynamicists to find what this assigned as the best high subsonic wing not developed for transport aircraft. In basic design was not made final until 1955. By the completion, the DC-8's most cruise at about Mach .45. Rapid and most conventional wing design for that time. Mach. number would be characteristic to a 30 deg. sweep and swept wings leading edge curvature, but these characteristics provide low speed performance and structural strength.

With the speed, Douglas design, which is shorter and has less area, 30 deg. at the quarter chord line. It originated from a series of tests in a transonic wind tunnel program. Cost for all DC-8 wind tunnel tests not counts to \$7 million and is continuing to grow. Wings and stability semispan were tested independently from the rest of the airplane on some runs to get the benefit of higher Reynolds numbers.

Wing Evolution

Analyzed method used a set of theoretical supersonic vertices to represent the wing in its function of air trailing the flow of air to provide lift. Characteristics of the vertices were used to make a rounded flow pattern and were then translated into a wing shape capable of developing the flow around pressure and velocity distributions. This preliminary wing was then modified as small was to find the best compromise between skin pressure and velocity patterns and the many other lesser factors which affect the development of a wing design. Douglas aerodynamicists believe their approach is unique in the industry, but Howard Ladd, a Douglas assistant chief engineer, says, "The airplane in part of could do it better or even do it not with a single headed education." In Douglas.

Fairfan's problem for the aerodynamicists was to expand the difference between leading speed and cause-speed DC-8's initial cruise is 15% greater than that of the DC-7 while its final approach speed is only 14% greater than that of the earlier airplane. That was made possible by the reduction of three factors:

- Increased camber than stationary.
- New arrangement of the airfoil slugs between leading edge shape and leading edge use at high subsonic Mach number included in the design.
- Much-improved transonic number airflow.

Advantage of swept wings lies in creating a static pressure component of flow over them. Mach number of component perpendicular to sweep angle is made less than free stream value. Direction and velocity of flow is controlled by pressure distribution on

the wing. Losses of equal pressure should evenly profile the span if the swept wings are perfectly adapted to them. In practice, swept wings have a much higher effective camber near the root this can be explained by the geometry of the airfoil profile causing loss of equal pressure and velocity to shear off at their appointed positions on the center section of the wing. Aeronodynamically, the wing has less sweep than it does structurally. Effect spreads outwards as speed increases and is reinforced by the presence of the leading edge which also generates flow and its consequent effect on camber near the wing root. From the flow differences with speed, the wing retains all the disadvantages of the full transonic sweep back at slow flight.

To combat this, Douglas aerodynamicists reduced the physical camber of airfoil surface while increasing the camber of the onboard sensors. Physical camber airfoil must actually be negative to get effective camber down to the desired value. Tests show that the shape chosen holds the pressure velocity pattern in the proper orientation. This has allowed Douglas to use a wing with only 30 deg. of sweep for transonic and supersonic flight, while maintaining running at the high end of the speed scale.

Bright considered airfoil sections here being leading edges and over high-machined by camber to ensure that they did not hit at one high angle of attack. Trailing sections had relatively sharp leading edges and less airfoils lift coefficients to make them stall at lower angles and earlier than the tips. The difference in airfoil profile at stall and tip is represented by $\Delta \alpha_{stall}$ in the angle of attack at the tip a 1.5 times larger than the angle of the swept back planform. These features guarantee good pitch down at the wing begins to stall. This also is protective against reversible symmetry of the stall by giving the controlled wing tips and allows a mechanical advantage over the stalled wing root.

Aileron Control

Spoilers are not used differently for layered control on the DC-8 despite the fact that Douglas has experience with that without in earlier designs. Douglas engineers feel that ailerons give adequate roll rate for a transport at all speeds.

According to aerobatic belief the first wings of the DC-8 would bring on the leading edge due due to extra gusts effect at a relatively low Mach number. At the time the DC-8 began to manufacture, most aerodynamicists held that drag due to control at a Mach number clearly below critical Mach number at which a shock wave begins to form somewhere on the

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High power capability in a compact package is a necessity in practical as boom and in-space applications. Ceramic tubes are ideal for this service. The stacked ceramic 4CK100A shown above — conservatively rated at one thousand watts plate dissipation — is less than 5 inches high and 2½ inches in diameter. Compare it with the conventional glass tube or the same plate dissipation device.

The exceptional chemical and mechanical stability of ceramic makes of high temperatures makes the compact, powerful tube possible. The higher temperature ratings of Eimac ceramic tubes make heat transfer more efficient and reduce cooling air requirements.

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wing. Because of this it has been necessary to design thin skin wings to minimize local accelerations of flow preventing it from reaching Mach 1 at any point while most of the wing is subsonic.

Douglas learned from NACA that the conversion between drag rise and critical Mach number is not as slow as had been reported. NACA transonic amplitude work done at Germany during World War II, showed that critical Mach number can be exceeded with out recontacting the drag rise until the shock wave moves off the end base (this place is the upper surface of the wing at a shock wave reflects to bring it to the bottom of the flight).

As long as the shock is forward of the trailing edge it is avoiding any ratio of a region of compressing flow and maintaining high subsonic pressure. Then suddenly pressure drops across the shock wave (pressure drop). When the shock passes the trailing edge it begins to release subsonic pressure, expanding flow and pressure drag rises sharply. Also, the shock wave is shorter or the density of air is compressing flow and results in a narrower included angle.

Strength and Storage

Application of the end base theory has enabled Douglas to use the high maximum SRI coefficient outer sections and thick wings for greater strength and reduced storage space. From nose to tip, thickness ratios of the three aircraft models used are 12.5, 30.0 and 10.0.

To prevent losses due to interference in wing flow of the engine installation a combined pylon was adopted which is strengthened with the addition of a portion of an fairings over the wing. Straight pylons were found to increase drag at speeds up to Mach 1. At supersonic speeds, the drag of straight pylons amounts to about 20% of the total parasite drag on the airplane. With the combined pylon there is no appreciable drag increase even at Mach number exceeds that of cruise—about Mach 0.7.

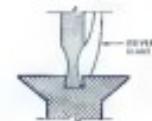
The 5½ chord dips on the DC-8 are an improved version of the double-delta type used on the DC-6 and DC-7 series and give us much additional lift at Fowlis dips with less pitching moment. Douglas calls this the most powerful completely aerodynamic trailing edge lift lift down to us in 20 years. A special bonded linkage permits greater chord extension as the tailplane position to give higher moments of resistance. A roll moment of 20 degrees is required against the 25 deg. roll moment to clear the 21 deg. at the angle greater than the 25 deg. lateral setting. Use of the roll moment, rather than a simple cut-out, increases



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Write for complete specifications
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THE LOCKHEED F-104A, the fastest and
lightest flying fighter ever ordered for
manned operation by the U.S. Air Force,
became operational for the first time on
February 18, 1958 at Muroc A. F. B.
after completing one of the most
comprehensive proving programs ever
exercised in a military airplane.

Along with many outstanding features, the
Starfighter's "magician's dozen" concept
provides far easy maintenance. For
example, Hi-Torque bolts from second
access add-on-not-passthru. The Hi-Torque
was selected because of its unique shallow
recess that locks the driver into it when a high torque load is applied during
installation or removal. This eliminates
the chance of driver slippage which can
damage the roots of a conventional bolt
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For power by-Hi-Shear laminar engines
for strength, elevated temperature and
sealing applications, the Hi-Torque is a
National Aircraft Standard and is readily
available in alloy steel, stainless steel and
stainless materials.

THE LOCKHEED F-104A, THE FASTEST AND
LIGHTEST FIGHTER EVER ORDERED FOR
MANNELED OPERATION BY THE U.S. AIR FORCE



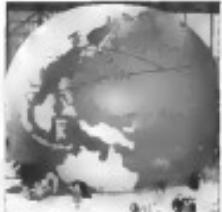
11:05 a.m., March 6, 1958. Test inflation of Pan Am's giant globe begins at New York's (Idlewild) Airport.



11:13 a.m., The Pan Am Model begins to show signs of flattening caused by over 200 visitors of the press.



11:25 a.m. Beauty of air blowers goes away as the 100,000-cubic-foot structure now measures less than half-filled.



11:30 a.m. Completely deflated, the World map is reduced. Rate deflated, was stopped by Clipper to Brussels the same day.



How to walk around the world at the Brussels World's Fair

From now to October 19, America's colors are flying at the Brussels World's Fair. And after more proudly

Upwards of forty million people from all over the world are expected to come together in an atmosphere of peaceful interchange. This is a powerful step toward improved understanding among nations.

Pan American—the first "round-the-world airline"—believes there is no better way to do this.

Indeed, bringing the peoples of the world together, *hostes à hostes*, has been Pan Am's main theme in over 30 years of overseas flying.

One million people have crossed this bridge to understanding on the wings of the Flying Clippers®.

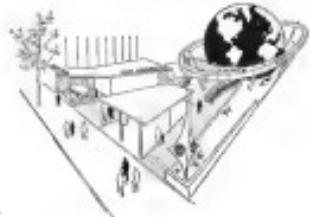
Now Pan Am has joined with 30 nations, seven intergovernmental agencies, public and private organizations of six continents to create the first World's Fair of the Americas. At Brussels, the Pan American World is represented by a gigantic, inflated globe. Built in the Irving Air Chute Co., this video globe, 57 ft. in diameter, is the biggest earth on earth! Continental land masses and islands of the world are painted on the outside along with the names of 143 of the cities served by Pan Am.

Inside, 160 visitors at a time will be able to watch a demonstration of aerial effects in the "theater of the stars," worked out with the cooperation of the

Hopdes Planetaryarium and Sporn Laboratories. A short film will also be shown dramatizing the change from Magellan's "round-the-world cruise 400 years ago to the Jet Clippers of the first "round-the-world airline."

After the show, the audience will exit through a door that leads to the "walk around the world" which sweeps out nearly 30 feet to ring the globe, affording an excellent view of the Pan American World and the surrounding fair grounds.

This first "round-the-world airline" is for the in-spirited nation.



PAN AMERICAN

VIBRATION

This General Controls 3-way selector valve is undergoing a vibration cycle of 8 minutes duration, 75 to 500 cps, at gravitational loads in the order of 10 g's. Test specifications required fluid pressures of 1200 psi across the parts, with three port capped.

Purpose of test was to determine leakage under vibrational stresses comparable to those encountered in the fuel lines to jet engine afterburners.

Qualification report by an independent testing laboratory stated that internal and external leakage after vibration cycle was zero.

Every product is the complete General Controls 3-way, ranging from gate valves to pressure switches, undergoes an exhaustive program of tests before production release.

Another reason why the General Controls 3-way trademark has become a recognized standard of quality in the field of aircraft controls.

Why not talk to the man from General Controls about your next aircraft valve requirement?



GENERAL CONTROLS
AMERICAN CONTROLS DIVISION
Glendale, California • Spokane, Washington
20 plants • 47 sales offices
through the United States and Canada

picked for earliest long range, open form. The flap is driven by multiple actuators to cut torque load and save structural weight. All flap segments are linked together by a combination of mechanical and hydraulic means to prevent simultaneous extension.

Up to now, the landing gear supports with a single 14-in. dia. tire have reflected through an angle of 60 deg. when the associated tandem down and compression of the shock strut plus shear is started. Their only purpose is to dismiss life and gas most of the equipment's weight onto the wheels for effective braking. No good braking will be installed unless it is expected that thought inversion will be used to decelerate in flight.

Control Characteristics

Douglas claims that stability and control characteristics of the DC-8 are as good as those of earlier Douglas transports.

Engineers chose to use elevons alone for lateral control area that is better than spanlets at very low speeds if all characteristics are as good as those designed into the DC-8. Spanlets except may notwithstanding at high speeds but this problem will be dealt with by the use of DC-8 wing leading edge slats. It would be wise to add that the slat system design is a departure from previous designs in that it is a separate, flat slat rather than one that represents a hinged segment that is hydraulically powered and driven by an onboard unit through a preselected torque limit. At low speeds, maximum hinge moments are less than the pre-load and the two pieces operate as one unit. As speed increases, hinge moment exceeds the pre-load at progressively decreasing deflections. Other segments will not deflect beyond the pre-set limit and the upper segments that are attached to the aileron control surface effectively limit the deflection of the wing. From trials to associate speeds, use of roll slats at speeds of least 25 deg. or chord twice that of the DC-7C and in successful temperatures.

When hydroelectric power fails at either altitude, that role automatically reverts to manual control operating through a cable system with an intermediate board to switch between the two control configurations. On manual control, maximum effectiveness and control forces are approximately equal to those of the DC-7.

Cleist problems in lateral stability is "Dutch roll" manifested in yaw and roll motion caused by rough air due to poor aerodynamics damping of aircraft at high speeds and high altitudes. Large airplanes have a high long period of oscillation—about 4-7 sec. for the DC-8—but it is still undesirable. Design engineers believe the revised DC-8 Dutch roll condition has an improvement again, as they have satisfied a simple

yaw damper based upon a very sensitive accelerometer located in the tail. Conclusion that this was needed was based upon flight tests of fighter designs with Dutch roll periods under 1 sec. Since there is a question of the applicability of the theory, if it is possible, the new design will be taken out when flight tests are completed.

Aerodynamic philosophy of the DC-8 regards tail surface design as more critical than wing design at high Mach numbers. Correspondingly affects the wing are associated with a drag rise which tends to bleed the airplane out of

of dangerous speed regimes, whereas the same effects on the tail surfaces do little to limit speed and can cause loss of control.

To prevent that, the tail of the DC-8 has been designed to be effective at speeds up to 250 mph and the capability of the tail to do its job is possible, the new design will be taken out when flight tests are completed.

Interrelationship of fuselage shape and tail surface location has been extremely concerned to put the surface in

Douglas DC-8 Basic Data

AREAS, DIMENSIONS

	Wing
Span	197.9 ft.
Area	3,758.2 sq. ft.
Mean Aerodynamic Chord (MAC)	20.6 ft.
Aspect ratio	7.1
Taper ratio	0.244
Deflection (at trailing edge)	±5 deg.
Sweepback (25% chord line)	30 deg.
Affine span (root chord)	24 ft.
Affine chord (5% wing chord)	24.0
Fly span (root chord)	66.7 m
Fly chord (5% wing chord)	39.0
Fly span	496.9 sq. ft.

Fuselage

	Maximum cross section (exterior)
Height	162.5 in.
Width	147 in.
Length	160.3 ft.
Static ground angle (base down)	1 deg.
Tail (horizontal stabilizer)	

	Tail (horizontal stabilizer)
Span	47 ft. 6 in.
Area	559.1 sq. ft.
Aspect ratio	4
Taper ratio	0.259
Sweepback (25% chord line)	35 deg.
Inclination (vertical)	-10 deg. to +12 deg.
Deflection	±5 deg.
Location of elevon hinge line (5% chord)	75%
Distance from 25% wing MAC to 25% horizontal tail MAC	82.4 m
Tail (vertical stabilizer)	

	Tail (vertical stabilizer)
Span	25 ft. 11 in.
Area	336.7 sq. ft.
Aspect ratio	1.5
Taper ratio	0.269
Sweepback (5% chord line)	35 deg.
Location of rudder hinge line (5% chord)	65%
Distance from 25% wing MAC to 25% vertical tail MAC	79.8 m

	Landing Gear
Wheel area	250 in.
Wheel base	659.9 in.
Vertical travel of tire from extended to fully compressed (at positive—one wheel	16.5 in.
—one wheel)	16 in.

	Clearance Dimension
Height over tail	42.3 ft.
Length (maximum)	158.5 ft.
Ground angle	15.2 deg.



WHAT IS "TOTAL ELECTRONICS"?

The picture suggests the answer.

In the new world of missiles and space systems to come, it's the total complex of control, guidance and communication—the whole interrelated nervous system controlling the eye, the hand, the head and the heart of the missile is that of man himself.

And in the company producing that missile, it's the total electronics capability necessary to specify, design, create and test the central nervous system as an integral part of the whole machine—from its conception, through delivery to the customer, to the final completion of its mission.

In the period of a dozen years since the word "electronics" first gained common currency in our industry, Martin has been systematically building toward just such a total electronics capability.

As a result of the rapid evolution in advanced electronics development, today one-third of all Martin engineering manpower is devoted to the electronics requirements of our customers' present and future products. And a major part of Martin's investment is in the special facilities necessary to this new concept of total electronics.

We believe that the capability is essential to our increasingly important function as a prime contractor to all branches of the military.

MARTIN
BALTIMORE-DENVER-ORLANDO



basis and resembles the pathing moment curve of straight wing designs more than that of the swept wing booster.

Most DC-3s now in order will be powered by the Pratt & Whitney JT4 (JT5) and some avionics craft will be powered by the Rolls-Royce Centaur. First airplane will be with the Pratt & Whitney JT4 (JT5). Second will be the JT4 (JT5). Third, JT4-powered DC-3 will be delivered to Pan American Airlines by December, 1958, and the first with Centaur will be delivered in February, 1960.

Powertech utilization has rated equal attention in this first generation of commercial jet aircraft. Engines have always been designed for military applications and have contributed to civilian jet, but this is the first time civilian operators have ever had to cope with principles of aerospace aerodynamics different from those of fighters and bombers. Now, for example, care must be taken to insulate areas from the heat that is generated by the engine, as engineers go along on such less than transfer attention than in military service.

Central cycles may be as much as five times as long. Military aircraft assume much rougher usage than commercial aircraft do. The safe record of circulation in the past can be stretched largely to the success of aerodynamic design, protecting against catastrophic failure due to thermal stresses.

Most DC-3s will be built in two-wheel designs, based on incident and reflected laser beams, government and military sources should consider using various powerplants in service.

Reliability Factor

In the engine and in all of the airplanes around the DC-3 has been so designed that a single failure seldom if ever will bring down the aircraft. Operation of each engine at minimum power of the maximum use other engines or subsystems continue to operate. For the sake of reliability, all valve controls are cable operated.

Three types of fire protection have been installed. Fuel may be shut off internally and sealed off from the engine, and a CO₂ extinguisher system is provided for each engine. Fuel and source of ignition such as a fuel leak or a short circuit can easily be ignited. No combustible exists in the hot section of the engine pod except for a relatively small film of lubricating oil. Firewalls are installed in the upper and lower ends of each engine pod. To prevent the spread of fire within the pod, there also is a firewall in the engine annulus at the front of the compressor section and a secondary fire seal between the turbine and compressor sections. All materials in the hot areas are fire proof. There is a bimetal-

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make it possible for us to complete jobs such as Hangar 45 for **McDONNELL AIRCRAFT** ... right on schedule"

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INTERNATIONAL STEEL COMPANY

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fire extinguisher fire extinguisher system in each wing and each aircraft has one. The two extinguisher systems weigh less than a single central system.

Pad shutoff is no more than a handle for the engine and the pressure tank has been replaced by a central tank. Douglas is interested from these three points: point, two forward and one aft. All forward and central tanks are taken out through these points to the pumps.

Brake differential between maximum inertial and decelerate remains of the DC-8 is the amount of fuel carried although each has an eight tank pressure system. Overdrive system has been taken out of auxiliary tank, which is replaced by the pressure tank and auxiliary. Douglas has determined that separate splitting of the pressure tank is necessary because the relative difficulty of rupturing tanks under fuel interruption is more serious than storage. Fuel mass and flow determine both of the pressure ratios constitute a separate system for each engine. Fuel can be taken off of the pilot during an emergency, but Douglas engineers feel it is undesirable because of possible fire hazard when liquid quantity of fuel is removed.

The tanks are interconnected through a pipe of about 400 pounds between the planes and near the leading edge on each wing. With present ground equipment, fuel can be put ashore at a rate of about 2,000 gpm, using oil less than 500 pounds per cu. ft. The system is capable of handling 1,400 gpm. High filling rate is intended to cut turn-around time. Dual fuel oil valves close automatically when tank is full. Vent assembly using the tanks, aggregate and linking to the overboard drain has been satisfactorily tested. Molded rubber bags are used and ensure that rate of response fast if one is available. If both the intertank valves are open fuel runs through the straining ports.

Fuel shutoff in static discharge lines during fueling has become a bigger problem with the advent of jets. Static charge is lost by friction between fuel and line and by sheathing within the tank. Douglas has determined to keep the static charge and the static will be reduced by more erosion tools. Douglas has found that by locating the filter strainer close to the bottom of the tank so that mixing fuel places as possible against gap exists in the fueling passes before the charge becomes dangerous.

Fuel is supplied to each engine by two fuel pumps, an engine-driven unit and pump in the nacelle and an electric pump for the fuel tank. Hot oil is heated from the constant speed drive for each magneto-driven electric motor through an oil cooler heat exchanger to service fuel. Fuel tank DC-8 is not cooled into the exchanger and fuel temperature drops as the

flowing point of water raises the possibility of ice formation in the filter.

Mobile starting equipment can be used on six of the four engines. When the first is running, the others then may be started by the use of the main gear selector system.

Douglas is offering a larger pressure and gravity contact air jet and hot air exhaust air. In this way, air is dispersed sooner and easier to the engine exit. Since exhaust is parallel downstream, forming a conical nozzle at contact with outside air, early dispersion of energy reduces the area on which sound is radiated.

Exhaust air velocity, which controls the exhaust jet, becomes a key of several studies because closed and the bypass exit strength which outside air is measured. This reduces exhaust velocity and increases its noise. Mass flow rate and thrust are affected slightly, but lower velocity, airfoil loss, shear and less energy dissipation is noise.

Each engine has its own lubricating system with a supply pump inserted in the swirl. This is divided into two sections, one for engine oil and the other for transmission, driven by the same pump. Engine oil is fed to the drive system from draining off engine oil and eliminates possibility of one solvent contaminating the other. Total capacities are 6.5 gal. for the Convair, 10.7 gal. for the DC-8 and 12 gal. for the DC-9. As an engine turns air flows through the oil cooler which is integral to running on the ground.

DC-8 emulsification. Diesel emulsion seems to improve gross fuel of 4,500 ppm

rate, fuel rate of nearly 6000 sugges-

ts design and its combinations which were required to move faster than the supersonic plane.

Douglas suggests consists of a "shoulder load" type of engine and a removable cowl. This low pressure air baffle air restricts a larger pressure and gravity contact air jet and hot air exhaust air. In this way, air is dispersed sooner and easier to the engine exit. Since exhaust is parallel downstream, forming a conical nozzle at contact with outside air, early dispersion of energy reduces the area on which sound is radiated.

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Since installed at a fixed level, system gains in the system, a set course that like a biplane or afterburner and produces a wide net increase in static thrust. The maximum thrust of the system is increased by 50% at a given altitude, and the specific fuel consumption is reduced by about 10% at a given altitude. At 15,000 ft., specific fuel consumption is decreased by about 10% due to the increased thrust. As specific resistance to increase, drag of the system rises more rapidly than thrust, which is reduced to overcome this resistance and it is reduced to overcome this.

Since the dual panel valving shortens the sound generation cycle, a good solution to the problem is to reduce the valve lift in the primary shutoff. Total travel reduction of the shutoff is 13.5 ft., of which 3 to 5 ft. are extended to the cockpit. With the dual panel valve alone, average intervals throughout the noise frequency band is lowered but at high frequencies it is increased. With the valve and dual panel controlled distribution of intensity with frequency has two peaks at opposite ends of the frequency band, but both are very small. The corresponding level change figure for the noise suppressed DC-8 is about 3 dB below the average produced by propeller-driven four engine transports.

Douglas believes it is conventional aircraft design producing intense thrust equal to 40% of forward thrust.

Trim change is of the same order in magnitude as extension of the flap. Its primary cause is not the pitch/yaw moment caused by the moment that is a change in the pattern of air flow over the wings due to circulation of the air around the flap.

(This is the last of two articles on the Douglas DC-8. The second part appears in next week's issue.)



Soviet Oxygen Mask

Russia has developed a sharp, plastic oxygen oxygen mask for passengers on Antonov's jet and turboprop transports. Mask which is discarded after continuous use, is used to cover off hypoxia requirements.

How Research Shapes Our Future Prosperity

If you are looking for an industry that is going to keep on boozing in 1958 and every year for the next decade, here it is. It is the industry of technological innovation through research and development.

Last year this great new industry spent over \$7 billion to discover and develop new industrial products, processes and equipment. This year the preliminary McGraw-Hill survey indicates that total expenditures for industrial research and development will be even greater, perhaps as much as \$8 billion. Of the companies surveyed, 57% plan to spend as much as in 1957 and 34% plan to spend more.

The sustained expansion in research and development is the best guarantee we have that the current decline in business investment in new plants and equipment will be relatively short-lived. There can be no prolonged decline in investment in an economy where technology is changing rapidly.

This editorial is designed to show how the continued surge in research and development can be expected to lead first to new products, and eventually to increased expansion of investment in new industrial plants and equipment. Such expansion is the essence of national economic growth.

A Slow Start

The impact of research on sales and investment is still very gradual. Research spending itself has more than doubled in the last four years. But only 32% of all manufacturing firms report significant capital outlays to make new products. We are not seeing the full dividends of industrial research as yet in several reasons:

- Research expenditures were relatively small until the Korean War of 1950 brought substantial government contracts in aviation, electronics and related fields. Heavy research outlays for civilian and industrial products came even later.

- There is an average lag, according to research directors consulted by the McGraw-Hill Department of Economics, of roughly seven years from the start of research until the product is ready for large scale output — about five years of research and at least two years to solve production problems and develop markets.

- Complex products, such as new consumer durables and industrial machinery, have an even longer time lag.

However, new developments are certainly underway. Research begins to increase in all lines of business when Korea War restrictions and

the excess profits tax came to an end in 1953. The tax revision of 1954 added a new incentive by making research outlays deductible as a current business expense. By 1955, the research boom was on.

When Is The Payoff?

With a lag of about seven years, it will be the early 1960s before these new developments become a dominant factor in capital investment. But once the flow of new products and new processes starts, it will accelerate sharply — just as research spending has accelerated in the past few years.

By 1960, over \$50 billion in sales will be coming from products not on the market as recently as 1956. Sales of new products will increase year by year, but they will gain most in 1960-1962, or five years after the recent spurt in research expenditures.

Capital expenditures to manufacture new products will also rise, but with a slightly longer lag. Here the sharpest rise should come in 1963-1965, as the new products reach a volume that calls for a significant amount of new capacity. In most cases, initial output of new products will come from existing capacity.

This timing of a new wave in capital investment appears logical on other grounds. Population experts forecast an upsurge in marriages and births around 1965. So by 1968, industry will be starting to tool up for new mass markets.

The important point is this: As we approach the 1960s more and more sales and investment will be in new products growing out of research. By 1960 well over 10% of manufacturing sales will be in new products not on the market in 1956.

Meanwhile — research will help stabilize capital spending by raising the level of maintenance and replacement expenditures. Of course, research does not eliminate all the ups and downs in the demand for capital goods, for there remain variations in the amount spent to expand capacity. But a high level of modernization, to cut costs and improve quality, does put a floor under any drop in investment.

What To Expect

During the next few years we can expect an increasing flow of new materials, new metallic alloys, new machinery — primarily those developments coming out of long-established research programs in the chemical and electrical industries. Industry will raise under use of specialized computers and automated equipment.

But the dramatic payoff on research comes even later. In the early 1960s the consumer goods industry will begin tooling up for their really new products — things so basically new they can change the way a family lives. Such items as plastic houses, paper apparel, turbine motors are under development right now. But it will take several years to get costs down and for populations and incomes to grow to the point where mass markets are created.

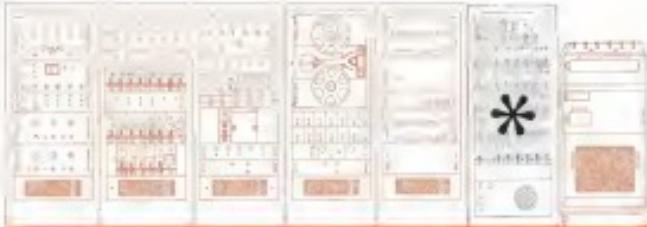
When we reach that point in the mid-1960s, there will begin the greatest surge of capital investment in all history. And then — around 1965 — the new processes (full automation, atomic power, continuous steel casting) which are the slowest and most expensive part of the research chain to develop, will come into play.

The combined impact of new products and new processes, to meet an expanding market, will then be felt in the mid-1960s — eight to ten years after the recent sharp increase in research spending. The full impact is that far away because of the lags for applied research, pilot plant studies and market introduction. But to a large degree the prosperity of the 1960s has already been shaped by the research programs now underway.

This message is one of a series prepared by the McGraw-Hill Department of Economics to help increase public knowledge and understanding of important nation-wide developments. Permission is freely extended to newspapers, groups or individuals to quote or reprise all or parts of the text.

Donald C. McElroy
Partner

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Designed around a concept entirely new to the electronic field, the Model 6140 phase-lock discriminator eliminates a great improvement in noise cancellation by filtering and stabilizing at low input-to-noise levels. In addition the

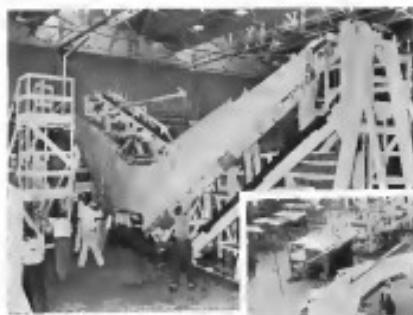
one exception for space applications all systems meet and exceeds in the specification of operational performance. For complete specification and information data, write Hallamore Electronics Co., Dept. R, P.O. 4530, Doubletree Avenue, Anaheim, Calif.



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**UTX Sabreliner
Wing, Fuselage
Joined Together**

North American Aviation's UTX Sabreliner prototype fuselage is reported to roll out at the end of May. Flight date is set for July 30, subject to delivery of General Electric jet turbines engines. Aircraft will be tested at Edwards, Calif. Power and flight will be derived from jet, learned to rapid support parts for mating with the fuselage.



Interior fuselage is being tested with wing. Fuselage is raised by jacks, moved into position over wing center section, then lowered onto wing for joining to center section. Nacelle (jet/turbine engine) are supported on horizontal plates extending from side fuselage side.

AVIONICS

Tests Predict Avionics Field Failures

By Philip J. Klein

New York— New component failure rate data obtained from more than one cellular brain of tests on over 50,000 avionic equipments now makes it possible for Collins Radio Co. to accurately predict failure of its equipment in the field.

Using component failure rate data gathered in 28 of 40 "action testing," Collins' estimated mean time between failures for two types of airborne equipment checked within 15% of the actual field life experience for two types of aircraft. The two types are the F-4 Phantom II and the B-57 Canberra reported during Institutes of Radio Engineers convention here.

• **High frequency communications transmitters**, used both by airlines and military, experienced a ratio rate between failure of 200 hr., extrapolated to predicted figure of 650 hr.

• **Complex military transmitters**, with predicted mean between failures of 61 hr., actually showed to be ten times fail rate in field use.

Accurate prediction of equipment reliability in the field part has been difficult because of wide discrepancies between published figures on component failure rates, often obtained from different sources under different conditions, Vinsler Hansen said. For instance, vacuum tube failure rates per 1,000 hr. we have been reported at between 75 and 27%, extremes between 0.04% and 1.5%, capacities between 0.08% and 2.4%, Vinsler Hansen said.

If designer uses most pessimistic figure for predicting mean life to failure he might come up with a figure of 1,000 hours, Vinsler Hansen said. For example, if obtained by using the most optimistic failure rate data previously available.

Notes of Caution

Vinsler Hansen cautions, however, that the Collins component failure rate data cannot be used indiscriminately. In other words, manufacturers become at possible differences between own parts design, production quality control procedures, assembly techniques and other variables. For example, Collins employs a minimum temperature data-taking factor of two, whereas some companies may use larger, smaller, or no factor.

With this sort of caution, Vinsler Hansen presented the following figures on component failure rates per 1,000 hr. in operation:

• **Vacuum tubes**, 3.4%

during which equipment might be operated continuously between 2 hr. and 40 hr.

Reliability tests consisted first of a 100-hour exposure to 2G vibration at 30 cps, followed by 20 hr. of proven operation which included mechanical cycling and testing to simulate actual field operation.

Test results, type failure rates, those occurring during normal factory testing, were not initially excluded from the Collins' component failure rate data summary; these results are needed out before equipment reaches the field.

In order to obtain data which reflect future failures in the field, Collins also

Component Part Failure Rates

(Derived From Equipment Life Reliability Test)

TYPE AND SUB-TYPE	Total Operating Hours x 10 ³	Total Failures	Failure Rate 60/365 Hours	
			Total	Per Hour
Capacitors				
Electrolytic, aluminum, Commercial	490	2	8.44	
Electrolytic MIL-C-55 MIL-C-7981	1,460	7	6.03	
Mica, Shaded, MIL-C-10912	45,000	12	0.027	
Mica, Shaded, MIL-C-7	3,400	8	9.15	
Mica, Flat, MIL-C-5	650	0	0.000	
Diodes, ITRK, MIL-C-11015	15,000	16	0.027	
Diodes, JAN-C-26	30,000	13	0.004	
Diodes, Variable, MIL-C-81	12,000	2	0.005	
Pipes, MIL-C-21	3,500	2	0.14	
Pipes, Metal, MIL-C-21	16,000	37	0.009	
Alt. Transformer, Transmitting	170	0	0.000	
Alt. Transformer, JAN-C-97	3,100	1	0.047	
Total (Capacitors)	160,540	100	0.006	
Resistors				
Wirewound, Fixed, Low-Power, JAN R-104	850	0	0.000	
Wirewound, Fixed, High-Power, JAN R-26	1,800	1	0.007	
Wirewound, Fixed, Precision, MIL-R-93	560	0	0.000	
Composition Variable, MIL-R-94	1,100	5	0.15	
Composition, Fixed, MIL-R-11	91,000	4	0.000	
Film, Fixed, MIL-R-1649	2,900	0	0.000	
Total (Resistors)	116,240	10	0.000	
Vacuum Tubes				
Transistor, Junction, Glass	5,000	20	0.4	
Switches				
Metal, Diaphragm	2,680	60	2.2	
Conductors				
K. T. Cells	18,400	7	0.07	
Diodes	3,200	60	0.04	
Rectifiers	7,800	51	0.06	



LMEC flight-controls the McDonnell F3H-2N Demon



LMEC-designed 63H flight control systems sense the effects of outside forces... contact them with effortless, better-than-human speed. Assured of stability, relieved of basic flight control problems, the pilot is free to make his own essential contribution to mission effectiveness. • A LMEC-designed 63H flight control system has already logged several thousand hours of operational flying in the McDonnell Demon. For information on how LMEC's basic family of integrated flight control components can be adapted to your aircraft or missile applications, write for booklet Dept. 34.

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Aeronautics between these two speed tests and the first test flight, test date, Vendor Name etc.

Cutter is basically continuing his acrobatic testing program and plans to calculate all data in another year. An other check on the validity of the data should be possible late this year or early in 1959. Under suggestion of Astronautical Radio Inc. (Astron) program is underway to compare the computer data taken on a number of isolator AN/ARC-57 ultra-high frequency transistors operated both in long term tests and in field use under actual field conditions to obtain reliable information.

FUNDINGS, AIRLINES, FBO's, PILOTS:

Where Were You When The Source Went Out?



DARK IS DANGEROUS IN SUPPLY EMERGENCIES

Whatever occurs in a company needs, be it in the dark, happens very speedily. If he has his own source of supply and storage, he'll be prepared to knock it out of commission.

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New as a distributor of American Power Systems' compactors and parts, Southwest Airmotive Co., Dallas, Texas, has added a second source of supply. With an office, a fixed base operation, as a third source, located in the manufacturing as a single source, it can be surely in the dark supply wise in time of strike, disaster, or

stampede at the factory, or bottleneck, along the way to transit. Southwest Airmotive has been using the unit successfully for several years, and is now a necessary part of operations to stand ready for instant delivery whenever needed.

The second source, purchased in, we think, the most valuable service is delivered, can perform, another reason for its success, and ever increasing value to the industry.

Whether manufacturer or user, isn't a question you should answer. Where will YOU be when the source goes out? For vital second sources have contact Marketing Manager, Distribution Division, Southwest Airmotive Co., Love Field, Dallas, Texas.

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- Small Wonder—Aerospace digital computers being developed by Litton Industries for use in Navy airborne early warning (AEW) aircraft will use 40,000 diodes and 10,000 transistors which partially explains rising costs of semiconductor industry.
- Good News—Improved funds for basic research and development program, totaling \$10 million, perhaps as much as \$10 million over present levels, will be authorized

by Defense Department shortly. Most is intended to finance component reliability. Funds will be spent by three military services, supplementing their existing programs.

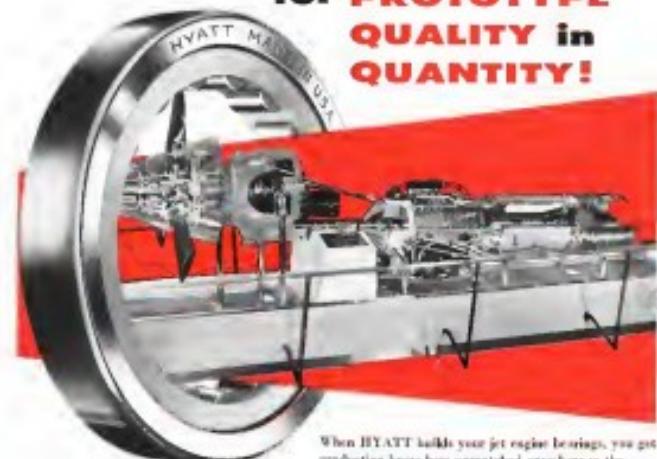
► Doppler Navigator—Collins Radio Co. expects to flight test first model of its transversal Doppler ground-speed wind drift indicator late this year, hopes to have prototypes of the 65 lb. unit available for airline evaluation early in 1959. Production models are slated to be available later in 1959. Collins will use PAF-CW system, employing Jones filter, and will have a range of 10 miles. Overall accuracy (including random errors) is expected to be within 1 to ±30% for ground speed, ±1 deg for drift angle.

► No AIA Decision on DME—Airways are holding back on an official verdict, decisions to make flexible purchases of Transoceanic DME pending results of airline evaluations of DME T and of Doppler radio navigation which seems obvious before will eliminate need for DME. United Airlines and Northwest are among the airlines expected to participate in the evaluation.

► Rite Muster—East Coast Conference of Aerospace and Navigation Electronics, to be held Oct. 27-28 in Baltimore, will award two prizes, \$100 and \$50, for the two best technical papers submitted. To be eligible, paper must be received by Sept. 1 with 100-word abstracts with brief professional record of the author by May 15. Write to W. A. Staggs, Technical Program Committee, Aerospace Assessments, Inc., Cockeysville, Md.

► "Poor Man's" DME—New type of distance measuring equipment which requires only small addition to existing VOR or VHF communications receiver to give pilot an distance from a minimally equipped ground VOR or VHF station has been developed by Bendix Radar Division of Belden Aviation Corp. Accuracy is quoted as within one to ten percent of 10 miles. The long distance adapter (adaptor) provides an 800 cycle/sec modulation of omnibeam transmitter which is translated for one-quarter second, causing ground station to reply with similar time modulation with identical phase to receive signal. Airborne unit compares phase of time modulated and received signals to compute distance to ground station. Physical air borne DME adaptor weighs 5 lbs., mass uses 7 x 6 x 10 in. but this could be reduced to 10% in production, Bendix estimates. Company is not pushing technique for civil use as alternative to Vortac DME, but plans to demonstrate equipment to interested civil and military agencies.

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EQUIPMENT

Aeroquip Makes Titan Hose Joints, Nuclear-Powered Aircraft Valves

Jackson, Mich.—Aeroquip Corp. is upgrading and diversifying new products and techniques to keep pace with developments in the fields of jet aircraft, building materials and mobile powerplants.

While still devoting considerable time and energy to refining its single production-line use of vacuum and high pressure, high temperature flexible aircraft hose, associated fittings and couplings—Aeroquip and its divisions are embarking on several new ventures some of which are initially unrelated to flexible aircraft hose.

Product Developments

Latest product developments at Aeroquip and its divisions include:

* For jet engines precision hollow-wall and solid sections which are formed, welded and inspected in a fully automated machine which holds and controls section blades at constant stress. Marmon Division of the West Coast Company has incorporated these able to make blades with twisted section and hyperboloid walls by passing them through additional processes.

* For ballistic missiles 6,000 psi operating, 24,000 psi burst pressure flexible hose and flexible fittings for mounting pneumatic systems on missiles. (Previous maximum operating pressure for flexible hose was in the 1,000 psi range). Marmon-made Con-

sod walls temperature range, high pressure (up to 6,000 psi) aluminum tubing goods are used on many missiles and entirely on the Minuteman missile, according to the manufacturer.

* For nuclear-powered aircraft, vacuum check valves, 16 in. in diameter weighing 575 lb. and 400 lb. respectively, have been produced for General Electric's Atomic Nuclear Propulsion Division.

These are rugged valves used at the Marmon plant to check valves on various types of nuclear aircraft and jet engines. Valves incorporate jackets which weigh 90 lb. and when mounted in a dropoff specially designed, thick asbestos seal on 16 in. diameter, metal end caps to withstand test. Each valve will handle a flow of 7,700 lb. of air per minute and will withstand a pressure of 300 psig.

Aeroquip says that the Marmon check valves will be installed initially in an atomic nuclear aircraft in a program now being carried out by the Air Force. When the check valves have successfully passed their operating tests, this will be used in an atomic engine designed to power a nuclear aircraft. Consider also are going into General Electric's space engine.

Aeroquip's Marmon Division developed a new stainless forming and welding technique to manufacture pressure flexible hose and other assemblies for gas turbine engines. To date,



PUMP PRIMERS

by
Arthur A. Nichols

High efficiency liquid coolant pumps for electronic equipment

For electronic equipment operation at high altitudes, air cooling of high-current semiconductors becomes troublesome because low air density limits cooling. An alternative method of heat removal is to use liquid coolants frequently preferred for this type of service due to extensive experience over the years in the production of high performance centrifugal pumps which have been of great value in the development of aircraft. One liquid that provides maximum heat load and spaces savings with efficient heat transfer capabilities is:



Fig. 1. Marmon stainless outlet pump. It has a straight and bell mouth throat of pump especially well suited to turbulent response, nuclear service for both airborne and ground installations (Fig. 1).

Our ability to tailor pump configurations to meet specific performances and requirements is due to our unique arrangement of the electrode segments fixed with a precision fit weight and space minimization. While this is of definite benefit in nearly all aircraft and missile applications, it is particularly important to electronic equipment where the most compact arrangement is made in the smallest spaces. The availability of Marmon pump design, unusual design adaptability, plus inherent flexibility and compactness in construction makes them particularly attractive.



FRECTION idle valve was first fabricated by a fully automated machine at Aeroquip Corp.'s Marmon Division on the West Coast. Valve for the Continental 3600/3610 turbine engine.

This feature of pump design not only means maximum simplicity in piping but permits further simplification of system design by incorporation of related valves and other system accessories in one compact unit. Marmon can supply pump units adaptable to suit drive requirements in accordance with the method of coupling best suited to the application. We can also supply complete motor driven pumps with integral combination of pump and motor.

Technical data is available and your application makes them particularly attractive.

W. H. NICHOLS CO.
88 World Ave., Waltham 34, Mass.

more 100 blades have been furnished to Continental Aeronautics & Engineering Corp. for trials at the rates of the company's 200-T-5 and 100-T-10 powerplants.

Continental research officials told AVIATION WEEK that "our interest in the Matrix is high, although our experience with the units to date is not sufficient to make a final evaluation."

The 100-T long nosecone cost \$298,800, requires a single operator

plus a helper who funds to stand and performs other operations as required.

Shells or coil stock is automatically fed into the machine which forms, welds and imports continuous lengths of permeable media by passing the metal through 16 smaller stations. Vane surfaces are cleaned to clean bores thoroughly during the permeabilizing process to hold number tolerances to .007 in.

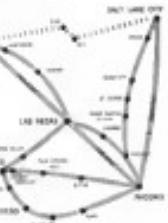
Amonging are that a new welding method of development which is used in the machine produces a more homogeneous weld sample 1 in. wide to make the hollow core as structurally sound as a drawn tube.



FIVE-LITER Liquid oxygen converter can supply 24 hr of patient oxygen for one carrier member. Converter, manufactured by Leslie Co., Division of Union Carbide Corp., can deliver up to 150 liters of medical oxygen per min. Weightlight unit weighs 9 lb. Vacuum of the oxygen canister is checked by manometer. Container consists of two insulated shells with annular space under positive vacuum

tubes of .75 in. The converter delivers up to 190 liters of oxygen per min.

Containing two concentric shells with the annular space under positive vacuum, the unit meets specifications of MIL-C-26572 (USAF) requirements at ambient temperatures of -65 to +260°.



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NEW AVIATION PRODUCTS

Acceleration Switch

Low cost unipolar single throw acceleration switch with linear differential sensitivity in a single plane and adjustable acceleration setting now is in production. When preset acceleration level is reached in the plane of travel, the switch trips and holds in a closed position regardless of subsequent

Correlation Counter

Mechanical counter for instrumentation systems provides a simple method of correlating events recorded by two or more sensors or other instrumentation elements with the pulse lighting, the nuclear state.

Model 512 correlation counter is used in photographic stereoscopy cameras for missiles and aircraft as well as in various industrial and military applications.

Mounted on a bracket attached to the optics, the counter displays lighted numbers in front of the shutter which are rotated on the edge of the film frame.

By simultaneously actuating counters mounted on several counter or memory units, the user obtains positive correlation.

A counter number appears on front of which are related to time.

Tron Corp., Encino, Calif.



with a setting accuracy of $\pm 0.2\%$ from greater range of 0 to 90 to 1000, and sensitivity in selection of 3 to 100 g's of acceleration.

Moore Instruments Corp., Long Island City, New York.

Pneumatic System Valve

Valve for pneumatic systems developed for British Comet IV airliner is a low air electro-pneumatic stop valve with a thermal safety limit setting of 490° ambient to plus 380°.

Type Model 32 can be held continuously open during the highest temperatures. Rated at 24 v., it consumes 0.03 amp at 300°. When the solenoid is energized, air from upstream side of the valve passes through pilot valve to actuator of a larger size than the valve proper, thus opening valve. It is claimed to shut off flow of air within 16 ms. The piston rod bleeds part of air before entering the spring chamber, which is vented to atmosphere. Valve is normally used for pressures up to 100 psig. It can be used as blow-off, holding outlet or in own function with a pressure drop of 7 psig at 5 hz/min flow.

Applications include control of compressed-air for air ring, defrost and emergency fuel priming, hot air pneumatic systems, airborne guided weapons and boundary layer control. Hunts Engineering Co., Redditch, Worcestershire, England.



Unit Moves Jetliner Wheels

Mobile power vehicle of self-contained system developed by Comsatellite Div. of General Electric Corp. is pulled beneath wing of Boeing 707 jetliner at start of Seattle, Wash. test flights. Hydraulic gear transmits torque power to 707 wheel tire to propel aircraft on ground. Note pit engine noise suppression of right.

IERC HEAT-DISSIPATING ELECTRON TUBE SHIELDS



**- AND EQUIPMENT "DOWN-TIME" LOSSES
CAUSED BY HEAT, SHOCK AND VIBRATION!**



Investigate this revolutionary measuring, recording, protecting system of IERC heat-dissipating tube shields. The only complete commercially-available line of effective heat-dissipating tube shields for resistors, transistors and electronic components from 100 to 1000 watts, provides a needed line of heat-dissipating tube shields for the larger size power tubes often, for the first time, a practical method to reduce these losses in severe shock and vibration environments?

The most complete electron tube heat-dissipating system available for the market! Practical design and price of \$150.00 per component. Individual test reports will be sent upon request to your company letterhead.

Write or wire for more information or phone 2-4000.
Atlanta 4-4000 U.S.A.



AMERICAN addition to IERC's product line is the IERC HEAT DISSIPATOR for POWER TUBE SISTEKS. Effective reduction of temperatures, extension of life, large or small surfaces play especially for use in enclosed spaces or prime failures. Technical Bulletin PB112 is available with general IERC information and its request.

Heat-dissipating devices are widely used for resistance, stabilizers, cold and power tubes



Pressure Transducer

Coin-type pressure transducer utilizes the uniform strain wave principle for accurate pressure readings in both conventional conditions. Designed for measurements of fluctuating or steady state pressures in liquids or gases, the instruments are used for measurements in hydraulic and pneumatic lines in missiles and aircraft. Water depth determinations, bottom of oil wells, gas and oil pipelines and the monitoring of chemical and petroleum vessels are other applications.

Calibration: PGC-1. Seven pressure transducers are available in ranges of 25, 40, 100, 125, 200, 250 and 300 psi for pressures or temperatures from -100° F. to +100° F. Constructed of Monel housing and stainless steel cable leads can be used in a pressure media until they are equivalent to type 487 standard and

Calton Industries, Inc., 232 Durkin Ave., Methuen, N. J.



Fuel Tank Valve

Fuel valve prevents pressurizing air from being drawn from fuel tank by closing the tank vent before the fuel level falls below the valve seat. Fuel cannot return to tank. Vent air cannot leave the tank and fuel cannot return to the tank. Thus self-venting occurs, reducing valve weight, and valves are used to withstand extreme temperature variations. Valve seats are guaranteed of MIL-E-3272a.

Solen Engineering Dept., Auto Supply Mfg. Co., Corp., Pa.



Snark Lands on Skids With Little Damage



Post landing photographs of Northrop Snark missile show that its unique system of two Y-shaped dual retractable nose landing gear together with skids at wing tips make for a clean plane controlled landing with little damage. This inferior good land control during landing. Two doors open outward and one inward to allow skid to drop freely. Right wing sets out on top deck. Note drag chute at rear. Recovery Skids (below) is lowered for removal or launching storage.





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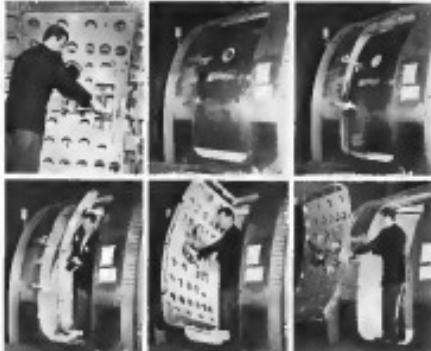
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Boeing 707 Door Tested

Given the pressure with this prototype door which will be installed on Boeing 707 jet transports when they take off, it's a bigger than downsize to open doors so robustly by folding upper and lower rods. Designed by Boeing, door is undergoing tests which include spinning and closing at least 20,000 times.

AEROMARINE, April 21, 1958

Specifications, applications and descriptions of Model 809 total temperature probe, Bulletin 75511, Research Engineering Co., 944 Larchmont Avenue So., Montclair 25, N.J.... Design information for construction drawings of several shapes and sizes, Bulletin AF-5-2, Aeroflex International Technical Inc., Anza, Calif.... Applications and descriptions of Alfrin air controllers, set No. 99, Catalog D, Alfa Corp., 123 East Second St., Minnetonka, Minn. 55375.

Photographs, dimensional drawings, charts and descriptions of Beckman standard electrochemical household pHs, Catalog 575, Elico Corp., division of Beckman Instruments, Inc., New York Branch, Calif.... Applications and test data with discussions of various types insulation may be obtained in technical publications, Bulletin 58, Jones Materials Corp., 481 Franklin Ave., Stamford, Conn.

Description of high performance photographic instrumentation cameras and accessories and serving folder for use with missile target areas, see page folder, Test Corp., 17136 Verdins Blvd., Encino, Calif.... Catalogs of sources and facilities available for the design, development and fabrication of models, mockups and prototypes hardware, P.O.D. Corp., 319 Market St., East Petersburg, Pa. 17522.

Publications Received:

Fire Down and Glory-by Gene Goyette and edited by Mark F. Pacholski, Jr.-Pub. G. P. Putnam's Sons, 210 Madison Ave., New York 16, N. Y. 5575.

A history of American aces from the Lafayette Escadrille through the Korean War. Contains detailed records on USAF fighter pilot kills in both World Wars and the Korean War. Written in the books' "growling" style of World War II Air Force pilots in basic, plain-sounding language, containing many illustrations of either USAF fighter pilots, their planes or their enemies not already in the published record.

Fatigue of Aircraft Structures-Pub. Bureau of Aircraft Strength Division, 1925 Race Street, Philadelphia 3, Pa. \$7.75, 164 pp. (PB 131-675)

Aero-Helicoid. Characteristics of a Cut-Alle of Constant Pre-tension. By N. Heister, H. McCauley and R. Edelstein, Franklin Arsenal, U. S. Army, Geddes, Cornell, July, 1956. \$3.50, 16 pp. (PB 131-297)

Organonics Compounds Part 2-The Organonitrogen Heterocyclics—by H. Rosenberg, C. Tschirhart and M. D. Rausch, Wright Air Development Center, U. S. Air Force, 830, 20 pp. (PB 131-199)

Investigation into the Use of Phenolic Resins as Lubricants Additives Part 2-by G. S. Seeger, O. L. Green and T. A. Horace, Firestone Chemicals and Minerals Co., for Wright Air Development Center, U. S. Air Force, February, 1956. \$3.50, 16 pp. (PB 131-233)

The Determination of Halogen as Gas-line-The Analysis of Organic-Halogen Compounds—by W. D. Garrett and J. A. Krenzien, Naval Research Laboratory, October, 1957. \$3.50, 16 pp. (PB 131-327)

High Speed Thrust Data for Gekkoic Acetate Bolts-by S. E. Els, Rockwell Arsenal, U. S. Army, July, 1956. \$7.50, 25 pp. (PB 131-242)

Tables of Thermodynamic Properties of Gases, Pub. Air in 15,000 K.—by J. Hirschfeld and C. W. Berliner, National Bureau of Standards for Air Research and Development Center, U. S. Air Force, September, 1956. \$1.25, 44 pp. (PB 131-276)

Properties in Fused Titanium-R. A. Wood, D. N. Wilson, H. S. Ogden and R. A. Jolley, Titanium Metallurgical Laboratory, Battelle Memorial Institute, for the Office of Assistant Secretary of Defense for Research and Development, May, 1957. \$1.25, 42 pp. (PB 121-631).

Reports Available:

The following report was sponsored by the Office of Technical Services, United States Department of Commerce, Washington 25, D. C.

High Temperature Protective Coatings for Megaropes—by C. R. Fitzpatrick, J. H. Miller and M. A. Glanz, Metalord Industrial Finishes Co., for Wright Air Development Center, U. S. Air Force, April, 1957. \$1.00, 112 pp. (PB 131-675)

Methods of Accelerated Weather Determination for Plastics Paints—by P. M. Nepon and J. E. Cowling, Naval Radiac Laboratory, November, 1957. \$3.75, 23 pp. (PB 131-180)

Aero-Helicoid. Characteristics of a Cut-Alle of Constant Pre-tension. By N. Heister, H. McCauley and R. Edelstein, Franklin Arsenal, U. S. Army, Geddes, Cornell, July, 1956. \$3.50, 16 pp. (PB 131-297)

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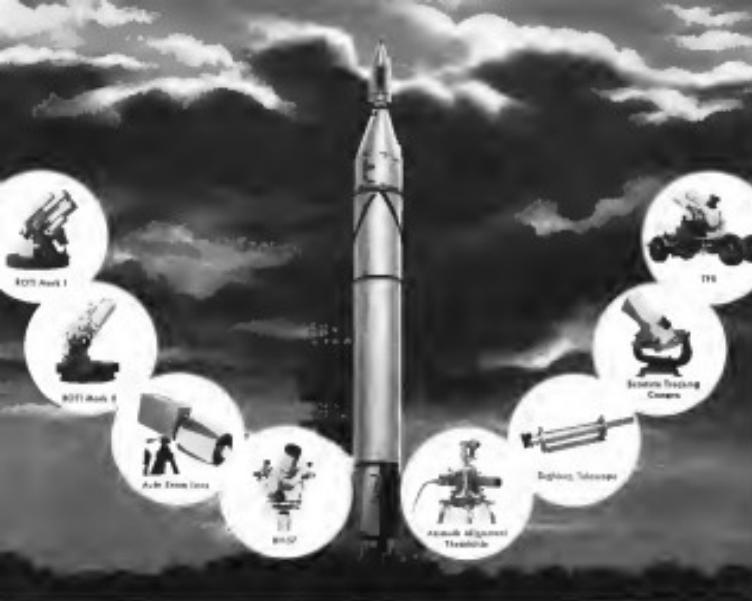


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Engineers want see what is happening to a missile at every stage of its life. And Perkin-Elmer optical range instruments help them see. On the production line where stable platforms are aligned by P-E photometers . . . at the launching site where P-E instruments first key the missile for flight and then record vital data about its performance; or when P-E tracking instruments set objectives which — and leave from — its mechanics journey across the skies.

Data grows more complex as the state-of-the-art advances and the demands upon optical range instrumentation multiply. That's why for a wide variety of aerospace engineering, the number of P-E instruments in the missile program today demonstrates the emphasis that Perkin-Elmer places on this singular talent.

ROTI Mark I—*Ground Optic Range Instrument*—Traces beam from laser to target. Beam splitter divides beam into two paths for simultaneous measurement of range and velocity.

ROTI Mark II—*Single Photometer and Camera*—Velocity, memory, sensitive for beam splitter reflection from point behind clouds or other obstructions.

PPS (Periscope Photographic Recorder)—A camera and monitor system for ROTI, but designed for fast mobility to any site accessible to private road.

PPS Data Recorder—Intermediate recorder developed for evaluation of Ad. 500, monitoring homing runs, etc.

Astrom-Automat Auto-Tracking—Short, intermediate and long range models; short-burst gradient systems of assembly

and at the missile launching site. Optics: Interferometer,康达管, wide field of view and high magnification for quick target acquisition.

SETRAIR Satellite Tracking Control—Employes P-12 atmosphere apertured systems. High light-gathering power, wide field of view and variable moment to photograph VGE satellite.

Auto-Timer 740—Extends sensitivity and range of standard closed circuit TV cameras. Permits remote controlled television observation of weapon, target or homing runs.

An interesting booklet, "Optical Tracking Instruments," describes these and other P-E instruments for the Space Age more fully. Write for it.

ENGINEERING AND OPTICAL DIVISION
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We do no direct sales of OFFICE PADS, nor have specific addressees.



Bell X-15 Hypersonic design study employs wind-tunnel mounted teobooth sensor at reducing complexity of big tail-sitting aircraft.

Hiller Considering Several VTOL Configurations



Research into ducted propeller applications by Hiller Helicopters has centered on coaxial rotors using a rear ducted-propeller heavy helicopter (above) in co-rotating-type (parallel) flight. A static test model which will be terminated this year in order to reduce helicopter complexity, ducted fan consumption. Ducted-propeller vehicle would have 70-80 mph speed, could be capable of supersonic speeds in horizontal flight. Another house-night helicopter (one configuration shown) is one of four designs, others rotatable ducted propellers. Sponsored by Navy, design includes low and off coast landing facilities on two decks. Wind tunnel work is conducted at Navy's David Taylor Model Basin, Washington, D. C.





"Research for Space"

1968's big AVIATION WEEK editorial theme

June 16th marks the publication date of AVIATION WEEK's "Research for Space", 1968's most ambitious and dramatic editorial effort. Major areas of editorial attention: manmade satellites, spacecraft, space probes, Dyna-Soar; avionics; exotic fuels and hypersonic aircraft; are presently engaging the attention of the world and exciting the imagination of scientific and military minds everywhere. The soaring national defense expenditures earmarked for national defense effort in these fields and general aviation procurement, amounting to some \$10 billion scheduled to be expended in the second half of 1968, L. (January-June 1968) spotlight the timeliness of this issue.

Last year's "Research and Development" issue generated over 18,000 top-quality inquiries. This year's "Research for Space" edition will exceed this number substantially. *Reserve your AVIATION WEEK contract copy well ahead.* Positions will be accorded on the basis of the date insertion orders are received. Now is the time to schedule your advertising message in 1968's most dramatic and useful edition, "Research for Space".

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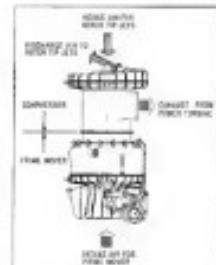
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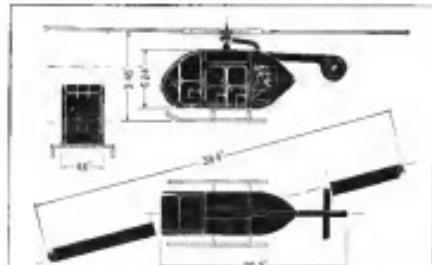
MECHANICAL DIVISION

INVESTMENT CASTINGS AND PRECISION MANUFACTURING

BUSINESS FLYING



Fiat 4900 gas generator powerplant flight for passenger-carrying duty. Fiat 7002 helicopter is rated at 180 hp gas generator for takeoff and 150 hp continuous continuous. An delivery rate is about 7.5 lb/sec. at takeoff and 7.4 during cruise. There view drawing of helicopter right shows revised installation of gas generator, simplicity of landing doors. Streamline fairings have been contoured.



Fiat Selected Cold-Jets for Simplicity

By David A. Anderson

Turin, Italy—Fiat's Fiat 7002 helicopter is expected to fly in about one year, although there is a possibility it may be airborne before the end of 1955.

Rotary wing, head and blades have been completed for the first instance. Components of the cold-jet system have been assembled in full size and are entering the dynamic test stage.

Modifications of the seven-blade aircraft and its control system have been made and are now being used to check out the detailed design which is nearing completion.

Fiat Aviation Division, which earned the charter to get a license from Bell Helicopter Corp. to manufacture its helicopters in Italy, is trying to make up for lost time by designing its own rotary-wing craft aimed at needs of the European operator.

Prospects for production of such aircraft are good in Italy, the company believes, and the seven-blade advanced design should be attractive to operators in Europe.

After the preliminary design, Fiat marketing experts made a survey of the prospects for such a helicopter and found them good. The company's design was accepted by the Italian government and is currently being supported by it to the extent of 10% of the estimated cost. Mated Weapons Development Program accounts for the other 50%, and will a Fiat engineer,

"We are putting up the third 30%, because it's going to cost more than the numbers."

Primary aim of the design team under Ing. Trinchieri were the usual factors of low initial cost and low maintenance cost.

Designers knew this could only be done with a simple design and, after

studying a number of possibilities, settled on the propjet principle with cold jets as being probably the simplest basic drive system.

Since then they have been developing the transmission, evolving it in a basic system and fitting it in component form. Rotary soils, flexible shafts and the blade stage have been devel-

Fiat 7002 Helicopter

PHYSICAL CHARACTERISTICS

	PRODUCTION WEIGHT	3,060
Empty weight, lb.	2,520	
Useful load, lb.	1,760	
Rotor diameter, ft.	16.4	
Blade chord, ft.	3.4	
Flight length, ft.	20.5	
Flight width, ft.	4.6	
Flight height, ft.	6.24	
Overall length to blade disk plane, ft.	9.48	

POWERPLANT CHARACTERISTICS

	MAX. CONTINUOUS GAS GENERATOR	900
Takes off gas horsepower	1,100	530
An delivery to rotor, max. continuous, lb/sec.	7.5	7.5
An delivery to rotor, takeoff, lb/sec.	7.4	
Compressor ratio at max. continuous rating	2.8	
Compressor ratio at idle	1.0	

PERFORMANCE AT NORMAL GROSS WEIGHT

	MAX. SPEED, MAX. ALTITUDE, DAY, MPH	185
Cruise speed, sea level standard day, mph	84	
Service ceiling, forward flight, cont. load, ft.	13,150	
Operational range, mi.	181	

THE ARMY H-23D . . .

HELICOPTER WITH TOMORROW BUILT IN



Design of the iconic H-23 helicopter was largely governed by a doctrine of ruggedness. It has produced a dependable helicopter, with a record of safety unequalled in its class.

Now, in the H-23D, a completely new 1000-hour+ drive system is introduced, seen as a major break-through in lower operating costs. A full-time 250 horsepower is available and, significantly, without "redline" restrictions, working of jeopardized service life. Thus, ruggedness has also afforded growth potential.

In the H-23D, growth potential assumes a new importance. Its existing components are designed to accept even greater power increases for the future's most challenging performance demands. Now, more than ever, the Army H-23 is an investment in tomorrow.



MILLER HELICOPTERS

PALO ALTO, CALIFORNIA

batten or freight spans. Different types of equipment can also be installed for spraying or dusting, for seismic work, for aerial seeding or fertilizing, or an number of the unusual tasks taken on by the aircraft.

The upper boom is attached to the upper portion of the center fuselage section, just above the fuel tanks.

The Disc 4700 gas generator is an adaptation of a light and small carburetor, the 4001, designed for a certain model Lirian.

An engine is from either through the public casting of magnesium-aluminum alloy.

The compressor-turbine section combination is mounted at the lower front portion of the combustion air inlet housing, just below the upper end of a radial housing. The unit is pneumatically balanced to minimize and reduce air flow losses.

Revolving combustion chamber is extremely short. Its design dimensions offer an ease of the engine built of the company's experience in helicopter development. Later engines have combined two chambers that are about hemispherical in shape and of even shorter lengths than those of the 4700.

An ideal time for expansion is read to cool surfaces and reduce weight.

In the H-23D, the power turbine is a single-stage, axial unit with clay steel disc and high nickel-steel blades. The compressor is made of magnesium alloy, its distance from and around the disc is a single magnesium monolithic after casting.

As for the compressor, it turns in front above and fed into the motor through a metal manifold. Exhaust from the disc-turbine casing is collected in an upper manifold and is turned through 90 degrees to discharge to the rear along the helicopter fuselage. Any residual exhaust remaining in the gas will then aspirate thrust for forward motion.

PRIVATE LINES

Their pneumatically operated rotor pitch has been approved for Convair 110, Sikorsky, nose-paced, seat tracks, as approved for Beech Baracuda (A model only), Convair 350, 351 and 540, Lear and Piper Apache.

Automatic Sensing Computer (ASC) electronics provide photo interpretation data, such as minute changes of range data, when adapted to a camera system. This allows one operator to control data up to 100 hours longer than obtainable using human operators using plotting machines. ASC was developed by Photographic Survey Company, Toronto, Canada.

Electronics engineers for preliminary analysis

The Columbus Division of North American has immediate openings for engineers and scientists with a strong background in electronic system development or analysis. Applicants should have the essential objective of working with overall systems in either high Mach aircraft or missiles.

If your engineering background includes data reduction techniques, there are unusual opportunities among various available today and tomorrow. Responsible positions are also open to communications systems, radar systems, systems apertures, and analysis of ECM effects on weapons survival.

You'll assume primary responsibility for the analysis of the entire electronic system in the early configuration state - make decisions that will play an important part in the success of the weapon.

Minimum requirements for these positions are engineering degrees (several preferred) and several years of directly related experience.

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Columbus, Ohio

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THE COLUMBUS DIVISION OF
NORTH AMERICAN AVIATION, INC.

A report to Engineers and Scientists from Lockheed

Missile Systems

...WHERE EXPANDING MISSILE PROGRAMS INSURE MORE

PROMISING CAREERS



**NEW SHOCK TUBE TESTS
MISSILE DESIGNS FOR POLARIS**

This huge new shock tube, performing tests required for the Polaris ballistic missile design data at speeds up to 25,000 feet per second, is now in operation at the Division's research and development laboratories.

The 64-foot long tube is used for advanced design work on the Polaris, a solid propellant weapon to be launched from submarines.

Temperature and pressures generated within the tube will simulate those experienced by a long-range ballistic missile as it passes from sea level to 25,000 feet altitude. This is the first time a test on the shock tube has been made.

Dr. Donald R. Eichinger, Director of Gas Dynamics Dept., with scientists Jerry D. Kennedy and Weyland Marlow III.

**NINE LOCKHEED HUMAN ENGINEERS
HELP MISSILE FLIGHT TESTER**

Polaris and other major Lockheed missile systems like "P" inter theater strategic aircraft of Divisions Missiles programs like Dr. Joseph W. Wheel, left, and Dr. John F. Mongan, right, they are helped by Dr. George A. Aranyan engineer Ray F. Novak, engineer flight control accuracy data on a display oscilloscope.

Studies of man-machine systems like the missile control station enable engineers-psychologists to develop advanced equipment which minimizes the opportunity for human error.

Missile flight control is one of the challenging problems which human engineers—working with other Lockheed scientists and engineers—solve in this area of complex missile systems.

**PAIR OF UNIVACS SOLVE
TOUGH DESIGN PROBLEMS**

Two Univac Superdatus today aid preliminary design work for Lockheed missiles by solving tough flight simulation problems in Divisions Missiles, Project, Palo Alto. The Univacs speed design computation of the most difficult missile characteristics, performing intricate and engrossing calculations and data reduction in much of the most advanced techniques.

Dr. J. P. Stark, left, and E. V. Maldonado review results of a Univac control program before starting problem. The \$10 million computers are part of an installation which is one of the largest and most complex in the West.

**LKS ENGINEERS DEVELOPING
TRANSISTOR FLIGHT CONTROLS**

Transistorized flight control systems for the Polaris ballistic missile program are being tested and developed under the direction of Gene Schles, Flight Controls Department Manager, right. Shown is Schles discussing results of a recent test with design engineer Charles Andle.

Transistorization of flight control systems is receiving top attention from Lockheed Missile Systems engineers and scientists in the interest of saving weight and space—over present flight control systems. This work is being conducted at the Division's Palo Alto and Sunnyvale laboratories.

Lockheed **MISSILE SYSTEMS** A DIVISION OF LOCKHEED AIRCRAFT CORPORATION

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WHO'S WHERE

(Continued from p. 25)

Changes

Robert H. Jewett, assistant general manager-chief engineer, Phillips Service Division, Boeing Airplane Co., Seattle, Wash. **John M. Hensley**, manager, business development department, George Sudzka, regional manager, John D. Alexander, advanced project engineer manager.

H. D. Lowry, director of operations, Middle Division, Learjet Corp., Denver, Colo. **J. F. Lomax**, division manager, and **E. J. McElroy**, special assistant to the general manager.

Robert Shad, chief engineer, Cantron Electric Co., Los Angeles, Calif.

Donald C. Frankel, director of customer support, Chemetco Metal Industries, Inc., Melrose Park, Ill.

Robert J. Holberg, manager RIM-VAC weapon systems, Phillips Aircraft Division, Boeing Airplane Co., Seattle, Wash. **Elmer Wood**, succeeds Mr. Holberg as RIM-VAC project engineer. **E. H. S. Smith**, new project engineer, RIM-VAC weapons.

Herbert H. Houck, Assistant in the Air manager for Development of Working for International Airport Coal Seamlessness Administration. **George R. Morris**, succeeds Mr. Houck as manager of the International Airport. **Mr. Edgar N. Scott**, Planning and Development Officer, **Bartholomew Parsons**, Deputy Director of the Office of Flight Operations and **Wernher von Braun**, Director of the Center of Flight Operations and Aerothermics.

Tom J. Vincze, chief application engineer, electronic systems and equipment, Vertecos division of Northern Genetics Inc., Rochester, Calif. **Alan DeRosa**, a Vertecos technical liaison and research engineer.

Charles Petrus, Jr., general manager Military Products Division, International Business Machines, New York, N.Y.

Albert W. Gough, present manager for international flights, manager of the Latin American Electronic Systems, division of Science Electronics Products Inc., New York, N.Y.

Claud William C. Frazier (32½ yrs.), a vice president, manager Solar Aircraft Co., San Francisco, Calif.

Ward D. Davis, manager, newly created Special Product Sales Division, Douglas Aerospace Inc., El Segundo, Calif. **B. Michael Young**, Jr., succeeds Mr. Davis in the capacity of manager of the division. **John W. Meyer**, sales manager military and commercial engineering services, and **Pat P. Mannion**, sales manager nonmilitary products division.

Donald W. Morris, director of sales services and advertising, Phillips Division, General Cable Co., Cincinnati, Ohio. **W. E. Van Donald L. Quarles**, sales manager, **Lennard D. Williams**, manager-sales office administrator, **William L. Bowles**, Jr., new product manager.

Robert W. O'Neill, U.S. Sales Manager, Lockheed Air Lines.

H. J. B. Winsell, manager of the newly established Test Department, Marconi's

Wraction Turbine Company Ltd., Chelmsford, England. **T. E. Toff** succeeds Mr. Winsell as chief test engineer.

Wilhelm N. Beaudier, project manager, Photo and remote sensing system, **Sylvania Electronic Systems**, division of Sylvania Electric Products Inc., White Plains, N.Y. **M. M. F. J. Anderson**, manager of Sylvania Electronic Systems' newly formed Data Processing Laboratory.

Joseph D. Thompson, manager-aerospace materials engineering, General Electric Research Laboratory, Research Park, New York, N.Y.

I. Weiss, company engineer in James Electronics Instrumentation Co., division of The Keen-Worledge Corp., Decatur, Ga.

F. E. Goss, assistant manager customer



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